## **OPEN-LOOP OSCILLATORY CONTROL**

Conventional control systems rely on feedback, feedforward, or a combination of the two. In a feedback control system, the controlled variable is usually compared with a reference variable, and the difference between the two, the error, is used to regulate the system. In a feedforward control system, an appropriate additive control signal is introduced to compensate for disturbances. While feedback and feedforward rely on different principles, both methods require measurements. In a feedback control system, the controlled variable is measured. Likewise, in a feedforward control system the measurement of disturbances is used in the implementation. However, measurements of states or disturbances are often costly, difficult, or even impossible to obtain. In these cases, feedback and feedforward are not feasible means of control.

Consider the particle accelerator originally described in (1) and later discussed in detail in (2). The control objective is to focus a beam of particles along the accelerator. In cyclic accelerators with azimuth symmetrical magnetic fields, the *plant,* a beam of particles, is described by

$$
\frac{d^2x}{d\theta^2} + \omega^2 (1 - n)x = 0
$$

$$
\frac{d^2z}{d\theta^2} + \omega^2 nz = 0
$$
 (1)

where *x* and *z* are state coordinates representing small beta- where the averaged potential is used to study global structure tron oscillations of the beam,  $\omega$  is the betatron wave number, and stability of the periodically forced system. The main re- $\theta$  is an independent variable (azimuth), and *n* is the field in- sult is that for a given forcing amplitude, there is a unique dex of refraction. For proper operation, the field index of re- critical forcing frequency at which the inverted equilibrium fraction should satisfy  $0 < n <$ *n*, the beam focusing is often unacceptable. Feedback is usu- higher than the critical frequency, the inverted equilibrium is ally not possible due to the difficulty of measuring *x* and *z*. stable. Reference 10 considers a generalization of the classic

sible for such 'unmeasureable' systems. For many systems, lum tends to stabilize in configurations aligned with the direcone alternative is *open-loop oscillatory control,* sometimes re- tion of the forcing, and in fact this phenomenon holds in the ferred to as *vibrational control* (not to be confused with *vibra-* general *n*-link case as well (11). *tion control* where the idea is to *reduce* vibrations). Open-loop Open-loop oscillatory control has been applied to many sysoscillatory control is a fairly recently developed control meth- tems, and new applications continue to emerge. In (12) the odology that does not require measurements of states or dis- technique was applied to exothermic chemical reactions in a turbances. Instead, zero mean periodic excitation is used to continuous stirred tank reactor (CSTR). In this work, it was modify the plant behavior in such a way that control is shown that by modulating the input and exit chemical feed achieved as the result of the system's natural response to the rates of the CSTR, it is possible to operate in stabilized averexcitation. For example, oscillations in the cyclic accelerator aged conversion rates that would otherwise be unstable uncan be introduced by appropriately focusing and defocusing less expensive feedback is applied. Although, on average, the sectors of the magnetic lens. This causes a suppression of same amount of input chemical has been used, the stable opbetatron oscillations and thereby makes the focus beam more erating regimes of the CSTR change substantially with the acceptable. An early heuristic description of this phenomena use of an oscillatory control input. In similar work, the results was given by Livingston (1), but it was not until 1980 that the of (12) are analytically extended to include chemical reactions heuristically controlled azimuth accelerator was explained in in a CSTR with delayed recycle stream (13,14). Historically, the context of open-loop oscillatory control in (2). all of this work on oscillatory open-loop control was prompted

tion of periodic (or almost periodic) control laws, without the mization, and asynchronous quenching (12,15). use of measurements in order to induce a desired dynamic Experimental applications of open-loop oscillatory control response in a system is referred to as open-loop oscillatory have also included laser illuminated thermochemical systems control or vibrational control. (16), stabilization of plasma (17), and car parking algorithms

offset by the difficulty added in introducing explicit time de- the result in (20) that showed that periodic paths improve pendence in the state system models. In order to simplify the aircraft fuel economy. Other analytic applications of open-loop analysis, open-loop oscillatory control algorithms may restrict oscillatory control include rotating chains (21,22), *n*-link penthe control action so the controlled system admits a small pa- dula (11), axial compressors (23), and population models rameter. One way to obtain a small parameter is to introduce (24,25). periodic excitation whose frequency is an order of magnitude In the work by Lehman et al. (13,25,26) the technique of larger than the highest system natural frequency. The small oscillatory open-loop control is developed for systems with parameter will then arise as the result of a rescaling of time. time-delays. Bentsman and Hong (27,28) have extended the For such systems, the time-varying open-loop controlled sys- technique to parabolic partial differential equations (PDEs). tem can be approximated by the behavior of a time-invarient The application of open-loop control to delay systems and averaged equation, to which the usual analytical techniques PDE's shows interesting potential since these types of infinite for time-invariant systems may be applied. This result forms dimensional systems are often difficult to control when using the basis of classical averaging theory in applied mathematics feedback. Likewise, there has been success in combining the and dynamical systems. Within the context of forced mechani- benefits of open-loop oscillations with conventional feedback cal systems and averaging, energy methods and a quantity in order to robustly stabilize systems with zeros in the opencalled the *averaged potential* provide the most direct method right half plane and systems with decentralized fixed zeros of analysis. In the absence of distinct system time or length (29–32). scales, the local stability of an equilibrium or periodic orbit As with all other control algorithms, the important issues can be studied by the analysis of the linearized system's first of designing open-loop oscillatory control include stability, return map, or monodromy matrix, obtained through Flo- transient response, and accuracy of the controlled system. quet theory. Certainly, the most important issue is stability. Many classi-

tory control is the stabilization of the simple pendulum's in- oscillatory inputs depend on eigenvalues of the averaged sysverted equilibrium by high frequency vertical oscillation of tem lying in the left half plane, or equivalently the eigenvalthe pendulum's suspension point. This discovery is usually ues of the monodromy matrix lying within the unit disk. Howattributed to Bogoliubov (3,4) and Kapitsa (5), although ear- ever, there has been growing interest in the stabilization of lier references to similar phenomena exist (6). More recent systems to which such classical results do not apply. These accounts of this stabilization may be found in (7,8,9,10), include the mechanical systems studied in (7,8,10,11,33),

experiences a pitchfork bifurcation. For forcing frequencies Feedforward also has similar measurement difficulties. problem, where the periodic forcing is directed along an in-In such cases, a natural question is whether control is pos- cline with respect to the horizontal. In this case, the pendu-

by the work of periodic operation of chemical reactors using *Definition 1. Open-loop Oscillatory Control.* The utiliza- the sometimes heuristic techniques of push-pull, periodic opti-

(18). In (19), sufficient conditions are given for a periodic pro-The simplicity of open-loop oscillatory control synthesis is cess to minimize periodic paths. This approach generalized

One of the most compelling examples of open-loop oscilla- cal results on stability of operating points for systems with

spect to the imaginary axis. Coron  $(34)$  has shown the exis-stabilizing effect on the system. tence of a time-varying feedback stabilizer for systems whose In this case, Eq. (2) becomes averaged versions have eigenvalues on the imaginary axis. Additional interest in this design derives from the observation that it provides a method of smooth feedback stabiliza-

Stability of a system is concerned with the asymptotic be-<br>havior of the system. Often it is important to study trajector-<br>ies of systems as steady-state behavior is being approached.<br>Analysis of such trajectories when th trol input is a difficult task. The oscillatory control is usually designed to be high frequency. As a result, the controlled system is composed of a fast zero average oscillatory trajectory superimposed on a slow trajectory. Therefore, the designer must attempt to control the slow part of the trajectory and It is often preferable that Eq. (3) has a fixed equilibrium

curacy. It is well known that driving a nonlinear system with the technique of vibrational stabilization is to determine via periodic signal generally excites an array of resonances, and brations  $\gamma(t)$  such that the (possibly unstable) equilibrium under appropriate conditions chaos in the homoclinic tangles point *xs* bifurcates into a stable periodic solution whose averof unstable resonances [See  $(36)$  for a complete exposition on age is close to  $x_s$ . this topic]. While subharmonic resonances and chaos tend to The engineering aspects of the problem consist of: be suppressed at high forcing frequencies,  $1:1$  resonances (primary resonances, or periodic orbits), whose averages cor- 1. Finding conditions for the existence of stabilizing perirespond to fixed points of an averaged representation of the odic inputs dynamics, persist. If a stable 1:1 resonance has no associa-<br>2. Determining which oscillatory inputs,  $u(\cdot)$ , are physition with a fixed point of the time-varying system (i.e., it cally realizable and arises through a bifurcation), it is called a *hovering motion*.  $\alpha$  Determining the s

$$
\dot{x} = f(x, u) \tag{2}
$$

always be assumed sufficiently continuous so that solutions to 0 has the desired system pole location and the degree of *p* is<br>Eq. (2) exist. Models of this form describe most of the systems greater or equal to the degree Eq. (2) exist. Models of this form describe most of the systems greater or equal to the degree of *d*. At times, this pole–zero appearing in the recent engineering literature on open-loop oscillatory control, as discussed in detail in (37). tem performance, especially if there is no need for feedback

found in (24,38). Suppose that (39) has an unstable equilib-  $d(s) = 0$  with positive real part, LTI open-loop control cannot rium point,  $x_s$  when  $u = constant = \lambda_0$ , and the goal is to stabilize a system. On the other hand when rium point,  $x_s$  when  $u = constant = \lambda_0$ , and the goal is to stabilize a system. On the other hand, when  $u(t)$  is an oscilla-<br>determine a control input  $u(t)$  that stabilizes this operating tory open-loop control input stabilizati point. In addition, suppose this stabilization is to be per- even when there is a pole in the right-half plane. Indeed, os-<br>formed without any state or disturbance measurements. cillatory open-loop controls have also shown

into Eq. (2) oscillatory inputs according to the law  $u(t) =$  control.  $\lambda_0 + \gamma(t)$  where  $\lambda_0$  is a constant vector and  $\gamma(t)$  is a periodic average zero (PAZ) vector, that is,  $\gamma(t) = \gamma(t + T)$  with

where eigenvalues locations are typically symmetric with re- equal to  $\lambda_0$ , it is hoped that the periodic forcing can impose a

$$
\frac{dx}{dt} = f(x, \lambda_0 + \gamma(t))\tag{3}
$$

tion for systems which Brockett (35) had previously shown<br>were not stabilizable by smooth, time-invariant feedback.<br>Stability of a system is concerned with the asymptotic be-<br>havies of Eq. (2) is said to be vibrationally

$$
\|\overline{x}^* - x_s\| \le \delta; \quad \overline{x}^* = \frac{1}{T} \int_0^T x^*(t) dt
$$

ignore (or filter out) the high frequency component. point, *xs*. However, this is not usually the case since the right One disadvantage of open-loop oscillatory control is its ac- hand side of Eq. (3) is time varying and periodic. Therefore,

- 
- 
- arises through a biturcation), it is called a *hovering motion*. 3. Determining the shape (waveform type, amplitude, these high frequency responses limit the utility of open-loop base) of the oscillations to be inserted wh

**PROBLEMS IN OPEN-LOOP OSCILLATORY CONTROL** At this time, it may be useful to explain why it is necessary to use time-varying control inputs as opposed to simply using **Classes of Systems Classical time-invariant open-loop control techniques. Sup-** pose that there is a single-input single-output linear time-This section considers systems of ordinary differential equa- invariant (LTI) system with proper transfer function *Y*(*s*)/ tions, with inputs, of the form  $U(s) = n(s)/d(s)$ , where *Y* and *U* are the Laplace transform of the output and the input, respectively, and *n* and *d* are poly*nomials in <i>s*. If all the roots of  $d(s) = 0$  have negative real parts, then open-loop control can be used to arbitrarily place where  $x \in \mathbb{R}^n$  and  $u \in \mathbb{R}^m$ . The function  $f: \mathbb{R}^n \times \mathbb{R}^m \to \mathbb{R}^n$  will system poles simply by letting  $U(s) = d(s)/p(s)$ , where  $p(s) =$ <br>always be assumed sufficiently continuous so that solutions to 0 has the (perhaps because there are no disturbances).

