versus linear models. Most physical systems exhibit nonlinear controllers.
behavior. Some common examples are saturations, rate limit-
The imm behavior. Some common examples are saturations, rate limit-
ers. hysteresis, and backlash. Predominantly nonlinear be-
picted in Fig. 1. An auxiliary variable, usually called the ers, hysteresis, and backlash. Predominantly nonlinear be- picted in Fig. 1. An auxiliary variable, usually called the
havior may be found in robotic manipulator dynamics, air- scheduling variable is used to undate the gai havior may be found in robotic manipulator dynamics, air-
craft or missile flight dynamics, undersea vehicle dynamics, controller. The scheduling variable should be a good indicacraft or missile flight dynamics, undersea vehicle dynamics, controller. The scheduling variable should be a good indica-
jet engine combustion dynamics, and satellite attitude dy-
tion of the current operating condition o jet engine combustion dynamics, and satellite attitude dy-
namics. Although analysis and control design for nonlinear should be correlated with the plant nonlinearities. The schedsystems remains an active topic of research, analysis of linear uling variable can be a combination of endogenous signals, systems is significantly less complicated, and there is an such as a plant measurements, or exogeno abundance of control design methodologies, ranging from which reflect environmental conditions.

Consider the example is missile autopilot design the example is missile autopilot design

One compromise is to *linearize* the system behavior, that variables are the angle-of-attack and dynamic pressure, both is, approximate the behavior of a nonlinear system near a of which characterize the aerodynamic flight is, approximate the behavior of a nonlinear system near a of which characterize the aerodynamic flight coefficients of particular operating condition by a linear system. This simpli-
the missile The angle of strack is the particular operating condition by a linear system. This simpli-
fication allows one to draw upon analysis and design methods sile hody and velocity vector, and can be considered a state fication allows one to draw upon analysis and design methods sile body and velocity vector and can be considered a state
for linear systems. However, this simplification comes at the variable and hence endogenous to the mi for linear systems. However, this simplification comes at the variable, and hence endogenous to the missile dynamics. The cost of certain limitations:

-
-
-

Despite these limitations, linearization remains a widely used Gain scheduling has seen widespread industrial applicamethod for control system analysis and design.

In many cases, confining a nonlinear system to operating aircraft flight control and missle autopilot design. Other ap-
near a specified operating condition is too restrictive. One ex-
plications include nower systems, pro near a specified operating condition is too restrictive. One ex-
ample is flight control. An aircraft typically experiences sev-
motive control. Despite its widespread usage traditional gain ample is flight control. An aircraft typically experiences sev-
eral flight conditions, including take-off, cruising at various
scheduling has been an ad boc design approach accompanied eral flight conditions, including take-off, cruising at various
altitudes, specialized maneuvers, and landing. No single by heuristic guidelines.
linearization can adequately describe the aircraft dynamics at all of these conditions. Another example is boiler-turbine control in power generation. Typical operating conditions include power ramp up, steady power delivery at various levels, and power ramp down. Again, no linearization can adequately describe the dynamics at all operating conditions.

WHAT IS GAIN SCHEDULING?

Gain scheduling is an approach to overcome the local limitations associated with linearizations. The idea is simple and intuitively appealing. Given a nonlinear plant with a wide range of operating conditions, one can select *several* representative operating conditions within the operating regime, perform several linearizations of the nonlinear dynamics, and de- **Figure 1.** Gain scheduled control implementation.

GAIN SCHEDULING Sign several linear controllers, one for each operating condition.

MOTIVATION Each individual controller is expected to achieve good performance whenever the nonlinear plant is near the control-A classical tradeoff in control design is model accuracy versus ler's associated operating condition. As the plant varies from model simplicity. While sophisticated models better represent one operating condition to another, the gains of the individual a physical system's behavior, the resulting analysis and con- controllers are interpolated, or scheduled, to match the trol design are more involved. The final result is a nonlin-One manifestation of this tradeoff is the use of nonlinear ear controller which is constructed out of several local linear

> should be correlated with the plant nonlinearities. The schedsuch as a plant measurements, or exogenous parameters

classical control to multivariable robust control. One example is missile autopilot design. Useful scheduling
One compromise is to *linearize* the system behavior, that variables are the angle-of-attack and dynamic pressur dynamic pressure, which is a function of missle velocity and atmospheric pressure, is indicative of the environmental con- • The nonlinear system must be confined to operating near ditions. Atmospheric pressure is clearly an exogenous signal. the specified operating condition of the linearization. Since the dynamic pressure is also a function of missile veloc-• The linearization analysis may give misleading or incon-
the velocity variations in a simplified model are decoupled
from the attitude dynamics, and hence, dynamic pressure • The linearization may ignore important nonlinear phe-
nomena which dominate the system behavior.
ity, namely that an "exogenous" signal is really an endoge-
ity, namely that an "exogenous" signal is really an endogenous signal in a more sophisticated model, is common.

ethod for control system analysis and design.
In many cases, confining a nonlinear system to operating aircraft flight control and missle autopilot design. Other an-

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lowing:

- The design of several controllers at several linearization points can be a tedious task. Then,
• Gain scheduled designs typically assume a fixed op-
- erating condition, even though the operating condition is varying. lim
x→*x_o*
- Although the resulting controller is nonlinear, it is still based on linearizations of plant dynamics, and hence where $|v|$ denotes the Euclidean norm of $v \in \mathbb{R}^n$, may neglect important nonlinear phenomena.

Two heuristic rules-of-thumb which guide successful gain $|v|$ scheduled designs are the following:

- The scheduling variable should vary slowly.
- The scheduling variable should capture plant nonlinear-
ities. $f(x) =$

It is easy to interpret these guidelines in terms of the Then aforementioned drawbacks. Since gain scheduled designs assume a constant operating condition, slow variations among operating conditions should be tolerable. Similarly, since gain scheduling relies on a family of linearizations, the changes within this family should be indicative of plant nonlinearities.

