

Figure 1. *RLC* circuit.

to neglect it and replace the second-order *RLC* equation (1) by the first-order *RC* equation

$$
RC\overline{v} + \overline{v} = u \tag{2}
$$

in Fig. 2. Neglecting several such ''parasitic'' parameters (small time constants, masses, moments of inertia, etc.) often leads to a significant simplification of a high-order model. To validate such simplifications, we must examine whether, and in what sense, a lower-order model approximates the main phenomenon described by the original high-order model.

To see the issues involved, consider the *RLC* circuit in Eq. (1) with $u = 0$. Its free transients are due to the amount of energy stored in *C* and *L*, that is, the initial conditions $v(0)$ and $\dot{v}(0)$, respectively. The simplified model in Eq. (2) disregards the transient due to $\dot{v}(0)$, that is, the dissipation of energy stored in the inductance *L*. When *L* is small, this transient is fast, and after a short initial time, the *RC* equation . (2) provides an adequate description of the remaining slow transient due to the energy stored in *C.*

The *RLC* circuit in Eq. (1) with a small *L* is a two-timescale system, and the *RC* circuit in Eq. (2) is its slow timescale approximation. In higher-order models, several small parameters may cause a multi-time-scale phenomenon, which can be approximated by ''nested'' two-time-scale models. In this article we consider only the two-time-scale systems.

In this example, a parameter perturbation from $L > 0$ to $L = 0$ has resulted in a model order reduction. Such parameter perturbations are called singular, as opposed to regular perturbations, which do not change the model order. For example, if instead of *L*, the small parameter is *R*, then its per- $\text{turbation from } R > 0 \text{ to } R = 0 \text{ leaves the order of the } RLC$ equation (1) unchanged. The resulting undamped sinusoidal oscillation is due to both $v(0)$ and $\dot{v}(0)$.

In the engineering literature of the past 30 years, singular perturbation techniques and their applications have been discussed in hundreds of papers and a dozen of books. This article presents only the basic singular perturbation tools for reduced-order modeling and systematic approximation of twotime-scale systems. Our main sources are the textbook by **SINGULARLY PERTURBED SYSTEMS** Kokotovic, Khalil, and O'Reilly (1) and the IEEE collection of

Many models of dynamic systems contain small parameters multiplying some of the time derivatives. When such small parameters are neglected, the dynamic order of the model is usually reduced, as illustrated by the series *RLC* circuit

$$
LC\ddot{v} + RC\dot{v} + v = u \tag{1}
$$

in Fig. 1, where *v* is the capacitor voltage and *u* is the applied voltage. If the inductance *L* is very small, then it is common **Figure 2.** *RC* circuit.

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Several examples in the next section will show that a common state-space model of many two-time-scale systems is

$$
\dot{x} = f(x, z, \epsilon, t), \quad x(t_0) = x_0 \tag{3a}
$$

$$
\epsilon \dot{z} = g(x, z, \epsilon, t), \quad z(t_0) = z_0 \tag{3b}
$$

where $x \in R^{n_s}, z \in R^{n_f}$, and $\epsilon > 0$ is the small singular perturbation parameter. The parameter ϵ represents the small time constants and other ''parasitics'' to be neglected in the slow time-scale analysis.

 \dot{z} = $O(1/\epsilon)$, which means that *z* exhibits a fast transient. When this fast transient settles, the longer-term behavior of to infinity even for finite fixed *t* only slightly larger than t_0 . In x and z is determined by the quasi-steady-state equation the τ scale, system (7) is represented by

$$
g(\overline{x}, \overline{z}, 0, t) = 0 \tag{4}
$$

where the bar indicates that this equation is obtained by setting $\epsilon = 0$ in Eq. (3b). This equation will make sense only if it has one or several distinct ("isolated") roots In the fast time-scale τ , the variables *t* and *x* are slowly

$$
\overline{z} = \phi(\overline{x}, t) \tag{5}
$$

for all \bar{x} and \bar{z} of interest. If this crucial requirement is satisfied, for example, when $\det(\partial g/\partial z) \neq 0$, then we say that system (3) is a standard model.

The substitution of Eq. (5) into Eq. (3a) results in the reduced model ω_0 which has equilibrium at $\gamma = 0$. The frozen parameters (x_0, y_0)

$$
\dot{\overline{x}} = f(\overline{x}, \phi(\overline{x}, t), 0, t), \quad \overline{x}(t_0) = x_0 \tag{6}
$$

If Eq. (4) has several distinct roots as shown in Eq. (5), then
each of them leads to a distinct reduced model as shown in
Eq. (6). The singular perturbation analysis determines which
of these models provides an $O(\epsilon)$ ap