Unfortunately, though, perfect pole–zero cancellation is **Stabilization** not possible. This may not be worrisome if all the roots are in Here we introduce the problem of vibrational stabilization, as the left-half plane, but when there exists at least one root of found in (24,38). Suppose that (39) has an unstable equilib-  $d(s) = 0$  with positive real part. tory open-loop control input, stabilization is often possible, formed without any state or disturbance measurements. cillatory open-loop controls have also shown a remarkable ro-<br>For the stabilization problem being considered, the meth-<br>bustness to disturbances in many experimental ap bustness to disturbances in many experimental applications ods of oscillatory open-loop control are as follows. Introduce (12,37). This is a quality that is absent in LTI open-loop

**Remark 1.** This subsection has attempted to state the prob- $\int_0^T \gamma(t) dt = 0$ . Even though the average value of *u*(*t*) remains lem of stabilization in its broadest terms. There are classes of

systems, however, for which discussion of stabilization and where, once again, *w* approximates *x*. The oscillatory openstability is problematic. Such systems include conservative loop control results in a superposition of fast oscillatory trasystems, or more specifically, Hamiltonian systems. Hamilto- jectories on slow trajectories. The slow dynamics are reprenian systems include dissipation-free mechanical systems, sented by *y*, and *h* can be a fast periodic function. In either of and include many electrical and optical systems as well. The the above two cases, it is hoped to find oscillatory control inprimary defect of Hamiltonian systems as far as control the- put *u*(*t*) such that the transient performance of *w* meets deory is concerned is that the strongest stability these systems sired objectives. Since the state equation of the approximacan possess is neutral stability; that is, eigenvalues/poles on tions are time-invariant, the analysis becomes simpler. In the imaginary axis. For this reason, standard concepts from fact, even though Eq. (5) is time-varying, it is only the output control theory seldom yield strong stability results. Progress equation which explicitly depends on *t*. Therefore, many of has recently been made in developing techniques for the sta- the well established tools can be applied directly to the state bility analysis of these systems. The new techniques make use of the system energy, and in the case of periodically forced values of matrix *A* help determine the qualitative features of systems the *averaged potential,* to assess the stability of equi- transient behavior. libriums. A technique for the equilibrium and stability analysis of a large class of periodically forced Hamiltonian systems **Steering and Path Planning for Kinematically** is presented later in this article.

Once a system is determined to be stable, the next issue in periodic functions in path generation for so-called kinemati-<br>evaluating its performance is to determine how quickly the cally nonholonomic systems. Such systems

For systems subject to oscillatory open-loop control, the analysis techniques are not so straightforward. As previously mentioned, the control inputs cause Eq. (3) to be time-varying, and analysis of time-varying systems remains an open<br>area of research. However, since it has been assumed that the<br>control inputs have a special structure, that is, periodic and<br>high-frequency, it will be possible to

Essentially the problem of controlling the transient behav-<br>in The prescribed endpoint steering problem requires that<br>is to time-varying system Eq. (2) is to given any pair of points  $x_0, x_1 \in \mathbb{R}^n$ , a vector of piece

- lutions to Eq. (3) can be approximated by the solutions to  $x_1$  at time  $t = T > 0$ .<br>of a simpler equation 2. The trajectory approximated
- 

Sometimes this simpler equation turns out to be purely time-<br>verges (uniformly) to  $\gamma$ . invariant and in the form of Several authors have suggested constructive methods for pe-

$$
\frac{dy}{dt} = P(y), \quad w = q(y) \tag{4}
$$

where w and y are both vectors in  $\mathbb{R}^n$ , and w approximates x, **STABILIZATION BY OSCILLATORY CONTROLS:**<br>the solution to Eq. (3). Often, though, a time-varying output **METHODS AND SOLUTIONS** equation is used and the approximate equation becomes **Applications of Classical Averaging Theory**

$$
\frac{dy}{dt} = P(y), \quad w = h(t, y) \tag{5}
$$

equation. In particular, when  $P(y) = Ay + B$ , then the eigen-

An application of open-loop oscillatory control which lies **Transient Behavior and Performance** largely outside the boundaries of this chapter is the use of

$$
f(x, u) = \sum_{i=1}^{m} u_i g_i(x)
$$

- analytic control inputs  $u(\cdot) = (u_1(\cdot), \ldots, u_m(\cdot))$  is to 1. Determine control inputs,  $\gamma(t)$  in Eq. (3) so that the so- be determined to steer from some state  $x_0$  at time  $t = 0$
- 2. The trajectory approximation steering problem requires 2. Control the transient behavior of the approximate that given any sufficiently regular curve  $\gamma: [0, T] \to \mathbb{R}^n$ , equation a sequence  $[u^{(1)}]$  of control input vectors is found such a sequence  $[u^{(1)}]$  of control input vectors is found such that the corresponding sequence of trajectories con-

riodic controllers in this context, and further details may be found in (37,39–44). *dy*

The goal of the open-loop oscillatory control is usually to stabilize an unstable equilibrium  $x<sub>s</sub>$  of Eq. (2). This is performed



**Figure 1.** Flow graph of typical open loop oscillatory control design procedure.

nals, such as sinusoidal inputs or zero average square waves. mation, consider the so called generating equation given as The frequency of the input is selected to be large, or equivalently, as Fig. 1 shows, the period is small. The periodic system, in the form of Eq. (3) can then be transformed into the form of Eq. (1.1) in Appendix 1, where  $\epsilon$  turns out to be proportional to the period. At this point, the transformed system Suppose that this generating equation has a period *T* general can be averaged. If the averaged system has a uniformly solution  $h(t, c)$ , for some  $\tilde{u}(\cdot)$  can be averaged. If the averaged system has a uniformly asymptotically stable equilibrium point, then this implies that there will be a uniformly asymptotically stable periodic  $x(t_0) \in \Omega \subset \mathbb{R}^n$ .<br>orbit of the transformed time-varying system in the vicinity Introduce into Eq. (7) the Lyapunov substitution  $x(t)$  = orbit of the transformed time-varying system in the vicinity Introduce into I<br>of the equilibrium point. The final criteria for vibrational sta-<br> $h(\omega t, z(t))$  to obtain of the equilibrium point. The final criteria for vibrational stabilization is that the periodic orbit satisfying Eq. (3) remain in the vicinity of  $x<sub>s</sub>$  (even though a transformation is used prior to averaging). This is the reason for introducing the definition of  $\bar{x}^*$ , which is the average value of the periodic solution of Eq. (3). If time is rescaled by letting  $\tau = \omega t$ ,  $\epsilon = 1/\omega$ , then using the

of open-loop oscillatory control laws by the classical method becomes of averaging. A brief introduction to the topic of averaging and references to more comprehensive accountings may be found in Appendix A.1. Many summaries of this procedure detailed in this section can also be found in the literature

$$
\frac{dx}{dt} = f_1(x(t)) + f_2(x(t), \gamma(t))\tag{6}
$$

where  $f_1(x(t)) = f_1(\lambda_0, x(t))$  and the function  $f_2(x(t), \gamma(t))$  is linear with respect to its second argument. Additionally, assume It is now possible to convert the averaged equation back to that  $\gamma(t)$  is periodic of period  $T$  ( $0 < T \ll 1$ ) and of the form fast time to obtain  $\gamma(t) = \omega \tilde{u}(\omega t)$ , where  $\omega = 2\pi/T$ , and  $\tilde{u}(\cdot)$  is some fixed period- $2\pi$  function. Since the primary interest is high frequency forcing, the usual implication is that the amplitude of  $\gamma(t)$  is large. It is possible, however, that  $\tilde{u}(\cdot)$  has small amplitude, making the amplitude of  $\gamma(t)$  small also.