This article provides an overview of the gain scheduling Approximating $f(x)$ near design procedure, discusses the theoretical foundations behind gain scheduling, as well as limitations of traditional gain scheduling, and presents emerging techniques for gain scheduling which address these limitations.

LINEARIZATION leads to

Linearization of Functions

We begin by recalling some concepts from multivariable calculus. Let $f: \mathbb{R}^n \to \mathbb{R}^p$ denote a multivariable function which maps vectors in \mathbb{R}^n to vectors in \mathbb{R}^p . In terms of the individual components,

$$
f(x) = \begin{bmatrix} f_1(x_1, \dots, x_n) \\ \vdots \\ f_p(x_1, \dots, x_n) \end{bmatrix}
$$
 (1)

In case *f* is differentiable, $Df(x)$ denotes the $p \times n$ Jacobian matrix of partial derivatives; i.e.,

$$
Df(x) = \begin{bmatrix} \frac{\partial f_1}{\partial x_1}(x) & \cdots & \frac{\partial f_1}{\partial x_n}(x) \\ \vdots & \ddots & \vdots \\ \frac{\partial f_p}{\partial x_1}(x) & \cdots & \frac{\partial f_p}{\partial x_n}(x) \end{bmatrix}
$$
(2)

For f continuously differentiable, i.e., if $Df(x)$ has continuous elements, we may approximate $f(x)$ by the truncated Taylor's series

$$
f(x) \approx f(x_0) + Df(x_0)(x - x_0)
$$
\n(3)

Some immediate drawbacks of gain scheduling are the fol- Let $r(x)$ denote the residual error of approximation; that is,

$$
r(x) = f(x) - (f(x_0) + Df(x_0)(x - x_0))
$$
 (4)

$$
\lim_{x \to x_0} \frac{r(x)}{|(x - x_0)|} = 0
$$
\n(5)

$$
v = (v^T v)^{1/2} \tag{6}
$$

Example 1. Let $f: \mathbb{R}^2 \to \mathbb{R}^2$ be defined as

$$
f(x) = \begin{pmatrix} x_1 | x_1 | + x_2 \\ x_1^2 x_2 \end{pmatrix}
$$
 (7)

$$
Df(x) = \begin{pmatrix} 2|x_1| & 1\\ 2x_1x_2 & x_1^2 \end{pmatrix}
$$
 (8)

$$
x_0 = \begin{pmatrix} 1 \\ 2 \end{pmatrix}
$$

$$
f(x) \approx f(x_0) + Df(x_0)(x - x_0)
$$

= $\binom{3}{2} + \binom{2}{4} + \binom{x_1 - 1}{x_2 - 2}$ (9)

Now, let $f: \mathbb{R}^n \times \mathbb{R}^m \rightarrow \mathbb{R}^p$ denote a multivariable function which maps vectors in \mathcal{R}^n and \mathcal{R}^m together to vectors in \mathcal{R}^p . In terms of the individual components,

$$
f(x, u) = \begin{bmatrix} f_1(x_1, ..., x_n, u_1, ..., u_m) \\ \vdots \\ f_p(x_1, ..., x_n, u_1, ..., u_m) \end{bmatrix}
$$
 (10)

In case *f* is differentiable, $D_1 f(x, u)$ denotes the $p \times n$ Jacobian matrix of partial derivatives with respect to the first variable,

$$
D_1 f(x, u) = \begin{bmatrix} \frac{\partial f_1}{\partial x_1}(x, u) & \cdots & \frac{\partial f_1}{\partial x_n}(x, u) \\ \vdots & \ddots & \vdots \\ \frac{\partial f_p}{\partial x_1}(x, u) & \cdots & \frac{\partial f_p}{\partial x_n}(x, u) \end{bmatrix}
$$
(11)

and $D_2 f(x, u)$ denotes the $p \times m$ Jacobian matrix of partial and derivatives with respect to the second variable,

$$
D_2 f(x, u) = \begin{bmatrix} \frac{\partial f_1}{\partial u_1}(x, u) & \cdots & \frac{\partial f_1}{\partial u_m}(x, u) \\ \vdots & \ddots & \vdots \\ \frac{\partial f_p}{\partial u_1}(x, u) & \cdots & \frac{\partial f_p}{\partial u_m}(x, u) \end{bmatrix}
$$
(12)

As before, if f is continuously differentiable, we may approximate $f(x, u)$ by Equation (22) is called the *linearization* of Eq. (16) about

$$
f(x, u) \approx f(x_0, u_0) + D_1 f(x_0, u_0)(x - x_0) + D_2(x_0, u_0)(u - u_0)
$$
\n(13)

$$
r(x, u) = f(x, u) - (f(x_0, u_0) + D_1 f(x_0, u_0)(x - x_0)
$$

+ $D_2(x_0, u_0)(u - u_0)$) (14)

denote the approximation residual. Then as before, **Definition 3.** Let x_0 be an equilibrium of Eq. (16).

$$
\lim_{\substack{x \to x_0 \\ u \to u_0}} \frac{r(x, u)}{\sqrt{|x - x_0|^2 + |u - u_0|^2}} = 0
$$
\n(15)

Equations (5) and (15) indicate that the approximations Otherwise, x_0 is unstable.
are accurate up to the first order.

Linearization of Autonomous Systems → α → β = α

In the last section, we saw how to approximate the static behavior of a nonlinear function by using a truncated Taylor's In words, stability implies that the solution to Eq. (16) series. We now show how similar tools can be used to approxi- stays near x_0 whenever it starts sufficiently close to x_0 , mate the dynamic behavior of a nonlinear system. whereas asymptotic stability implies that the solution also as-

Consider an autonomous nonlinear system ymptotically approaches x_0 .

$$
\dot{x} = f(x) \tag{16}
$$

$$
f(x_0) = 0 \tag{17}
$$

tion $x(0) = x_0$ leads to the solution $x(t) = x_0$ for all time. So if
the solution starts at x_0 , it remains at x_0 , hence the term equi-
librium.