Then Eq.  $(6)$  can be rewritten as

$$
\frac{dx}{dt} = f_1(x(t)) + \omega f_2(x(t), \tilde{u}(\omega t))
$$
\n(7)

by selecting the  $\gamma(t)$  in Eq. (3) to be periodic zero average sig- form Eq. (1.1) in Appendix 1. To make this desired transfor-

$$
\frac{dx}{dt} = f_2(x(t), \tilde{u}(t))
$$

 $\mathbb{R}^n$  and  $c \in \mathbb{R}^n$  is uniquely defined for every initial condition  $x(t_0) \in \Omega \subset \mathbb{R}^n$ .

$$
\frac{dz}{dt} = \left[\frac{\partial h(\omega t, z(t))}{\partial z}\right]^{-1} f_1(h(\omega t, z(t))\tag{8}
$$

What follows is a step-by-step procedure for the analysis standard abuse of notation of letting  $z_{\text{new}}(\tau) = z_{\text{old}}(\tau/\omega)$ , Eq. (8)

$$
\frac{dz}{d\tau} = \epsilon \left[ \frac{\partial h(\tau, z(\tau))}{\partial z} \right]^{-1} f_1(h(\tau, z(\tau)) \tag{9}
$$

(e.g. see Refs. 13,24,25,37). The following discussion is based<br>on (37).<br>Assume that f in Eq. (3) has a special structure so that Eq.<br>(3) can be rewritten as<br>(3) can be rewritten as<br>to Eq. (9) is given as<br>to Eq. (9) is gi

$$
\frac{dy}{d\tau} = \epsilon P(y(\tau)); \quad P(c) = \frac{1}{2\pi} \int_0^{2\pi} \left[ \frac{\partial h(\tau, c)}{\partial c} \right]^{-1} f_1(h(\tau, c)) d\tau
$$
\n(10)

$$
\frac{dy}{dt} = P(y(t))\tag{11}
$$

By the theory of averaging, there exists an  $\epsilon_0 > 0$  such that for  $0 \leq \epsilon \leq \epsilon_0$ , the hyperbolic stability properties of Eqs. (9) and (10) are the same. This also implies that for  $\omega$  sufficiently large, the hyperbolic stability properties of Eqs. (8) and (11) are also the same. Specifically, if  $y_s$  is an asymptotically stable equilibrium point of Eq. (11) (it will also be an asymptotically In order to proceed with the stability analysis, Eq. (7) will be stable equilibrium point of Eq. (10)), this implies that, for  $\omega$ transformed to an ordinary differential equation in standard sufficiently large, there exists a unique *T*-periodic solution, cally stable also. Furthermore, *T* is known to be equal to  $2\pi$  system in state space. This example is a slight modification  $\omega$ . Since the transform  $x(t) = h(\omega t, z(t))$  is a homeomorphism, of the problem discussed by (46). Consider the second-order there will exist an asymptotically stable *T*-periodic solution to system Eq. (7) given by  $x^*(t) = h(\omega t, \psi^*(t))$ . Equation (2) is said to be *vibrationally stabilized* provided that  $\bar{x}^* = 1/T \int_0^T x^*(t) dt$  remains in the vicinity of  $x_s$ .

**Example 1.** Oscillatory stabilization of scalar differential<br>equation where u is the scalar control. It is easy to verify that when<br>equations. Consider the scalar linear differential equation  $u = 0$  the equilibrium point

$$
x^{(n)} + (a_1 + u_1(t))x^{(n-1)} + \ldots + (a_n + u_n(t))x = 0 \qquad (12)
$$

In (45), the problem of stabilizing Eq.  $(12)$  is studied using zero average periodic control inputs in the form

$$
u_i(t) = k_i \omega \sin(\omega t + \phi_i) \quad i = 1, 2, \dots, n \tag{13}
$$

is nonexistent. Hence, assume that  $k_1 = 0$ .

This system can easily be rewritten in state space form of This system can easily be rewritten in state space form of  $\dot{x} = \dot{q} = Aq + \sum_{i=1}^{m} u_i(t)B_iq$ . However, due to the results determined in (45) there is no need for this. For sufficiently large  $\omega$  the hyperbolic stability properties of  $x_s = 0$  in Eq. (12) are the same as the hyperbolic stability properties of the equilibrium same as the hyperbolic stability properties of the equilibrium Now introduce the substitutions  $x_2 = z_2$  and  $x_1 = z_1 + \beta$ constant coefficients given by  $\omega$  to obtain

$$
y^{(n)} + (a_1 + \sigma_1)y^{(n-1)} + \ldots + (a_n + \sigma_n)y = 0 \qquad (14)
$$

where

$$
\sigma_i = \frac{k_2 k_i}{2} \cos(\phi_2 - \phi_i) \quad i = 1, 2, ..., n
$$

leads to the equation corresponding to Eq. (11) of The impact of the above result is that it presents a calculation formula for the system. Without knowledge of any periodic transformations or mathematical analysis, it is possible to select the gain and phase of each oscillatory control to stabilize the zero equilibrium of Eq. (12) based on the stability properties of Eq. (14), for sufficiently large  $\omega$ . Since all the coeffi- The eigenvalues of Eq. (18) have negative real part when  $\beta$ cients in Eq. (14) are known, the analysis becomes simple. 2.5. The equilibrium point at zero remains unchanged. There-

First, notice that since  $\sigma_1 = 0$ , this implies that the coefficient  $\epsilon > 0$ ) and for  $\beta > 2.5$  the equilibrium  $x_s = 0$  of Eq. (15) is of the  $n - 1$ th derivative in Eq. (14) cannot be changed. This vibrationally stabiliz of the  $n-1$ th derivative in Eq. (14) cannot be changed. This coefficient is equal to the negative of the sum of all system eigenvalues ( $= -trace[A]$ ). Hence, for vibrational stabilization *Example 3*. Oscillatory stabilization of a simple pendulum: to take place, it must be that  $a_1 > 0$ . Reference 45 shows this Classical Averaging. Consider a simple pendulum consisting to be a necessary and sufficient condition for scalar differen- of a massless but rigid link of l to be a necessary and sufficient condition for scalar differential equations. (In fact, for all systems  $\dot{q} = Aq + \sum_{i=1}^{m} u_i(t)B_iq$ with  $u_i(t)$  zero average, the *trace*[*A*] must always be less than zero for vibrational stabilization to be possible.) This trace pose the hinge point of the pendulum is forced to oscillate condition is never satisfied for linearized versions of the me- vertically, where the elevation of the hinge above some referchanical systems treated in the following section, indicating ence height at time  $t$  is given by  $R(t)$ . An illustration of such one direction in which the theory has been considerably ex- a system is given in Fig. 2. Accounting for Rayleigh damping tended in recent years. Next, note that  $\sigma_2$  is always positive, *b* $\dot{\theta}$  and gravitational forces, the pendulum dynamics can be and therefore, the coefficient of the  $n - 2$ th derivative in Eq. written (14) can only be increased. The quantitites  $\sigma_i$ ,  $i \geq 3$  can be made either positive or negative; however, they depend on  $k_2$ . Therefore, oscillatory control must enter through the  $a_2$ coefficient or else all  $\sigma_i$  will be zero and vibrational stabiliza- Suppose  $R(t) = \beta \sin \omega t$ . Then  $\ddot{R}(t) = -\eta \omega \sin \omega t$ , where  $\eta =$ tion will not take place.  $\eta(\omega) = \omega \beta$ . Writing Eq. (19) as a system of first order equa-

 $\psi^*(t)$  satisfying Eq. (8), in the vicinity of *y<sub>s</sub>* that is asymptoti- *Example 2.* Oscillatory stabilization of a second-order LTI

$$
\dot{x} = \left( \begin{pmatrix} 0.6 & 1.3 \\ 0.8 & -1.6 \end{pmatrix} + \begin{pmatrix} 0 & 1 \\ 0 & 0 \end{pmatrix} u(t) \right) x \tag{15}
$$

 $eigenvalues of the system$ ).

Suppose  $u(t) = \beta \omega \cos(\omega t)$ . Then

$$
\frac{dx}{dt} = \begin{pmatrix} 0.6 & 1.3 \\ 0.8 & -1.6 \end{pmatrix} x + \begin{pmatrix} 0 & \beta \omega \cos(\omega t) \\ 0 & 0 \end{pmatrix} x \tag{16}
$$

where  $k_i$  are constants. Furthermore, the results determined which is in the form of Eq. (7). The generating equation is in (45) show that the impact of the control  $u_1$  for stabilization

$$
\dot{x} = \begin{pmatrix} 0 & \beta \cos(t) \\ 0 & 0 \end{pmatrix} x
$$

which has solution  $x_2 = c_2$  and  $x_1 = c_1 + \beta \sin(t) c_2$ .

point  $y_s = 0$  of the corresponding differential equation with  $\sin(\omega t)z_2$  into Eq. (16) and convert time to  $\tau = \omega t$  with  $\epsilon = 1/2$ 

$$
\frac{dz}{d\tau} = \epsilon \begin{pmatrix} 1 & -\beta \sin(\tau) \\ 0 & 1 \end{pmatrix} \begin{pmatrix} 0.6 & 1.3 \\ 0.8 & -1.6 \end{pmatrix} \begin{pmatrix} 1 & \beta \sin(\tau) \\ 0 & 1 \end{pmatrix} z(\tau) (17)
$$

 $\sigma_i = \frac{k_2 k_i}{2} \cos(\phi_2 - \phi_i)$   $i = 1, 2, ..., n$  which is now in a form that averaging can take place. Taking the average of Eq. (17) and converting back to regular time *t* 

$$
\frac{dy}{dt} = \begin{pmatrix} 0.6 & 1.3 - 0.4\beta^2 \\ 0.8 & -1.6 \end{pmatrix} y(t) \tag{18}
$$

Some important comments on Eq. (14) need to be made. fore, for sufficiently large  $\omega$  (equivalently sufficiently small

*m* and inertia *I* is attached, and let  $\theta$  denote the counterclock-<br>wise rotation about the vertical hanging configuration. Sup-

$$
I\ddot{\theta} + b\dot{\theta} + m\ell \ddot{R}\sin\theta + mg\ell\sin\theta = 0
$$
 (19)



$$
\dot{x}_1 = 0,
$$
  

$$
\dot{x}_2 = \frac{\eta m\ell}{I} \sin t \sin x_1
$$

which has the solution

$$
\begin{aligned} x_1 &= c_1 = h_1(t, c), \\ x_2 &= -\frac{\eta m \ell}{I} \cos\,t \sin\,c_1 + c_2 = h_2(t, c) \end{aligned}
$$

$$
x_1 = z_1,
$$
  

$$
x_2 = -\frac{\eta m\ell}{I} \cos \omega t \sin z_1 + z_2
$$

$$
\dot{z}_1 = \epsilon \left[ -\frac{\eta m\ell}{I} \cos \tau \sin z_1 + z_2 \right]
$$
  
\n
$$
\dot{z}_2 = \epsilon \left[ -\left(\frac{\eta m\ell}{I}\right)^2 \cos^2 \tau \cos z_1 \sin z_1 - \frac{mg\ell}{I} \sin z_1 + \frac{\eta m\ell}{I} z_2 \cos \tau \cos z_1 + \frac{\eta m\ell b}{I^2} \cos \tau \sin z_1 - \frac{b}{I} z_2 \right]
$$

$$
\dot{y}_1 = y_2,
$$
\n
$$
\dot{y}_2 = -\frac{1}{2} \left( \frac{\eta m \ell}{I} \right)^2 \cos y_1 \sin y_1 - \frac{mg\ell}{I} \sin y_1 - \frac{b}{I} y_2
$$
\ncontrolled:

\n
$$
\frac{d}{dt} \left( \frac{1}{2} \right)^2 \cos y_1 \sin y_1 - \frac{mg\ell}{I} \sin y_1 - \frac{b}{I} y_2
$$

Notice that the averaging preserves the upper equilibrium, and  $x_s = 1/T \int_0^T h(t, y_s) dt$ . Therefore, by the previous discussion, if the inverted equilibrium is asymptotically stable for the averaged equation, then for sufficiently large forcing frequencies  $\omega$  there exists an asymptotically stable periodic orbit near the inverted equilibrium. A simple linearization of the averaged equation reveals that the stability condition for the inverted equilibrium is given by  $\omega^2 \beta^2 > 2Ig/m\ell$ .