In case *f* is continuously differentiable, we may rewrite $\frac{d^2\theta}{dt^2}$

$$
\dot{x} = f(x_0) + Df(x_0)(x - x_0) + r(x) \tag{18}
$$

$$
f(x) \approx f(x_0) + Df(x_0)(x - x_0)
$$
 (19)

Since x_0 is both fixed and an equilibrium,

$$
\frac{d}{dt}(x - x_0) = \frac{dx}{dt}
$$
\n(20)

$$
f(x_0) = 0\tag{21}
$$

Substituting these formulas into (18) and neglecting the residual term, $r(x)$, leads to the approximate dynamics

$$
\dot{\tilde{x}} = A\tilde{x} \tag{22}
$$

where

$$
A = Df(x_0) \tag{23}
$$

the equilibrium x_0 . Intuitively, whenever $x - x_0$ is small, then $f(x, u) \approx f(x_0, u_0) + D_1 f(x_0, u_0)(x - x_0) + D_2(x_0, u_0)(u - u_0)$ the equilibrium x_0 . Intuitively, whenever $x - x_0$ is small, then
(13) \tilde{x} in the linearization should be a good approximation of x x_0 , and hence the linear dynamics of Eq. (22) should be a good Let approximation of the nonlinear dynamics of Eq. (16).

It is often possible to make more definite statements about nonlinear dynamics based on an analysis of the linearization, in particular regarding the stability of the nonlinear system. First, recall the following stability definitions.

• x_0 is stable if for any $\epsilon > 0$, there exists a $\delta > 0$ such that

$$
|x(0) - x_0| < \delta \implies |x(t) - x_0| \le \epsilon \tag{24}
$$

• x_0 is asymptotically stable if in addition to being stable,

$$
x(0) - x_0 < \delta \implies \lim_{t \to \infty} x(t) = x_0 \tag{25}
$$

Theorem 4. Let f in Eq. (16) be continuously differentiable. The equilibrium x_0 is asymptotically stable if all of the eigen-*Definition 2.* The vector x_0 is an equilibrium of (16) if values of $Df(x_0)$ have negative real parts. It is unstable if $Df(x_0)$ has at least one eigenvalue with a positive real part.

Since the eigenvalues of $Df(x_0)$ determine the stability of the linearization (22), Theorem 4 states that one can assess The reasoning for this terminology is that the initial condi-
the stability of a nonlinear system based on its linearization.

$$
\frac{d^2\theta}{dt^2} + c\frac{d\theta}{dt} - \frac{g}{\ell}\sin(\theta) = 0
$$
 (26)

x where θ is the pendulum angle measure positive clockwise with $\theta = 0$ being the upright position, *c* is a friction coeffiwhere $r(x)$ denotes the residual error in the approximation cient, and g is gravitational acceleration. In state space form, ${\bf these \; equations \; becomes}$

$$
\frac{d}{dt}\begin{pmatrix} x_1 \\ x_2 \end{pmatrix} = \begin{pmatrix} x_2 \\ \frac{g}{\ell} \sin(x_1) - cx_2 \end{pmatrix}
$$
 (27)

where $x_1 = \theta$, and $x_2 = \dot{\theta}$.

$$
\dot{\tilde{x}} = \begin{pmatrix} 0 & 1 \\ g/\ell & -c \end{pmatrix} \tilde{x}
$$
 (28)

The linearization has an eigenvalue with positive real part. Hence, the upright equilibrium of the pendulum is unstable, as expected.
Linearizing about the hanging equilibrium where $r(x, u)$ denotes the residual error in the approximation

$$
x_0 = \begin{pmatrix} \pi \\ 0 \end{pmatrix}
$$

$$
\dot{\tilde{x}} = \begin{pmatrix} 0 & 1 \\ -g/\ell & -c \end{pmatrix} \tilde{x}
$$
 (29)
$$
\dot{\tilde{x}} = A\tilde{x} + B\tilde{u}
$$
 (37)

which is asymptotically stable. Therefore, the hanging equilibrium of the nonlinear pendulum is asymptotically stable, as expected. As before, Eq. (37) is called the *linearization* of the nonlinear

The above example demonstrates that different equilibrium points of the same nonlinear system can have different

m some cases, asing the inconclusive.
by the state feedback $u = G(x)$ if $y(x) = G(x)$

Example 6. Consider the scalar nonlinear systems

$$
\dot{x} = -x^3 \tag{30}
$$

and

$$
\dot{x} = x^3 \tag{31}
$$

It is easy to see that the equilibrium $x_0 = 0$ is asymptotically
stable for the former system, while the same equilibrium is
unstable for the latter system.
unstable for the latter system.

Both systems have the same linearization at the equilib- *rium* $x_0 = 0$,

$$
\dot{\tilde{x}} = 0 \tag{32}
$$

(note that \tilde{x} represents different quantities in the two linearizations). In this case, stability analysis of linearization is inconclusive; it does not indicate either the stability or instabil-
ity of the equilibrium $x_0 = 0$.
back for a nonlinear system by designing stabilizing feedback

for its linearization. **Linearization of Systems with Controls**

controllers for a nonlinear system. with a control torque input,

Consider the controlled system

$$
\dot{x} = f(x, u) \qquad (33)
$$

Definition 7. The pair (x_0, u_0) is an equilibrium of Eq. (33) if

$$
f(x_0, u_0) = 0 \tag{34}
$$

The reasoning behind the term equilibrium is similar to the autonomous case. The initial condition $x(0) = x_0$ along

Linearizing about the upright equilibrium $x_0 = 0$ leads to with the *constant* input $u(t) = u_0$ leads to the constant solution $x(t) = x_0$.