*Remark 2.* Note that in the absence of dissipative forces that the linearized averaged system will possess eigenvalues either of the form  $\lambda_{1,2} = \pm \lambda$  where  $\lambda \in \mathbb{R}$ , or of the form  $\lambda_{1,2} =$  $\pm i\lambda$ , where  $i\lambda \in \mathbb{C}$ . Hence the system is never asymptotically stable in the absence of damping, and stability results in this case are weak. The lack of asymptotic stability is a characteristic of Hamiltonian, and more generally, conservative systems. The averaging technique of the next subsection is more suited to such systems and yields stronger stability results.

**Figure 2.** A simple pendulum whose hinge point undergoes vertical **Remark 3.** Simple nonquantitative experiments demonstrating the stabilization described in this example are not diffi-<br>motion. is shown in Fig. 3. In this experiment, the rotary motion of a tions where  $x_1 = \theta$  and  $x_2 = \dot{\theta}$ , it is clear that the first order<br>system can be written in the form of Eq. (7). Following the<br>steps detailed in the previous section, the generating equation<br>is found to be<br>is found to til the forcing frequency reaches a critical frequency at which the inverted equilibrium experiences a bifurcation which renders it stable, as depicted in the right frame of Fig. 3. The inverted equilibrium is then stable for all higher frequencies.

*Remark 4.* To this point, the main goal has been to use averaging as a means of studying the local stability properties of periodically excited systems. Under certain conditions, however, the averaged system gives far more information about the global structure of the periodically excited system. As es-Introducing the transformation sentially a perturbation technique, averaging theorems as found in  $(36,47,48)$  give no clues as to how large the small parameter  $\epsilon$  can be perturbed off zero before the averaged dynamics fail to describe the forced dynamics. For  $\epsilon$  sufficiently large, a variety of undesirable nonlinear effects arise, such as subharmonic resonance and stochasticity, which are letting  $\tau = \omega t$ , and letting  $\epsilon = 1/\omega$ , Eq. (9) specializes to not captured in any way by the simple averaging of the nonautonomous dynamics. Because of the inherent difficulty of the analysis, theory for the prediction of nonlinear effects in this range has been slow to emerge. A later section briefly illustrates some of the features of periodically excited systems exhibit when  $\epsilon$  is allowed to vary.

## **Averaging for Mechanical Systems**

Recently, interest has emerged in using high frequency oscillatory forcing to control the dynamics of mechanical systems. Therefore the averaged equations, given by Eq.  $(11)$ , are The typical applications setting is a controlled Lagrangian system where only some of the degrees of freedom are directly

$$
\frac{d}{dt}\frac{\partial \mathcal{L}}{\partial \dot{q}_1} - \frac{\partial \mathcal{L}}{\partial q_1} = u \tag{20}
$$

$$
\frac{d}{dt}\frac{\partial \mathcal{L}}{\partial \dot{q}_2} - \frac{\partial \mathcal{L}}{\partial q_2} = 0
$$
 (21)





**Figure 3.** A simple experiment to demonstrate the stabilization of the inverted equilibrium of the vertically forced pendulum. The picture on the left (a) shows the mechanism which rectifies the rotary motion of the dc motor into periodic linear motion. The picture on the The *averaged potential* given in Eq. (24) is an energy-like

where it is assumed  $dim q_1 = m$  and  $dim q_2 = n$ , and *u* is an *m*-vector of controls. (Systems of this form have been called *super-articulated* in the literature, and the reader is referred to (49) for details and references.) Within this class of models, it is further assumed that there is enough control authority to always be able to completely specify any trajectory  $q_1(\cdot)$ over an interval of interest. When this is the case,  $q_1(\cdot)$ ,  $\dot{q}_1(\cdot)$ , and  $\ddot{q}_1(\cdot)$  are viewed collectively as generalized inputs, and are used to control the dynamics of the configuration variables  $q_2(\cdot)$ . The starting point may thus be taken to be a (generalized) control system [see Fliess (50) for an introduction to generalized control systems] prescribed by a Lagrangian

$$
\mathcal{L}(q, \dot{q}; x, v) = \frac{1}{2} \dot{q}^T \mathcal{M}(q, x) \dot{q} + v^T \mathcal{A}(q, x) \dot{q} - \mathcal{V}_a(q; x, v) \quad (22)
$$

If

$$
\mathcal{L}(q_1,\dot{q}_1;q_2,\dot{q}_2)=\tfrac{1}{2}(\dot{q}_1^T,\dot{q}_2^T)\begin{pmatrix} M_{11} & M_{12}\\ M_{12}^T & M_{22} \end{pmatrix}\begin{pmatrix} \dot{q}_1\\ \dot{q}_2 \end{pmatrix}-\mathcal{V}(q_1,q_2)
$$

is the Lagrangian associated with Eqs. (20) and (21), then with the identifications  $q_1 \rightarrow x$ ,  $\dot{q}_1 \rightarrow v$ ,  $q_2 \rightarrow q$ ,  $M_{22} \rightarrow M$ ,  $M_{12} \rightarrow \mathcal{A}$ , and  $\mathcal{V}_a(q; x, v) = \mathcal{V}(x, q_{\mathcal{V}}) - \frac{1}{2}V^T M_{11} V$ , the connection between the Lagrangian dynamics prescribed by Eq. (22) and Eq.  $(21)$  is clear.

To simplify averaging, perform the usual Legendre transform  $\mathcal{H} = p\dot{q} - \mathcal{L}$ , where  $p = \partial \mathcal{L}/\partial \dot{q}$ , and write the resulting Hamiltonian in terms of the variables  $q$ ,  $p$ ;  $x$ ,  $v$ 

$$
\mathcal{H}(q, p; x, v) = \frac{1}{2}(p - \mathcal{A}^T v)^T \mathcal{M}^{-1}(p - \mathcal{M}^T v) + \mathcal{V}_a \qquad (23)
$$

This quantity is not a proper Hamiltonian since in general  $\partial \mathcal{H}/\partial t \neq 0$ . It is remarkable that if the (generalized) input functions  $x(\cdot)$ , and  $v(\cdot) = \dot{x}(\cdot)$  are restricted to be periodic and the *simple average* of  $H$  over one period is computed, (i) the resulting quantity  $\mathcal{H}$  will itself be a proper Hamiltonian, and (ii) in many cases the dynamics associated with  $H$  will closely approximate the dynamics of the nonautonomous system prescribed by Eq. (23). Recall that the *simple average* is the time average over one period of  $\mathcal{H}(q, p; x(t), v(t))$  where  $q$ and *p* are viewed as variables which do not depend on the time *t*. The averaged Hamiltonian Eq. (23) can be written

$$
\overline{\mathcal{H}}(q, p)
$$
\n
$$
= \frac{1}{2} p^T \overline{\mathcal{M}^{-1}} p - v^T \mathcal{A} \overline{\mathcal{M}^{-1}} p + \frac{1}{2} v^T \mathcal{A} \overline{\mathcal{M}^{-1}} (\overline{\mathcal{M}^{-1}})^{-1} \overline{\mathcal{M}^{-1} \mathcal{A}^T v}
$$
\n
$$
+ \frac{1}{2} v^T \mathcal{A} \overline{\mathcal{M}^{-1} \mathcal{A} v} - \frac{1}{2} v^T \mathcal{A} \overline{\mathcal{M}^{-1}} (\overline{\mathcal{M}^{-1}})^{-1} \overline{\mathcal{M}^{-1} \mathcal{A}^T v} + \overline{\mathcal{V}}
$$
\n
$$
= \frac{1}{2} (\overline{\mathcal{M}^{-1}} p - \overline{\mathcal{M}^{-1} \mathcal{A}^T v})^T (\overline{\mathcal{M}^{-1}})^{-1} (\overline{\mathcal{M}^{-1}} p - \overline{\mathcal{M}^{-1} \mathcal{A}^T v})
$$
\n
$$
= \frac{1}{2} v^T \mathcal{A} \overline{\mathcal{M}^{-1} \mathcal{A}^T v} - \frac{1}{2} v^T \mathcal{A} \overline{\mathcal{M}^{-1}} (\overline{\mathcal{M}^{-1}})^{-1} \overline{\mathcal{M}^{-1} \mathcal{A}^T v} + \overline{\mathcal{V}}
$$
\n
$$
= \frac{1}{2} v^T \mathcal{A} \overline{\mathcal{M}^{-1} \mathcal{A}^T v} - \frac{1}{2} v^T \mathcal{A} \overline{\mathcal{M}^{-1}} (\overline{\mathcal{M}^{-1}})^{-1} \overline{\mathcal{M}^{-1} \mathcal{A}^T v} + \overline{\mathcal{V}}
$$
\n
$$
= \frac{1}{2} (24)
$$

right (b) shows the pendulum stabilized in the inverted equilibrium. function of the generalized coordinates *q* which is abbrevi-

ated  $\mathcal{V}_A(q)$ . A complete understanding of the relationship between the dynamics of nonautonomous Hamiltonian systems of Eq. (23) and the appropriate counterparts for averaged Hamiltonian  $\mathscr S$  of Eq. (24) does not presently exist. There are very broad classes of such systems, however, for which it is possible to prove the validity of the following:

**Averaging Principle for Periodically Forced Hamiltonian Systems.** The dynamics associated with Eq. (23) under periodic forcing  $(x(t), v(t))$  are locally determined in neighborhoods of critical points of the *averaged potential*  $\mathcal{V}_A(q)$  as follows:

- If  $q^*$  is a strict local minimum of  $\mathcal{V}_A(\cdot)$ , then provided<br>the frequency of the periodic forcing  $(x(\cdot), v(\cdot))$  is suffi-<br>the frequency of the periodic forcing  $(x(\cdot), v(\cdot))$  is sufficiently high, the system will execute motions confined to a neighborhood of *q*\*. This equation may be derived from a Lagrangian of the form
- If  $(q, p) = (q^*, 0)$  is a hyperbolic fixed point of the corresponding averaged system (i.e., the Hamiltonian system determined by Eq.  $(24)$ , then there is a corresponding periodic orbit of the forced system such that the asympaveraged system coincide with the asymptotic stability properties of the periodic orbit for the forced system.

This type of averaging for the analysis of periodically forced mechanical systems has been treated in (7) in the case in When  $\alpha = \pi/2$ , the hinge of the pendulum undergoes the ver-<br>which *M* and *A* in Eq. (22) do not depend explicitly on the tical oscillation described in the previ case, the motion of systems defined by Eq. (23) are organized  $2Ig/m\ell$ . According to the theory of averaging presented in (7) around local minima of the averaged potential which are not and (8), the pendulum will execute ing Poincaré maps of the nonautonomous system defined by *Remark 5.* This example illustrates nonclassical behavior in Eq. (23). the case  $\alpha \neq \pi/2$ . For this case there will be, for sufficiently

Averaged potential. This example illustrates the use of the (25). Nevertheless, the pendulum will still execute motions averaged potential in analyzing the dynamics of a pendulum confined to neighborhoods of such local min averaged potential in analyzing the dynamics of a pendulum confined to neighborhoods of such local minima. For whose hinge point is forced to undergo oscillatory linear mo- on this type of emergent behavior, see (33) and ( whose hinge point is forced to undergo oscillatory linear motion which is not necessarily vertical as in the last example. Suppose  $(x, y)$  gives the coordinates of the horizontal and ver- *Remark 6.* The strategy behind the very simple (open loop) to a sliding block which is controlled to execute the oscillatory

$$
\begin{pmatrix} x(t) \\ y(t) \end{pmatrix} = \begin{pmatrix} \cos \alpha \\ \sin \alpha \end{pmatrix} \sin \omega t
$$

where  $\alpha$  prescribes the direction of the oscillatory motion, and is the frequency. This system is illustrated in Fig. 4. If, as **Floquet Theory** in the last example, the pendulum has total mass  $m$  and iner-<br>tia  $I$  about its hinge point, the motion under this oscillatory<br>forcing is described by a second order differential equation<br>forcing is described by a second

$$
I\ddot{\theta} - m\ell\omega^2 \beta \cos(\theta - \alpha) \sin \omega t + mg\ell \sin \theta = 0 \qquad (25)
$$



$$
\mathcal{L}(\theta, \dot{\theta}, v) = \frac{1}{2}I\dot{\theta}^2 + m\ell\cos(\theta - \alpha)v(t)\dot{\theta} + mg\ell\cos\theta
$$

totic stability properties of the fixed point  $(q^*, 0)$  of the where  $v(t) = \omega\beta \cos \omega t$ . The averaged potential for this system averaged average

$$
\mathcal{V}_A(\theta) = \frac{(m\ell\omega\beta)^2}{4I} \cos^2(\theta - \alpha) - mg\ell\cos\theta
$$

ory appears in (8), but this is restricted to the case in which  $[0, 2\pi)$  depending on whether or not  $\beta^2\omega^2$  is less than or larger<br>local minima of the averaged potential correspond to rest than  $2Ig/m\ell$ . Clearly the point of the nonautonomous dynamics. In the more general minimum of the averaged potential if and only if  $\beta^2\omega^2$ 

large values of  $\omega$ , strict local minima of the averaged potential *Example 4.* Oscillatory stabilization of a simple pendulum: which are *not* equilibrium points of the nonautonomous Eq.<br>Averaged potential This example illustrates the use of the (25). Nevertheless, the pendulum will stil

tical displacement of the hinge point of a pendulum attached control designs associated with the averaged potential (and to a sliding block which is controlled to execute the oscillatory more generally with systems having motion puts) is to produce robustly stable emergent behavior which is related to the critical point structure of the averaged potential. The design method for control laws in this category involves designing the averaged potential functions themselves by means of appropriately chosen inputs. The guiding theory for this approach remains very much under development.

Appendix 2, the central idea behind the theory is that the local stability of an equilibrium or periodic orbit may be determined from the eigenvalues of the *monodromy matrix M.* The monodromy matrix represents the growth or decay of solutions of the linearized system, where the linearization is about that equilibrium or periodic orbit. In general, computing the monodromy matrix is not straightforward. The calculation is relatively easy, however, if the linearization of the system state equations is piecewise constant in *t*. For example, suppose that the linearized time-varying system is

$$
\dot{x} = A(t)x
$$

where  $A(t) = A_1$  on  $0 \le t < t'$ , and  $A(t) = A_2$  on  $t' \le t < T$ , such that  $A_1A_2 = A_2A_1$ . Then the monodromy matrix *M* can be obtained by computing the state transition matrix  $\Phi$  on the interval [0, *T*]; that is

$$
\Phi(T, 0) = \Phi(t', 0)\Phi(T, t')
$$
  
=  $e^{\int_0^{t'} A_1 dt} e^{\int_{t'}^{T} A_2 dt}$   
= M

While in mechanical problems the assumption of piecewise constant forcing is somewhat nonphysical, such approximalem in the analysis electrical and electronic systems, where such forcing is common.

*Example 5.* Oscillatory stabilization of a simple pendulum: Floquet theory. In the previous examples, it was shown that the inverted pendulum may be stabilized with high frequency and the usual stability condition is that both eigenvalues<br>vertical oscillation by averaging the time-varying equations must lie within the unit disk in the compl vertical oscillation by averaging the time-varying equations must lie within the unit disk in the complex plane. Given the and studying the stability of the inverted equilibrium with form of the characteristic polynomial, and studying the stability of the inverted equilibrium with the averaged system. In this example, stabilization is inferred for stability is trace  $\Phi(0, 1/\omega) \le 2$ , which in this case may be by linearizing the pendulum dynamics about the inverted written by linearizing the pendulum dynamics about the inverted equilibrium and studying the eigenvalues of the monodromy matrix. To facilitate the calculation, assume that the pendulum is forced by square wave forcing, that is, in Eq. (19)  $\ddot{R}(t) = \omega^2 \beta u(\omega t)$  where  $u(t) = u(t + 1/\omega)$  is a square wave which switches periodically between  $+1$  and  $-1$ . Also, assume that the damping term  $b = 0$ . This assumption simplifies the be approximated by expanding the trigonometric functions in following calculations and more importantly allows us to Taylor series around zero and solving a truncated inequality show that pendulum stabilization does not require dissipa- for  $\omega$  in terms of  $\beta$ . Note beforehand that in doing so, it has

$$
\begin{pmatrix} \dot{\theta}_1 \\ \dot{\theta}_2 \end{pmatrix} = \begin{pmatrix} 0 & 1 \\ \frac{1}{I} [mg\ell - m\ell\omega^2 \beta u(\omega t)] & 0 \end{pmatrix} \begin{pmatrix} \theta_1 \\ \theta_2 \end{pmatrix}
$$

$$
\begin{split} &\Phi(0,1/\omega)\\ &=\Psi(0,1/2\omega)\Psi(1/2\omega,1\omega)\\ &=\begin{pmatrix} \cos\frac{\Omega_1}{2\omega} & \frac{1}{\Omega_1}\sin\frac{\Omega_1}{2\omega}\\ -\Omega_1\sin\frac{\Omega_1}{2\omega} & \cos\frac{\Omega_1}{2\omega} \end{pmatrix} \begin{pmatrix} \cosh\frac{\Omega_2}{2\omega} & \frac{1}{\Omega_2}\sinh\frac{\Omega_2}{2\omega}\\ \Omega_2\sinh\frac{\Omega_2}{2\omega} & \cosh\frac{\Omega_2}{2\omega} \end{pmatrix}\\ &=\begin{pmatrix} \cos\frac{\Omega_1}{2\omega}\cosh\frac{\Omega_2}{2\omega}+\frac{\Omega_2}{\Omega_1}\sin\frac{\Omega_1}{2\omega}\sinh\frac{\Omega_1}{2\omega}\\ -\Omega_1\sin\frac{\Omega_1}{2\omega}\cosh\frac{\Omega_2}{2\omega}+\Omega_2\cos\frac{\Omega_1}{2\omega}\sinh\frac{\Omega_2}{2\omega}\\ \frac{1}{\Omega_2}\cos\frac{\Omega_1}{2\omega}\sinh\frac{\Omega_2}{2\omega}+\frac{1}{\Omega_1}\sin\frac{\Omega_1}{2\omega}\cosh\frac{\Omega_2}{2\omega}\\ \cos\frac{\Omega_1}{2\omega}\cosh\frac{\Omega_2}{2\omega}-\frac{\Omega_1}{\Omega_2}\sin\frac{\Omega_1}{2\omega}\sinh\frac{\Omega_1}{2\omega} \end{pmatrix} \end{split}
$$



**Figure 5.** Regions of stability (darkened) and instability (light) for the vertically forced simple pendulum. In this figure,  $m = g = \ell =$  $I = 1$ .

 $\Omega_1^2 = -1/I \, [mg\ell - m\ell\omega^2\beta] > 0$  and  $\Omega_2^2 = 1/I \, [mg\ell + m\ell\omega^2\beta]$  $m\ell\omega^2\beta$ ] > 0. Stability of the fixed point may be determined tions are often useful when the forcing frequency is suffi-<br>from the eigenvalues  $\lambda_1$ ,  $\lambda_2$  of  $\Phi(0, 1/\omega)$ , which are the roots of<br>ciently large. Piecewise constant forcing is often not a prob-<br>the characteristic pol

$$
\lambda^2 - \bigg[\bigg(\frac{\Omega_2}{\Omega_1}-\frac{\Omega_1}{\Omega_2}\bigg)\sin\frac{\Omega_1}{2\omega}\sinh\frac{\Omega_2}{2\omega}+2\cos\frac{\Omega_1}{2\omega}\cosh\frac{\Omega_2}{2\omega}\bigg]\lambda+1=0
$$

$$
2\cos{\frac{\Omega_1}{2\omega}}\cosh{\frac{\Omega_2}{2\omega}}+ \left(\frac{\Omega_2}{\Omega_1}-\frac{\Omega_1}{\Omega_2}\right)\sin{\frac{\Omega_1}{2\omega}}\sinh{\frac{\Omega_2}{2\omega}}<2
$$

The boundary between regions of stability and instability may tion. Linearizing Eq. (19) about the inverted equilibrium gives been implicitly assumed that  $\Omega_1/2\omega \sim 0$  and  $\Omega_2/2\omega$  which im-<br>rise to the linear system bigger of the stability regions plies  $\omega$  is large. For the present example, the stability regions displayed in Fig. 5 have been obtained by numerically calculating  $\lambda_1$  and  $\lambda_2$  over the indicated ranges of  $\beta$  and  $\omega$ .