> Proceeding as in the autonomous case, whenever *f* is continuously differentiable, we may rewrite (33) as

$$
\begin{aligned} \dot{x} &= f(x_0, u_0) + D_1 f(x_0, u_0)(x - x_0) \\ &+ D_2 f(x_0, u_0)(u - u_0) + r(x, u) \end{aligned} \tag{35}
$$

$$
f(x, u) \approx f(x_0, u_0) + D_1 f(x_0, u_0)(x - x_0)
$$

+
$$
D_2 f(x_0, u_0)(u - u_0)
$$
 (36)

Dropping this residual term and using that $f(x_0, u_0) = 0$ leads to the approximate linear dynamics

$$
\dot{\tilde{x}} = A\tilde{x} + B\tilde{u} \tag{37}
$$

where

$$
A = D_1 f(x_0, u_0), \quad B = D_2 f(x_0, u_0) \tag{38}
$$

The above example demonstrates that different equilib-
m points of the same nonlinear system can have different mates $x - x_0$, whereas the quantity \tilde{u} equals $u - u_0$, exactly.

stability conditions.
In some cases, using the linearization to assess stability **Definition 8.** The equilibrium (x_0, u_0) of Eq. (33) is stabilized

$$
u_0 = G(x_0)
$$

• x_0 is a stable equilibrium of the closed loop dynamics

$$
\dot{x} = f(x, G(x))\tag{39}
$$

The following is a direct consequence of Theorem 4.

$$
\tilde{u} = -K\tilde{x} \tag{40}
$$

stabilizes the linearization of Eq. (37). Then the equilibrium (x_0, u_0) of Eq. (33) is stabilized by the feedback

$$
u = u_0 - K(x - x_0) \tag{41}
$$

It is also possible to use linearization methods to *synthesize Example 10.* Recall the simple pendulum example, but now

$$
\frac{d}{dt}\begin{pmatrix} x_1 \\ x_2 \end{pmatrix} = \begin{pmatrix} x_2 \\ \frac{g}{\ell} \sin(x_1) - cx_2 + u \end{pmatrix}
$$
(42)

Linearizing about the upright equilibrium leads to *f*(\hat{A}) \hat{B}

$$
\dot{\tilde{x}} = \begin{pmatrix} 0 & 1 \\ g/\ell & -c \end{pmatrix} \tilde{x} + \begin{pmatrix} 0 \\ 1 \end{pmatrix} \tilde{u}
$$
 (43)

$$
\tilde{u} = -(k_1 \quad k_2)\tilde{x} \qquad (44) \qquad \tilde{x} = A\tilde{x} + B\tilde{x}
$$

stabilizes the linearization for any $k_1 > g/\ell$ and $k_2 > -c$. From Theorem 9, the feedback where where where \sim

$$
u = u_0 - (k_1 \ k_2)(x - x_0)
$$

= -(k₁ k₂)x (45)

stabilizes the upright equilibrium, where we used that $x_0 = 0$ and $u_0 = 0$. and $u_0 = 0$. mates $y - y_0$, and \tilde{u} exactly represents $u - u_0$.

In some cases, analysis of the linearization does not aid in **Theorem 13.** Let *f* in Eq. (33) and *g* in Eq. (49) be continu-
the construction of stabilizing feedback.

Example 11. Consider the scalar nonlinear system feedback,

$$
\dot{x} = x + xu \qquad (46) \qquad \dot{z} = \overline{A}z + \overline{B}\tilde{y}
$$

Linearizing about the equilibrium $(x_0, u_0) = (0, 0)$ leads to

$$
\dot{\tilde{x}} = \tilde{x} \tag{47}
$$

which is not stabilizable. However, the constant ''feedback'' *u* = -2 leads to the closed loop equations $\dot{z} = Az + B(y - g(x_0))$

$$
\dot{x} = -x \tag{48}
$$

Now suppose that the state is not available for feedback. Rather, the control is restricted to measurements

$$
y = g(x) \tag{49}
$$

By using a similar analysis, we can construct stabilizing output feedback for the nonlinear system based on stabilizing output feedback for the linearization.

Definition 12. The equilibrium (x_0, u_0) of Eq. (33) is stabi-
lized by the dynamic output feedback

$$
\begin{aligned}\n\dot{z} &= F(z, y) \\
u &= G(z)\n\end{aligned} \tag{50}
$$

• $(z_0, g(x_0))$ is an equilibrium of Eq. (50)

• $u_0 = G(z_0)$

$$
\begin{aligned} \n\dot{x} &= f(x, G(z)) \\ \n\dot{z} &= F(z, g(x)) \n\end{aligned} \tag{51}
$$

Let (x_0, u_0) be an equilibrium of Eq. (33), and define $y_0 =$ **Gain Scheduled Command Following** $g(x_0)$. In case *g* is continuously differentiable, we can approxi-
mate Eq. (49) as
context of command following. The nonlinear plant of interest

$$
y \approx y_0 + Dg(x_0)(x - x_0) \tag{52}
$$

The feedback Then, the linearization about the equilibrium (x_0, u_0) ,

$$
\begin{aligned}\n\dot{\tilde{x}} &= A\tilde{x} + B\tilde{x} \\
\tilde{y} &= C\tilde{x}\n\end{aligned} \tag{53}
$$

 $A = D_1 f(x_0, u_0), \quad B = D_2 f(x_0, u_0), \quad C = Dg(x_0)$ (54)

approximates the input-output behavior of Eq. (33) with measurement of Eq. (49). Here, \tilde{x} approximates $x - x_0$, \tilde{y} approxi-

ously differentiable, and let Eq. (53) be the linearization of Eq. (33) about the equilibrium (x_0, u_0) . Suppose the linear

$$
\begin{aligned}\n\dot{z} &= \overline{A}z + \overline{B}\tilde{y} \\
\tilde{u} &= \overline{C}z\n\end{aligned} \tag{55}
$$

stabilizes the linearization of Eq. (53). Then the equilibrium (x_0, u_0) of Eq. (33) is stabilized by the output feedback

$$
\begin{aligned} \dot{z} &= \overline{A}z + \overline{B}(y - g(x_0)) \\ u &= u_0 + \overline{C}z \end{aligned} \tag{56}
$$