*Remark 7.* Note that in the absence of damping,  $\Phi(0, 1/\omega)$  is Because the input  $u(t)$  is piecewise constant, the state transi-<br>tion matrix  $\Phi(t, 0)$  over one period of  $u(t)$  may be computed constrained to either lie on the real axis such that  $\lambda_0 = 1/\lambda_1$ . tion matrix  $\Phi(t, 0)$  over one period of  $u(t)$  may be computed constrained to either lie on the real axis such that  $\lambda_2 = 1/\lambda_1$ , as follows: or lie on the unit disk in the complex plane. In this case, stability results are fairly weak (lack of asymptotic stability), but are typical of Hamiltonian, and more generally, conservative systems. In the presence of dissipation, the eigenvalues may occur as conjugate pairs inside the unit disk, implying asymptotic stability.

> **Remark 8.** Note that results obtained through Floquet theory are strictly local results. In the case where the controlled system is nonlinear, proofs of stability *and instability* are strictly for the equilibrium or periodic orbit, and no information is given about the asymptotic behavior of solutions nearby. This remark is not surprising for neutrally stable equilibria, but it is in fact sometimes true for unstable equilibria as well. As an example, the reader is referred to the parametrically excited pendulum example later in this article,

## **PERFORMANCE IMPROVEMENTS AND TRANSIENT BEHAVIOR**

Sometimes stability in a system is not in question, but certain<br>performance specifications are the design constraints. This<br>section begins by revisiting the particle accelerator. Then a<br>nethod to analyze the transient beh

*Example 6.* Suppression of betatron oscillations in cyclic accelerators. Consider the cyclic accelerator subject to oscilla- • By the method of averaging, for sufficiently large  $\omega$ , the tions of focusing and defocusing sectors of the magnetic lens solutions of Eqs. (8) and (11) remain arbitrarily close to given in Eq. (1). The work of (1) describes an experimental each other provided they have the same initial condimethod of betatron oscillations by alternating gradient focus- tions. That is,  $y(t) \approx z(t)$  for all time. ing. To make the focusing improve, it is desirable that both • The solution to Eq. (7) is given by  $x(t) = h(\omega t, z(t))$ , the  $\omega^2(1 - n)$  and the  $\omega^2$  $0 \le n \le 1$ . Clearly, though, if *n* is viewed as the control vari- is a homeomorphism. able, then it is not possible to both increase and decrease *n* and  $1 - n$  simultaneously. Instead, the technique of focusing Therefore, the quantities in Eq. (5) can be selected so that and defocusing a lens is introduced, which, according to (2) is  $h(t, c)$  is the solution to the gene and defocusing a lens is introduced, which, according to (2) is  $h(t, c)$  is the solution to the generating equation and *P* is as<br>modelled defined in Eq. (11) Now define  $w = h(\omega t, v(t))$  where  $v(t)$  is

$$
\frac{d^2x}{d\theta^2} + \omega^2 [1 - (n + K(\Omega, \theta))]x = 0
$$
  

$$
\frac{d^2z}{d\theta^2} + \omega^2 [n + K(\Omega, \theta)]z = 0
$$
\n(26)

where *K* is a PAZ function with frequency  $\Omega$ ; for example, two different manners:

 $\theta$  =  $\beta \Omega$  sin( $\Omega \theta$ ), that is,  $\alpha(\Omega) = \omega \Omega$ , where  $\beta$  is a constant.<br>
the behavior of  $x(t)$  through the direct relation  $x(t) \approx$ <br>  $w(t)$ . This technique predicts many of the fast oscilla-Define  $x_1 = x$ ,  $x_2 = \dot{x}$ ,  $x_3 = z$  and  $x_4 = \dot{z}$ . Then the state space  $w(t)$ . This technique predicts representation of Eq. (26) becomes tory parts of the trajectory x.

$$
\frac{dx_1}{d\theta} = x_2
$$
\n
$$
\frac{dx_2}{d\theta} = -\omega^2 (1 - n)x_1 + \beta \omega^2 \Omega \sin(\Omega \theta) x_1
$$
\n
$$
\frac{dx_3}{d\theta} = x_4
$$
\n
$$
\frac{dx_4}{d\theta} = \omega^2 nx_3 - \beta \omega^2 \Omega \sin(\Omega \theta) x_3
$$
\n(27)

becomes In either of the two methods, controlling  $y(t)$  in Eq. (11) gov-

$$
\frac{d^2y}{d\theta^2} + \omega^2 (1 - n + 0.5\beta^2) y = 0
$$
  

$$
\frac{d^2\zeta}{d\theta^2} + \omega^2 (n + 0.5\beta^2) \zeta = 0
$$
 (28)

where in Fig. 8(b) it is seen that while the origin is unstable. It is seen that the net effect was to increase both the orbits which pass arbitrarily close to the origin are in fact  $\omega^2(1 - n)$  and the  $\omega^2n$  terms, which as a result, improved sysbounded by KAM (for Kolmogorov, Arnold, and Maser) tori tem performance by suppressing the betatron oscillations in and subharmonic resonance bands. Understanding when both the *x* and *z* directions (this simplified model did not take proofs of instability obtained by Floquet theory imply un- into account the interaction of oscillations along *x* and *z*). bounded solutions for such systems is a current topic of re- Hence, performance has been improved via the introduction search.  $\Box$  of oscillatory open-loop control, provided that  $\Omega$  is sufficiently large. This example demonstrates that the benefits of introducing oscillations can, at times, help system performance even when stabilization is not an issue.

## **Analysis Method For Transient Behavior**

cinity of the equilibrium point of Eq. (2),  $x_s$ , with  $u =$ *constant* =  $\lambda_0$ .

- 
- where  $z(t)$  is the solution to Eq. (8). This transformation

defined in Eq. (11). Now define  $w = h(\omega t, y(t))$  where  $y(t)$  is the solution to  $\dot{y}(t) = P(y(t))$  in Eq. (11). For sufficiently large  $\omega$ .

$$
x(t) \approx h(\omega t, y(t)) = w(t)
$$

It is possible to analyze the transient behavior of Eq. (7) in

- $K(\Omega, \theta) = \alpha(\Omega) \sin \Omega \theta$ .<br>Suppose that the oscillatory control input is given by  $K(\Omega,$ <br> $\theta = \rho \Omega \sin(\Omega \theta) + \text{hot} \sin \phi(\Omega) = \rho \Omega$  where  $\theta$  is a constant the behavior of  $x(t)$  through the direct relation  $x(t) \approx$ 
	- 2. Analyze a moving average of  $x(t)$ , given by  $\overline{x}(t)$  where

$$
\overline{x}(t) \equiv H(z(t)); \quad H(c) \equiv \frac{1}{T} \int_0^T h(\lambda, c) d\lambda \tag{29}
$$

and *T* is the period of  $h(t, \cdot)$ . This is done by approximating  $\bar{x}(t) \approx H(y(t))$  and once again, analyzing the transient behavior of  $\nu$  (making this technique more typical of Eq. (4)). Since the fast dynamics in *h* are averaged out, this technique introduces some error. On the which is precisely the form of Eq. (7). Using the previously other hand, since *H* does not explicitly depend on *t*, the analysis becomes simpler. discussed techniques, the equation corresponding to Eq. (11)

erns how to control *x*(*t*).

*Example 7.* Transient behavior of the vertically oscillating pendulum. Referring back to Example 3 and using the same notation, it is possible to now approximate the trajectories of

 $\theta$ , the angular position, and  $\dot{\theta}$ , the pendulum's angular veloc- where ity. The averaged equations of the system are

$$
\dot{y}_1 = y_2
$$
\n  
\n $\dot{y}_2 = -\frac{1}{2} \left( \frac{\eta m \ell}{I} \right)^2 \cos y_1 \sin y_1 - \frac{mg \ell}{I} \sin y_1 - \frac{b}{I} y_2$ 

$$
x_1 = z_1
$$
  

$$
x_2 = -\frac{\eta m\ell}{I} \cos \omega t \sin z_1 + z_2
$$

behavior of the vertically oscillating pendulum using one of the two following methods: the lower right figure it is seen that the Method 1 approxima-

$$
\dot{\theta} \approx -\frac{\eta m\ell}{I} \cos \omega t \sin y_1 + y_2
$$

*Method 2.* The estimate on  $\bar{\theta}$  is given by  $\bar{\theta} \approx \gamma_1$ . The estimate on  $\bar{\theta}$  is given by

$$
\overline{\dot{\theta}} \approx H(\nu)
$$

$$
H(y) = \frac{1}{2\pi} \int_0^{2\pi} \left[ -\frac{\eta m\ell}{I} \cos s \sin y_1 + y_2 \right] ds
$$
  
=  $y_2$ 

Note that the approximation obtained by Method 2 is merely The transformations utilized are the averaged system. The approximation obtained by Method 1, as well as the transient solution obtained from the original equation of motion, are compared to the averaged solution in Fig. 6. It is clear from the figure that the averaged system accurately captures the averaged behavior of the system as it stabilizes. The phase plot in the lower left of the figure shows Therefore, it is possible to obtain estimates on the transient some discrepancies between the trajectory obtained by behavior of the vertically oscillating pendulum using one of Method 1 with the trajectory of the original tion of  $\dot{\theta}$  is actually quite close to the velocity of the original *Method 1.* The estimate on  $\theta$  is given by  $\theta \approx y_1$ . The esti- system. As Fig. 6 shows, Method 1 is more accurate than mate on  $\ddot{\theta}$  is given by  $\qquad \qquad$  Method 2. However, Method 1 utilizes a time-varying output equation, making it a more complicated technique.

# **RESONANCE PHENOMENA IN**

Resonances arise as the natural result of subjecting any system to periodic excitation. The resonances produced vary



**Figure 6.** A comparison of the behaviors of the averaged/Method 2 system, the Method 1 system, and the original system. In these plots,  $m = g = \ell = I = 1, b = 0.5, \beta = 0.2$ , and  $\omega = 10$ .