Example 14. Suppose we wish to control the simple pendu-
lum under the output feedback

$$
y = (1 \quad 0)x \tag{57}
$$

Linearizing about the upright equilibrium leads to

$$
\dot{\tilde{x}} = \begin{pmatrix} 0 & 1 \\ g/\ell & -c \end{pmatrix} \tilde{x} + \begin{pmatrix} 0 \\ 1 \end{pmatrix} \tilde{u} \stackrel{\text{def}}{=} A\tilde{x} + B\tilde{u}
$$
\n
$$
\tilde{y} = (1 \ 0)\tilde{x} \stackrel{\text{def}}{=} C\tilde{x}
$$
\n(58)

$$
\dot{z} = F(z, y) \qquad \dot{z} = (A - BK - HC)z + H\tilde{y} \tag{59}
$$

 α if for some z_0 , α is tabilizes the linearization for appropriate gain matrices

$$
K = (k_1 \quad k_2), \quad H = \begin{pmatrix} h_1 \\ h_2 \end{pmatrix} \tag{60}
$$

• (x_0, z_0) is an asymptotically stable equilibrium of the Since x_0 , u_0 , $y_0 = 0$, the same controller (with input y and closed loop dynamics doesn't all stabilizes the nonlinear pendulum.

BASIC GAIN SCHEDULING

context of command following. The nonlinear plant of interest is Eq. (33). The objective is to make the measured output Eq. *(49)* approximately follow reference commands, *r*.

The primary motivation of gain scheduling is to address *Example 17.* Recall the controlled pendulum local limitations associated with a control design based on a single linearization. The main problem is that the performance and even stability of the closed loop system can deteriorate significantly when the system is not operating in the vicinity of the equilibrium.

Example 15. Consider the scalar system

$$
\begin{aligned}\n\dot{x} &= x|x| + u\\
y &= x\n\end{aligned} \tag{61}
$$

By linearizing about the equilibrium $(x_0, u_0) = (0, 0)$, we obtain the linear control $u = -x + r$. For $r = 0$, this control law stabilizes the equilibrium $(0, 0)$. The resulting closed loop system is

$$
\dot{x} = x|x| - x \tag{62}
$$

For $|x(0)| < 1$, the solution asymptotically approaches 0. However, for $|x(0)| > 1$, the solution diverges to infinity.

Step 1: Construction of Linearization Family. Gain scheduling velope. attempts to overcome local limitations by considering a *family*

rium family for the nonlinear system Eq. (33) over the set *S* if

$$
f(x_{\text{eq}}(s), u_{\text{eq}}(s)) = 0 \tag{63}
$$

Associated with an equilibrium family are the output equilibrium values This step constitutes the core of gain scheduling, and ac-

$$
y_{\text{eq}}(s) \stackrel{\text{def}}{=} g(x_{\text{eq}}(s)) \tag{64}
$$

The equilibrium family induces the following linearization operating conditions. family for Eq. (33) with measurement Eq. (49),

$$
\dot{\tilde{x}} = A(s)\tilde{x} + B(s)\tilde{u}
$$

\n
$$
\tilde{y} = C(s)\tilde{x}
$$
\n(65)

$$
A(s) = D_1 f(x_{eq}(s), u_{eq}(s)),
$$

\n
$$
B(s) = D_2 f(x_{eq}(s), u_{eq}(s)),
$$

\n
$$
C(s) = Dg(x_{eq}(s))
$$
\n(66)

parameterizes a family of equilibrium points and plant linear- held constant during the design phase, is now time varying. izations. Typically, *s* can be a combination of both endogenous These matrix and vector functions are used to update the conand exogenous signals (recall discussion of missile autopilot trol parameters according to the variations in the scheduling earlier). Any fixed *s* will be called an *operating condition,* and variable. the set *S* defines the *operation envelope,* or range of op- There are different options in how to schedule the controlerating conditions. ler parameters in Eq. (73).

$$
\frac{d}{dt}\begin{pmatrix} x_1 \\ x_2 \end{pmatrix} = \begin{pmatrix} x_2 \\ \frac{g}{\ell} \sin(x_1) - cx_2 + u \end{pmatrix}
$$
(67)

An equilibrium family over the set $S = [-\pi, \pi]$ is

$$
\dot{x} = x|x| + u
$$
\n(68)\n
$$
\dot{x} = x|x| + u
$$
\n(69)

The associated linearization family is

$$
\tilde{x} = \begin{pmatrix} 0 & 1 \\ \frac{g}{\ell} \cos(s) & -c \end{pmatrix} \tilde{x} + \begin{pmatrix} 0 \\ 1 \end{pmatrix} \tilde{u}
$$
(69)

Step 2: Fixed Operating Condition Designs. Let us select several operating conditions,

$$
\{s_1, s_2, \dots, s_N\} \subset S \tag{70}
$$

which characterize the variations within the operating en-

At the i^{th} equilibrium, s_i , we can linearize the plant dynamof linearizations, rather than a single linearization. ics about the equilibrium $(x_{eq}(s_i), u_{eq}(s_i))$ and design a stabilizing linear controller to achieve approximate command follow-*Definition 16.* The functions $(x_{eq}(\cdot), u_{eq}(\cdot))$ define an equilib- ing using any suitable linear design methodology. The result rium family for the nonlinear system Eq. (33) over the set is an indexed collection of cont

$$
\begin{aligned}\n\dot{z} &= \overline{A}_i z + \overline{B}_i \tilde{y} + \overline{L}_i \tilde{r} \\
\tilde{u} &= \overline{C}_i z\n\end{aligned} \tag{71}
$$

for all $s \in S$.

$$
\tilde{r} = r - y_{\text{eq}}(s_i) \tag{72}
$$

cordingly, accounts for the bulk of the effort in a gain sched $y_{eq}(s) = g(x_{eq}(s))$ (64) uled control design. Designing fixed operating point controllers is especially tedious in the case of several design

Step 3: Scheduling. The remaining step is to piece together a global controller from the individual local controllers. As the scheduling variable varies in time, the control gains are updated to reflect the current operating condition of the plant. The resulting overall controller is

$$
\begin{aligned} \dot{z} &= \overline{A}(s)z + \overline{B}(s)(y - \overline{y}_{\text{eq}}(s)) + \overline{L}(s)(r - \overline{y}_{\text{eq}}(s)), \\ u &= \overline{u}_{\text{eq}}(s) + \overline{C}(s)z \end{aligned} \tag{73}
$$

The matrices $\overline{A}(s)$, $\overline{B}(s)$, $\overline{C}(s)$, and $\overline{L}(s)$ are functions of the scheduling variable, as are the vectors $\bar{y}_{eq}(s)$ and $\bar{u}_{eq}(s)$. It is The variable, *s*, which we will call the *scheduling variable,* important to note that the scheduling variable, *s*, which was

into disjoint regions, R_i , so that **Example 15.**

$$
S \subset R_1 \cup \ldots \cup R_N \tag{74}
$$

$$
\overline{A}(s) = \begin{cases}\n\overline{A}_1, & \text{if } s \in R_1; \\
\overline{A}_N, & \text{if } s \in R_N.\n\end{cases}
$$
\n
$$
= \begin{cases}\n u_{\text{eq}}(s_1), & \text{if } s \in R_1; \\
 u_{\text{eq}}(s_N), & \text{if } s \in R_N.\n\end{cases}
$$
\n(75)

the design conditions so that system.

$$
\overline{A}(s_i) = \overline{A}_i, \dots, \overline{u}_{\text{eq}}(s_i) = u_{\text{eq}}(s_i)
$$
 (76) tem are

Some important points to consider are the following:

- Gain scheduling is still based on linearizations, and hence can ignore important nonlinear phenomena.
- The fixed operating point designs assume a constant scheduling variable which is actually time varying.
- Implementing the gain scheduled controller introduces feedback loops which are not present in the fixed op- In general, the scheduling variable, *s*, can be written as erating point designs.

Example 18. Recall the system of Example 15. The equilib-

$$
x_{\rm eq}(s) = s, \quad u_{\rm eq}(s) = -s|s| \tag{77}
$$

$$
\tilde{x} = 2|s|\tilde{x} + \tilde{u} \tag{78}
$$

Because of the simplicity of this system, we are able to design predominantly nonlinear in the scheduling variable.

controls for all s, rather than selected s. A suitable linear de-

strictions. sign for command following is

$$
\tilde{u} = -3|s|\tilde{x} + (\tilde{r} - \tilde{x})\tag{79}
$$

$$
u = u_{eq}(s) + \tilde{u}
$$

= $u_{eq}(s) - 3|s|(x - x_{eq}(s)) + ((r - x_{eq}(s)) - (x - x_{eq}s))$ (80)

For the scheduling variable $s = x$, the control becomes

$$
u = -x|x| + (r - x)
$$
 (81)

This feedback leads to the closed loop dynamics θ are magnitude bounds; for example,

$$
\dot{\mathbf{r}} = -\mathbf{x} + \mathbf{r} \tag{8}
$$

• *Switched Scheduling.* The operating envelope is divided which are globally stable, as opposed to the local stability of

 $S \subset R_1 \cup ... \cup R_N$ (74) Notice in this example that the linear control term (*x* and the controller matrices in (73) are scheduled ac-
cording of the smooth scheduling implementation with $s = x$ as
cording to
the smooth scheduling variable. In this case, a desirable feedback loop was eliminated in the scheduling implementation. It is also possible to *introduce* undesirable feedback during implementation.

Theoretical Foundations

The gain scheduled controller is designed so that stability and performance are achieved whenever the plant is in the vicinity one of the design operating conditions. Since the plant actually varies throughout the entire operating regime, an im- • *Continuous Scheduling.* Any interpolation algorithm is portant question is to what degree the *local* properties of the used to construct continuous matrices which interpolate individual operating point designs carry over to the *global*

The overall closed loop equations for a gain scheduled sys-

$$
\begin{aligned}\n\dot{x} &= f(x, u) \\
\dot{z} &= \overline{A}(s)z + \overline{B}(s)\tilde{y} + \overline{L}(s)\tilde{r} \\
u &= \overline{C}(s)z + \overline{u}_{eq}(s) \\
\tilde{y} &= g(x) - \overline{y}_{eq}(s) \\
\tilde{r} &= r - \overline{y}_{eq}(s)\n\end{aligned} \tag{83}
$$

$$
s = \gamma(x, r) \tag{84}
$$

rium family for an appropriate function, γ . Clearly, the overall system is

nonlinear and hence, requires nonlinear methods for analysis. An analysis of these equations (see Bibliography for leads to the linearization family sources) leads to the conclusion that the overall gain sched-
leads to the linearization family stability and performance as the local designs whenever (1) the scheduling variable, *s*, changes "sufficiently slowly," and (2) the plant dynamics are

LPV Systems. It is convenient to consider slow variation restriction in the context of linear parameter varying (LPV) sys-Implementing this design using smooth scheduling leads to
the gain scheduled control the gain scheduled control
which are unknown a priori, but can be measured upon opera-
example 2.1 and 2.1 and 2.1 and 2.1 and 2.1 and 2. tion of the control system.

An LPV system can be represented in state space form as

$$
\begin{aligned} \dot{x} &= A(\theta)x + B(\theta)u \\ y &= C(\theta)x \end{aligned} \tag{85}
$$

where θ is a time varying parameter. Typical assumptions on

$$
\dot{x} = -x + r \tag{82}
$$

$$
|\dot{\theta}| \le \dot{\theta}_{\text{max}} \tag{87}
$$

LPV systems form the underlying basis of gain scheduling. It is convenient to associate the "parameter" with the scheduling variable and the LPV structure with the linearization If $v_{eq}(s)$ is differentiable, family, although this is not always the case as will be seen.

The following is a classical result from differential equations stated in an LPV context.