 1. **Figure 7.** The periodically forced spring–mass system (left) and the parametrically excited pendulum (right).

simple resonance produced by exciting the spring-mass sys-<br>every forcing the spring method by exciting the spring method by the parameteric method in the change portant by<br>indifferent from the Fig. 7 is different from the completely outside the boundaries of conventional control the-<br>org. In this section, only broad concepts and observations are<br>the lower frames [(e, f),  $\omega = 7$ , 9], the general trend is clearly<br>presented in the form of an presented in the form of an example, and the reader is re-<br>ferred to such texts as  $(36.51.52)$  for extensive overviews of pressed as  $\omega$  increases. In Fig. 8(b), the origin is unstable, ferred to such texts as  $(36,51,52)$  for extensive overviews of the field. and this instability would be predicted by a Floquet analysis

pendulum dynamics. To illustrate the kind of responses which is the minimum radius at which subharmonic resonance<br>might arise in subjecting a nonlinear system to periodic exci-<br>hands exist. Progressing from Figs. 8(b) to ( tation, consider the parametrically excited pendulum (PEP) the inner resonance bands are being pushed out towards the with no damping, which is often described by the differential separatrix as frequency increases. As a co with no damping, which is often described by the differential separatrix as frequency increases. As a consequence, the re-<br>equation in which requies quasipariodic flow dominates increases

$$
\ddot{q} + (1 + \gamma \sin \omega t) \sin q = 0 \tag{30}
$$

**Remark 9.** Note that after a rescaling of time  $\tau = \omega t$ , Eq. manifolds of the two periodic points. The significance of this (30) takes the form observation is that as frequency increases, the set of initial

$$
q'' + \left(\frac{1}{\omega^2} + \frac{\gamma}{\omega^2} \sin \tau\right) \sin q = 0
$$

which, if  $\alpha = 1/\omega^2$  and  $\beta = \gamma/\omega^2$ , is recognizable as a nonlinear version of Mathieu's equation

$$
q'' + (\alpha + \beta \sin t)q = 0
$$

Note also that by letting  $\gamma = \omega^2 \beta$ , Eq. (30) is merely the equation of the motion of the vertically forced pendulum discussed in Examples 3, 4, and 5. In this example, however, it is assumed that  $\gamma$  < 1.

The natural tool for visualizing the global dynamics of single-degree-of-freedom periodically forced system is the Poinbased on the type of system being excited; for example the caré map, where Poincaré sections are taken at the end of simple resonance produced by exciting the spring-mass sys-<br>every forcing period. Features of the phase po

of the origin. In Figs.  $8(c)$ –(f), the origin is stable, as indicated *Example 8.* Qualitative features of parametrically excited by the presence of KAM tori. What varies with the excitation pendulum dynamics. To illustrate the kind of responses which is the minimum radius at which subharmon bands exist. Progressing from Figs. 8(b) to (f), observe that gion in which regular quasiperiodic flow dominates increases.

> In addition to pushing subharmonic resonance bands out, increasing frequency also has the effect of reducing the area of lobes formed by the intersection of stable and unstable conditions which are transported out of the region between the separatrices of the averaged system decreases. This is an important observation, because the averaged phase portrait



**Figure 8.** Poincaré maps showing separatrix splitting and resonance bands for the parametrically excited pendulum. The phase portrait for the unperturbed simple pendulum is shown in the upper left frame, with Poincaré maps of the forced system to the right and underneath. Forcing parameters used in the Poincaré maps are indicated in each plot. Note that the Poincaré maps of the forced system more closely resemble the averaged phase portrait as the forcing frequency becomes large.

provides no information whatsoever on the existence of a set *Remark 10.* The example gives somewhat anecdotal eviindicated by the averaged phase portrait in (a). vertically forced rotating chain (33), and in an entire class of

of initial conditions which lies inside the separatrices but are dence that choosing a sufficiently high forcing frequency transported out. In (f), it is unlikely that any subharmonic tends to suppress the negative features of periodic excitation. resonances exist, and the stable and unstable manifolds of the This has also been found to be the case in the cart and penduperiodic points have closed to form a barrier to transport as lum problem described in Examples 3, 4, and 5 [see (10)], the

to other classes of single-degree-of-freedom systems and given by multi-degree-of-freedom systems.

*Remark 11.* It has also been observed that dissipation tends to have beneficial effects beyond guaranteeing the asymptotic where  $f_0: \mathbb{R}^n \to \mathbb{R}^n$  is defined stability of certain equilibriums. Dissipation generally has the effect of breaking phase space separatrices and imposing  $f_0(y) \equiv \frac{1}{T}$ <br>hyperbolicity on systems with elliptic structures. As a result, elliptic structures, such as KAM tori and resonance bands, are destroyed and initial conditions which, in the absence of and where *y* is treated as a constant in the integration. For dissipation, belong to a KAM torus or resonance limit on a sufficiently small  $\epsilon$ , solutions of Eq. (1.2) provide good approxfixed point or periodic orbit. In addition, with sufficient dissi- imations to solutions of Eq. (1.1). Since there are many mathpation, intersecting stable and unstable manifolds completely ematical tools that can be used to analyze and control the separate, giving rise to distinct basins of attraction. As with time-invariant system Eq. (1.2), the problem of determining frequency, the extent to which dissipation helps eliminate un- the behavior of time-varying periodic system Eq. (1.1) has desirable nonlinear effects is largely dependent on its magni- been greatly simplified. tude. In (10), it was seen for the cart and pendulum problem Specifically, if it is assumed that  $x(t_0) = y(t_0)$  then for suffiof Example 4 that there exists a minimum damping coeffi- ciently small  $\epsilon$ , the following statements hold: cient such that for all values less than this minimum value, manifold intersections and resonances persist. Recent work  $\cdot$  On any finite time interval, assuming the same initial has suggested that the same issue arises in multi-degree-of-<br>conditions at time  $t_0$ , the solutions t has suggested that the same issue arises in multi-degree-of-<br>freedom systems.<br>Then then the solutions at time  $t_0$ , the solutions to Eqs. (1.1) and (1.2)<br>freedom systems.

*Remark 12.* Unfortunately, there is no known generally ap- the limit. plicable rule-of-thumb for deciding what constitutes a suffi-<br>
ciently large forcing frequency. Experiments, simulation, and<br>
preliminary analysis suggest that for many systems, a rule of<br>
thumb might be for the forcing f magnitude larger than the largest natural frequency of the As  $\epsilon$  becomes smaller, then the approximation becomes controlled system. This rule of thumb presents a problem for better and tends to zero in the limit. many-degree-of-freedom systems like rotating chains or inn-<br>nite dimensional systems like strings and beams, where natural<br>ral frequencies tend to be very large. In addition, there exist<br>counterexamples where the averaged Poincaré map resemble each other for small forcing frequen-<br>cies, but possess completely different features at high fre-<br>quencies. These topics represent the current focus of much of<br>eraging, the reader is referred to  $(36$ the research in this field.

Brad Lehman gratefully acknowledges the support of the National Science Foundation through an NSF President Faculty Fellowship, grant CMS 9596268. John Baillieul would like to where  $x \in \mathbb{R}^n$ ,  $A(t): \mathbb{R}^{n \times n} \times \mathbb{R} \to \mathbb{R}^{n \times n}$ , and  $A(t + T) = A(t)$ .<br>express gratitude for support from the United States Air

The classical method of averaging was originally developed is, for periodic systems of the form

$$
\dot{x} = \epsilon f(t, x) \tag{1.1}
$$

+  $T$ , ·) =  $f(t, \cdot)$  and 0 <  $\epsilon \ll 1$ . For simplicity, assume that f has continuous second matrix exponential of a constant  $n \times n$  matrix B. By the peripartial derivatives in its second argument. Classical averag-

periodically forced single-degree-of-freedom systems (33). ing theory addresses the relationship between the original, Current work revolves around extending this understanding time-varying Eq. (1.1) and the autonomous averaged system

$$
\dot{y} = \epsilon f_0(y) \tag{1.2}
$$

$$
f_0(y) \equiv \frac{1}{T} \int_0^T f(s, y) \, ds
$$

- remain close to each other. As  $\epsilon$  becomes smaller, then the approximation becomes better and tends to zero in
- 
- 

## **APPENDIX 2. FLOQUET THEORY**

**ACKNOWLEDGMENTS** Floquet theory is concerned with local stability for systems of the form

$$
\dot{x} = A(t)x \tag{2.1}
$$

Express grantude for support from the Office States All<br>Force Office of Scientific Research under grant F49620-96-1-<br>0059.<br>Buch systems arise as linearizations of Eq. (2) around an<br>equilibrium or periodic orbit. As describ which Floquet theory is built is that the fundamental matrix **APPENDIX 1. CLASSICAL AVERAGING THEORY** solution of Eq. (2.1) can be written as the product of a periodic matrix  $P(t)$  and a constant exponential growth or decay; that

$$
\Phi(t) = P(t)e^{Bt}
$$

where  $\Phi(t)$  is a fundamental matrix associated with Eq. (2.1), *P*(*t*) is a *n*  $\times$  *n* matrix periodic in *t* of period *T*, and  $e^{Bt}$  is the odicity of  $A(t)$ , if  $\Phi(t)$  is a fundamental matrix, so is  $\Phi(t + T)$ 

$$
\Phi(t+T) = \Phi(t)M
$$

where  $M = e^{BT}$  and *B* is a constant  $n \times n$  matrix. Without loss  $AIChE J$ ., **33**: 353–365, 1987.<br>of generality, let  $t = 0$  and  $\Phi(0) = I$  where *I* is the  $n \times n$  13. B. Lehman, Vibrational control of time delay systems. In

$$
\Phi(T) = IM = M
$$

and therefore *M* represents the rate of growth or decay of the *Anal. & Appl.,* **193**: 28–59, 1995.<br>solution. Stable solutions decay or at least remain bounded. Solution. Stable solutions decay or at least remain bounded.<br>
Hence, the condition for an equilibrium to be stable is that all<br>
the eigenvalues  $\lambda_i$  of M satisfy  $|\lambda_i| \le 1$ . M itself is called the<br>
monodromy matrix, and

quet multipliers. Complex numbers  $\xi_i$  such that  $\lambda_i = e^{\xi T}$  are<br>called Floquet exponents.<br>Floquet theory is a very classical method, and there is a<br>vast literature describing the basic theory and applications.<br>Vast. Me in the study of Hill's equation and Mathieu's equation, which,<br>because of their second order structure, bear special relevance<br>in the stability of periodically forced mechanical systems<br> $\frac{20}{3}$ . J. L. Speyer, Nonoptimal in the stability of periodically forced mechanical systems.  $^{20. J. L. Speyer, Nonoptimality of steady-state cruise for aircraft.$ <br>Hill's equation is the topic of (55), and various classic applica-<br>tions to (electrical) engineering are given in (56). References 21. 21. S. Weibel and J. Baillieul, Averaging and energy methods for ro-<br>(47) and (57) give comprehensive summaries of the funda. bust open-loop control of mechanical systems. To appear in Es-(47) and (57) give comprehensive summaries of the funda-<br>mental theory as well as examples.<br>mental theory as well as examples.