Theorem 2. If the equilibrium $x_0 = 0$ of Eq. (85) is asymptotically stable for all constant θ , then it is asymptotically stable for all time varying θ provided that $\dot{\theta}_{\text{max}}$ is sufficiently small. where the matrices $M_{12}(s)$ and B_1 are appropriate sub-matri-

The relevance of Theorem 19 to gain scheduling is as fol-
Combining these equations leads to the alternate form of lows. A closed loop LPV system is such that good stability and E_q . (89), performance is expected for fixed values of the parameter/ scheduling variable. However, performance and even stability can deteriorate in the presence of parameter time variations. Theorem 19 provides a sufficient condition for the fixed parameter properties to carry over to the varying parameter setting.

Example 20. A classical example of instability from fixed parameter stability is the time-varying oscillator,

$$
\dot{x}(t) = \begin{pmatrix} 0 & 1 \\ -(1 + \theta(t)/2) & -0.2 \end{pmatrix} x(t) \tag{88}
$$

These equations can be viewed as a mass-spring-damper system with time-varying spring stiffness. For fixed parameter values, $\theta(t) = \theta_o$, the equilibrium $x_o = 0$ is asymptotically stable. However, for the parameter trajectory $\theta(t) = \cos(2t)$, it where becomes unstable. An intuitive explanation is that the stiff v are timed to pump energy into the oscillations.

$$
\frac{d}{dt}\begin{pmatrix} s \\ v \end{pmatrix} = \phi(s) + M(s)\begin{pmatrix} s \\ v \end{pmatrix} + Bu \tag{89}
$$

These equations represent the extreme case where the nonlin-
earities are *entirely* captured in the scheduling variable, *s*. *Example 21*. Consider the nonlinear system Let $(x_{eq}(s), u_{eq}(s))$ be an equilibrium family, with

$$
x_{\rm eq}(s) = \begin{pmatrix} s \\ v_{\rm eq}(s) \end{pmatrix} \tag{90}
$$

$$
0 = \phi(s) + M(s) \begin{pmatrix} s \\ v_{\text{eq}}(s) \end{pmatrix} + Bu_{\text{eq}}(s) \tag{91}
$$

and rate bounds; for example, and rate bounds; for example, and rate bounds; for example,

$$
\frac{d}{dt}\begin{pmatrix} s \\ v \end{pmatrix} = M(s)\begin{pmatrix} 0 \\ v - v_{\text{eq}}(s) \end{pmatrix} + B(u - u_{\text{eq}}(s)) \tag{92}
$$

$$
\frac{d}{dt}v_{\text{eq}}(s) = Dv_{\text{eq}}(s)\dot{s}
$$
\n
$$
= Dv_{\text{eq}}(s)M_{12}(s)(v - v_{\text{eq}}(s)) + Dv_{\text{eq}}(s)B_1(u - u_{\text{eq}}(s))
$$
\n(93)

ces of $M(s)$ and B .

$$
\frac{d}{dt} \begin{pmatrix} s \\ v - v_{\text{eq}}(s) \end{pmatrix}
$$
\n
$$
= \begin{pmatrix} 0 & M_{12}(s) \\ 0 & M_{22}(s) - D v_{\text{eq}}(s) M_{12}(s) \end{pmatrix} \begin{pmatrix} s \\ v - v_{\text{eq}}(s) \end{pmatrix}
$$
\n
$$
+ \begin{pmatrix} B_1 \\ B_2 - D v_{\text{eq}}(s) B_1 \end{pmatrix} (u - u_{\text{eq}}(s))
$$
\n(95)

which can be written as

$$
\frac{d}{dt}\begin{pmatrix} s\\ \tilde{v} \end{pmatrix} = A_{\text{new}}(s)\begin{pmatrix} s\\ \tilde{v} \end{pmatrix} + B_{\text{new}}(s)\tilde{u} \tag{96}
$$

$$
\tilde{v}(t) = v(t) - v_{eq}(s(t)), \quad \tilde{u}(t) = u(t) - u_{eq}(s(t))
$$
\n(97)

Quasi-LPV Representation. It is also convenient to consider
the original equations now take a quasi-LPV form, where
the relationship between the scheduling variable and plant
nonlinearization approximations were made to scheduling variable enter as high order perturbations in Eq. (96). This transformation then reveals the importance of the scheduling variable to capture the plant nonlinearities.

$$
\dot{x} = \begin{pmatrix} x_1 | x_1 | + x_2 \\ x_1^2 x_2 + u \end{pmatrix}
$$
 (98)

and let $s = x_1$ be the scheduling variable. These equations take the form of Eq. (89). The resulting equilibrium family is so that

$$
0 = \phi(s) + M(s) \begin{pmatrix} s \\ v_{\text{eq}}(s) \end{pmatrix} + Bu_{\text{eq}}(s) \tag{91}
$$

Performing the transformations described above leads to the In order to convexify the problem, consider the change in quasi-LPV form variables

$$
\frac{d}{dt}\begin{pmatrix} s \\ x_2 - (-s|s|) \end{pmatrix} = \begin{pmatrix} 0 & 1 \\ 0 & s^2 - 2|s| \end{pmatrix} \begin{pmatrix} s \\ x_2 - (-s|s|) \end{pmatrix}
$$

$$
+ \begin{pmatrix} 0 \\ 1 \end{pmatrix} (u - (-s^3|s|)
$$
(100)

Because of the availability of numerically efficient methods determine definitely that no solution exists. for large scale problems, convex optimization is an emerging Some important points to consider are the following: technique for gain scheduling design of for LPV and quasi-LPV systems. As seen in the previous section, the develop- • The scheduling process is built into the construction of ment of methods for LPV systems is directly pertinent to gain the state feedback; it is *not* necessary to perform several scheduling, since LPV systems form the underlying structure fixed parameter designs. of a gain scheduled design. • Stability for arbitrarily fast parameter variations is as-

The main idea in convex optimization methods for LPV sured. systems is to combine stability and performance parameters • Theorem 22 is only a sufficient condition for stability, with controller parameters in a single convex optimization ob- and hence may be conservative.