## 1997. **BIBLIOGRAPHY**

- 
- 2. S. M. Meerkov, Principle of vibrational control: theory and appli-<br>cations *IEEE Trans Autom Control* **AC-25**: 755–762, 1980 24. R. Bellman, J. Bentsman, and S. M. Meerkov, Vibrational control
- 3. N. N. Bogoliubov, Perturbation theory in nonlinear mechanics,<br>St. Stroit. Mekh. Akad. Nauk Ukr. SSR, 14: 9–34, 1950.<br>A Mitropoldur Acumptatic Mathodoxia 25. B. Lehman, J. Bentsman, S. V. Lunel, and E. I. Verriest, Vibra
- 
- 
- **AC-37**: 1576–1582, 1992. 6. A. Stephenson, On induced stability, *Phil. Mag.,* (17): 765–766,
- 7. J. Baillieul, Stable average motions of mechanical systems sub-<br>ject to periodic forcing. In M. J. Enos (ed.), Dynamics and Control Trans. Autom. Control, **AC-36**: 501–507, 1991.<br>of Mechanical Systems: The Falling Cat a *of Mechanical Systems: The Falling Cat and Related Problems:*
- 1607, 1993.<br>
1607, 1918.<br>
1710x 29. S. Lee, S. Meerkov, and T. Runolfsson, Vibrational feed 1995. Special Issue on the Control of Nonlinear Mechanical Systems. **AC-32**: 604–611, 1987.
- motions of rapidly forced mechanical systems. In *Proc. 34th IEEE* delay systems. In *Proc. 34th IEEE CDC,* 936–941, 1995. *CDC, New Orleans,* 533–539, 1995. 31. K. Shujaee and B. Lehman, Vibrational feedback control of time
- rapidly forced cart and pendulum, *Nonl. Dyn.,* **13**: 131–170, 1997. 1997.
- 11. S. Weibel, J. Baillieul, and B. Lehman, Equilibria and stability *CDC,* pp. 1147–1152, San Diego, 1997.
- 12. A. Cinar, J. Deng, S. Meerkov, and X. Shu, Vibrational control of an exothermic reaction in a CSTR: theory and experiments.
- of generality, let  $t = 0$  and  $\Phi(0) = I$  where I is the  $n \times n$  13. B. Lehman, Vibrational control of time delay systems. In J. Wiendentity matrix. Then<br>identity matrix. Then Research Notes in Mathematical Series 272. Harlo gland: Longman Scientific & Technical, 1992.
	- 14. B. Lehman, I. Widjaya, and K. Shujaee, Vibrational control of chemical reactions in a CSTR with delayed recycle, *J. Math.*
	-
	-
	-
	-
	-
	-
	-
	- 22. S. Weibel and J. Baillieul, Oscillatory control of bifurcations in rotating chains. In *Proc. 1997 ACC, Albuquerque,* 2713–2717,
- 23. J. Baillieul, S. Dahlgren, and B. Lehman, Nonlinear control de-1. M. S. Livingston, *High Energy Accelerators,* New York: Wiley- signs for systems with bifurcations and applications to stabilization and control of compressors. In *Proc. 34th CDC, New Orleans,* Interscience, 1954.
	- cations, *IEEE Trans. Autom. Control*, **AC-25**: 755–762, 1980. <sup>24</sup>. R. Bellman, J. Bentsman, and S. M. Meerkov, Vibrational control<br>of nonlinear systems: Vibrational stabilizability, *IEEE Trans. Au*
- 4. N. N. Bogoliubov and Y. A. Mitropolsky, Asymptotic Methods in  $25$ . B. Lehman, J. Bentsman, S. V. Lunel, and E. I. Verriest, Vibra-<br>the Theory of Nonlinear Oscillators, International Monographs on<br>Advanced Mathematics
- 5. P. L. Kapitsa, Dynamic stability of a pendulum with a vibrating  $\frac{26}{5}$ . B. Lehman and J. Bentsman, Vibrational control of innear time<br>point of suspension, Zh. Ehksp. Teor. Fiz., 21 (5): 588–598, 1951. In a systems
	- 1909.<br>1909. 27. J. Bentsman and K.-S. Hong, Vibrational stabilization of nonlin-<br>1909. 27. J. Bentsman and K.-S. Hong, Vibrational stabilization of nonlin-<br>1909.
	- *Fields Institute Communications.* 1–23, Providence, R.I.: Ameri- brationally controlled nonlinear parabolic systems with neumann can Mathematical Society, 1993. boundry conditions, *IEEE Trans. Autom. Control,* **AC-38**: 1603–
	- with oscillatory inputs, *Int. J. Robust Nonl. Cont.*, 285–301, July 29. S. Lee, S. Meerkov, and T. Runolfsson, Vibrational feedback con-<br>1995. Special Issue on the Control of Nonlinear Mechanical trol: zero placement capa
- 9. S. Weibel, J. Baillieul, and T. J. Kaper, Small-amplitude periodic 30. K. Shujaee and B. Lehman, Vibrational feedback control of time
- 10. S. Weibel, T. J. Kaper, and J. Baillieul, Global dynamics of a delay systems, *IEEE Trans. Autom. Control,* **AC-42**: 1529–1545,

### **166 OPERATIONAL AMPLIFIERS**

- 32. L. Trave, A. M. Tarras, and A. Titli, An application of vibrational 53. P. Lochak and C. Meunier, *Multiphase Averaging for Classical*
- 33. S. Weibel, *Applications of Qualitative Methods in the Nonlinear* 54. J. A. Sanders and F. Verhulst, *Averaging Methods in Nonlinear* ton University, 1997. lin: Springer-Verlag, 1985.
- 34. J. M. Coron, Global asymptotic stabilization for controllable sys- 55. W. Magnus and S. Winkler, *Hill's Equation. Tracts of Mathemat*tems without drift, *Math. Control, Signals, Syst.,* **5**: 295–312, *ics.* vol. 20, New York: Interscience Publishers, 1966.
- 35. R. W. Brockett, Asymptotic stability and feedback stabilization. *Systems.* New York: Interscience Publishers, 1950.
- 36. J. Guckenheimer and P. Holmes, *Nonlinear Oscillations, Dynami-* Publishing House Jerusalem, Ltd., 1975. *cal Systems, and Bifurcations of Vector Fields.* Vol. 42, *Applied Mathematical Sciences.* Berlin: Springer-Verlag, 1983. B. LEHMAN
- 37. J. Baillieul and B. Lehman, Open-loop control using oscillatory S. WEIBEL inputs. In W. S. Levine, ed., *The Control Handbook*, Boca Raton, Northeastern University FL: CRC Press and IEEE Press, 1996, pp. 967–980.<br>38. R. Bellman, J. Bentsman, and S. M. Meerkov, Vibrational control and Decay J. BA
- 8. Bellman, J. Bentsman, and S. M. Meerkov, Vibrational control Boston University of nonlinear systems: Vibrational controllability and transient behavior, *IEEE Trans. Autom. Control,* **AC-31**: 717–724, 1986.
- 39. J. Baillieul, Geometric methods for nonlinear optimal control problems, *JOTA,* **25** (4): 519–548, 1978.
- 
- mer 1976.<br>
1. G. W. Haynes and H. Hermes, Nonlinear controllability via lie **OPERATING SYS** 41. G. W. Haynes and H. Hermes, Nonlinear controllability via lie **OPERATING SYSTEMS, UNIX.** See UNIX. theory, *SIAM J. Control,* **<sup>8</sup>** (4): 450–460, 1970.
- 42. N. E. Leonard and P. S. Krishnaprasad, Motion control of driftfree, left invariant systems on lie groups, *IEEE Trans. Autom. Control,* **40** (9): 1995.
- 43. H. J. Sussman and W. Liu, Lie bracket extension and averaging: The single bracket case. In Z. Li and J. F. Canny (eds.), *Nonholonomic Motion Planning.* 109–147. Boston: Kluwer Academic Publishers, 1993.
- 44. H. J. Sussman and W. Liu, Limits of highly oscillatory controls and approximation of general paths by admissible trajectories. In *Proc. 30th IEEE CDC,* December 1991.
- 45. S. M. Meerkov and M. Tsitkin, The effectiveness of the method of vibrational control for dynamic systems described by a differential equation of order n, *Automation and Remote Control (translation of Avtomatika i Telemekhanika),* 525–529, 1975.
- 46. R. Bellman, J. Bentsman, and S. M. Meerkov, Stability of fast periodic systems, *IEEE Trans. Autom. Control,* **AC-30**: 289–291, 1985.
- 47. J. K. Hale, *Ordinary Differential Equations.* Texts and Monographs in Pure and Applied Mathematics, Malabar, FL: Robert E. Krieger Publishing, 1969.
- 48. J. A. Sanders, On the fundamental theory of averaging, *SIAM J. Math. Anal.,* **14**: 1–10, 1983.
- 49. D. Seto and J. Baillieul, Control problems in superarticulated mechanical systems, *IEEE Trans. Autom. Control,* **AC-39-12**: 2442–2453, 1994.
- 50. M. Fliess, Generalized controller canonical forms for linear and nonlinear dynamics, *IEEE Trans. Autom. Control,* **AC-35-9**: 994– 1001, 1990.
- 51. A. J. Lichtenberg and M. A. Lieberman, *Regular and Chaotic Dynamics.* Volume 38 of *Applied Mathematical Sciences.* Berlin: Springer Verlag, 1992.
- 52. S. Wiggins, *Introduction to Applied Nonlinear Dynamical Systems and Chaos.* In *Texts in Applied Mathematics.* vol. 2, Berlin: Springer Verlag, 1990.
- control to cancel unstable decentralized fixed modes, *IEEE Trans. Systems with Applications to Adiabatic Theorems.* In *Applied Autom. Control,* **AC30**: 95–99, 1985. *Mathematical Sciences.* vol. 72, New York: Springer-Verlag, 1988.
	- *Dynamical Systems, Applied Mathematical Sciences.* vol. 59, Ber-
	-
- 1992. 56. J. J. Stoker, *Nonlinear Vibrations in Mechanical and Electrical*
- ferential Geometric Control Theory. Basel: Birkhaüser, 1983.<br>
36. J. Guckenheimer and P. Holmes, Nonlinear Oscillations, Dynami-<br>
36. J. Guckenheimer and P. Holmes, Nonlinear Oscillations, Dynami-<br>
1975.

40. J. Baillieul, Multilinear optimal control. In *Proc. Conf. Geometry* **OPEN REGION PROBLEMS.** See INTEGRAL EQUATIONS. *Control Eng.: NASA-Ames, Brookline, MA: Math. Sci. Press, Sum-* **OPERATING SYSTEMS, NETWORK.** See NETWORK OP-