We will demonstrate these methods in the following simple The method extends to more general control objectives, context. Consider the open loop LPV system other than stabilizing state feedback, including

$$
\dot{x} = (\theta A_1 + (1 - \theta)A_2)x + Bu \tag{101}
$$

• output feedback where the parameter is constrained by • rate constrained parameter variations

$$
0 \le \theta \le 1 \tag{102}
$$

$$
u = -(\theta K_1 + (1 - \theta)K_2)x \tag{103}
$$

$$
\dot{x} = (\theta(A_1 - BK_1) + (1 - \theta)(A_2 - BK_2))x \tag{104}
$$

A sufficient condition which guarantees the stability of Now consider the nonlinear feedback (104) is the following.

Theorem 22. The equilibrium $x_0 = 0$ of Eq. (104) is asymptotically stable if there exists a positive definite matrix, $P =$ where $P^T > 0$, such that for both $i = 1$ and $i = 2$,

$$
P(A_i - BK_i) + (A_i - BK_i)^T P < 0 \tag{105}
$$

It is important to note that Theorem 22 only provides a sufficient condition for stability. The main idea is that the matrix *P* defines the Lyapunov function, $V(x) = x^T P x$, for the is closed loop system Eq. (104).

Our objective is to find matrices K_1, K_2 , and P which satisfy Eq. (105). It can be shown that the set of matrices which satisfy Eq. (105) is *not* convex. This lack of convexity significantly complicates any direct search process.

$$
Q = P^{-1}, \quad Y_i = K_i P^{-1} \tag{106}
$$

Given $Q > 0$ and Y_i , one can solve for the original variables P and *Ki*. With these variables, condition Eq. (105) is equivalent to

$$
A_i Q - BY_i + QA_i^T - Y_i^T B^T < 0 \tag{107}
$$

ADVANCED METHODS FOR GAIN SCHEDULING Now the set of $Q > 0$, Y_1 , and Y_2 which satisfy Eq. (107) *is* **Convex Optimization for LPV Systems CONVEX CONVEX** CONVEX CONVEX CONVEX CONVEX CONVEX FEASIBLE SET ON THE SYSTEM **algorithms** which either produce a feasible set of matrices or

-
-
-

- disturbance rejection and command following
-
-

⁰ [≤] ^θ [≤] 1 (102) **Extended/Pseudo-Linearization**

We are interested in constructing stabilizing gain scheduled The objective in extended and pseudo-linearization is to imstate feedback. Let us impose the feedback structure pose that the closed loop system has a linearization family which is invariant in some desired sense.

Let us consider the special case of a tracking problem with full state feedback for the nonlinear system Eq. (33). The obwhich mimics the LPV variations of the system. The closed jective is for the first state, x_1 , to approximately track referloop dynamics are then ence commands, *r*. Let the equilibrium family ($x_{eq}(s)$, $u_{eq}(s)$) be an such that

$$
(1 \ 0 \ \ldots \ 0)x_{eq}(s) = s \tag{108}
$$

$$
u = G(x, r) \tag{109}
$$

$$
u_{\text{eq}}(s) = G(x_{\text{eq}}(s), s) \tag{110}
$$

Then, a linearization family for the closed loop system

$$
\dot{x} = f(x, G(x, r))\tag{111}
$$

$$
\dot{\tilde{x}} = (D_1 f(x_{eq}(s), G(x_{eq}(s), s)) + D_2 f(x_{eq}(s), G(x_{eq}(s), s))D_1 G(x_{eq}(s), s))\tilde{x} + D_2 f((x_{eq}(s), G(x_{eq}(s), s))D_2 G(x_{eq}(s), s)\tilde{r}
$$
\n(112)

$$
\dot{\tilde{x}} = A_{\text{des}} \tilde{x} + B_{\text{des}} \tilde{r}
$$
\n(113)

One can state appropriate conditions involving partial differ- A. Bacciotti, *Local Stabilizability of Nonlinear Control Systems*, Singa-

ential constraint equations under which there exists any G pore: World Scientific ential constraint equations under which there exists any G which achieves this objective. The extension of the K. Khalil, *Nonlinear Systems*, 2nd Ed., New York: Macmillan,

The intention is that improved closed loop performance is 1996. possible if the closed loop linearizations have some sort of invariance property. Such improved performance, if any, is dif- *Overview of Gain-Scheduling and Its Theoretical Foundations* ficult to quantify, but simulation studies indicate the poten-
tial benefits.
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Syst. Magazine, **¹¹**: 79–84, 1991. *x*˙1 *x*˙2 = *x*² *x*1*x*² + *x*² ¹ + *u* (114)

$$
x_{\text{eq}}(s) = \begin{pmatrix} s \\ 0 \end{pmatrix}, \quad u_{\text{eq}}(s) = -s^2 \tag{115}
$$

dynamics matrices 559–564, 1991.

$$
A_{\text{des}} = \begin{pmatrix} 0 & 1 \\ -1 & -2 \end{pmatrix}, \qquad B_{\text{des}} = \begin{pmatrix} 0 \\ 1 \end{pmatrix} \tag{116}
$$

$$
G(x,r) = -x_1x_2 - x_1^2 - x_1 - 2x_2 + r \tag{117}
$$

Then setting $u = G(x, r)$ leads to the linear closed loop dynam- P. Apkarian and R. J. Adams, Advanced gain-scheduling techniques

$$
\dot{x} = \begin{pmatrix} 0 & 1 \\ -1 & -2 \end{pmatrix} x + \begin{pmatrix} 0 \\ 1 \end{pmatrix} r \tag{118}
$$

ization, since the resulting closed loop dynamics are linear. 638, 1984.

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$$
\begin{pmatrix} \dot{x}_1 \\ \dot{x}_2 \end{pmatrix} = \begin{pmatrix} x_2 + x_2^2 u \\ x_1 x_2 + x_1^2 + u \end{pmatrix}
$$
 (119)

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