When, and in what sense, will $\bar{x}(t)$, $\bar{z}(t)$ obtained from Eqs. (6) and (5) be an approximation of the true solution of system (3)? To answer this question, we examine the variable *z*, which has been excluded from the reduced model in Eq. (6)
by $\overline{z} = \phi(\overline{x}, t)$. In contrast to the original variable z, which
starts at t_0 from a prescribed z_0 , the quasi-steady state \overline{z} is
not free to start initial state *z*₀. Thus, $\bar{z}(t)$ cannot be a uniform approximation $\|y(\tau)\| \le k \|y(0)\|e^{-\alpha \tau}$ (11) of *z*. The best we can expect is that the approximation *z* - $\overline{z}(t) = O(\epsilon)$ will hold on an interval excluding t_0 , that is, for for some positive constants *k* and α . Furthermore, we assume $t \in [t_0, t_1]$ where $t_0 > t_0$. On the other hand, it is reasonable to that $v(0)$ be expect the approximation $x - \overline{x}(t) = O(\epsilon)$ to hold uniformly Expect the approximation $x - x(t) = \overline{x}(t_0)$. If the error $z - \overline{z}(t)$ are these conditions, a fundamental result of singular pertur-
for all $t \in [t_0, t_1]$ because $x(t_0) = \overline{x}(t_0)$. If the error $z - \overline{z}(t)$ is bation th indeed $O(\epsilon)$ over $[t_b, t_f]$, then it must be true that during the bation theory, can
initial ("boundary-layer") interval $[t_0, t_b]$ the variable z aptitude approximations proaches *z*. Let us remember that the speed of *z* can be large since $\dot{z} = g/\epsilon$. In fact, having set $\epsilon = 0$ in Eq. (3b), we have made the transient of *z* instantaneous whenever $g \neq 0$. It is

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the change of variables $y = z - \phi(x, t)$, which shifts the quasisteady state of *^z* to the origin. Then Eq. (3b) becomes **TIME-SCALE PROPERTIES OF THE STANDARD MODEL**

$$
\epsilon \dot{y} = g(x, y + \phi(x, t), \epsilon, t) - \epsilon \frac{\partial \phi}{\partial t} - \epsilon \frac{\partial \phi}{\partial x} f(x, y + \phi(x, t), \epsilon, t),
$$

$$
y(t_0) = z_0 - \phi(x_0, t_0) \quad (7)
$$

Let us note that ϵy may remain finite even when ϵ tends to zero and \dot{v} tends to infinity. We introduce a fast time variable τ by setting

$$
\epsilon \frac{dy}{dt} = \frac{dy}{d\tau}
$$
; hence, $\frac{d\tau}{dt} = \frac{1}{\epsilon}$

The rate of change of *z* in Eq. (3b) is of order $1/\epsilon$, that is, and use $\tau = 0$ as the initial value at $t = t_0$. The fast time variable $\tau = (t - t_0)/\epsilon$ is "stretched": if ϵ tends to zero, τ tends

$$
\frac{dy}{d\tau} = g(x, y + \phi(x, t), \epsilon, t) - \epsilon \frac{\partial \phi}{\partial t} - \epsilon \frac{\partial \phi}{\partial x} f(x, y + \phi(x, t), \epsilon, t),
$$

$$
y(0) = z_0 - \phi(x_0, t_0)
$$
 (8)

varying because $t = t_0 + \epsilon \tau$ and $x = x(t_0 + \epsilon \tau)$. Setting $\epsilon = 0$ freezes these variables at $t = t_0$ and $x = x_0$ and reduces Eq. (8) to the autonomous system

$$
\frac{dy}{d\tau} = g(x_0, y + \phi(x_0, t_0), 0, t_0), \quad y(0) = z_0 - \phi(x_0, t_0) \quad (9)
$$

 t_0) in Eq. (9) depend on the given initial state and initial time.

Tikhonov's Theorem

$$
\frac{dy}{d\tau} = g(x, y + \phi(x, t), 0, t)
$$
\n(10)

$$
||y(\tau)|| \le k||y(0)||e^{-\alpha \tau}
$$
\n(11)

$$
x = \overline{x}(t) + O(\epsilon) \tag{12a}
$$

$$
z = \phi(\overline{x}(t), t) + \hat{y}(t/\epsilon) + O(\epsilon)
$$
 (12b)

$$
z = \phi(\overline{x}(t), t) + O(\epsilon)
$$
 (13)

holds uniformly for $t \in [t_b, t_f]$.
Local exponential stability of the boundary-layer system

$$
\operatorname{Re}\left[\lambda\left\{\frac{\partial g}{\partial z}(x,\phi(x,t),0,t)\right\}\right] \le -c < 0\tag{14}
$$

for all (x, t) in the domain of interest, where λ denotes the eigenvalues and *c* is a positive constant. Alternatively, it can be verified by a Lyapunov analysis if there is a Lyapunov function $W(x, y, t)$ that depends on (x, t) as parameters and where all the partial derivatives are evaluated at *x* and $z =$ satisfies $\varphi_0(x)$.

$$
c_1 \|y\|^2 \le W(x, y, t) \le c_2 \|y\|^2 \tag{15}
$$

$$
\frac{\partial W}{\partial y}g(x, y + \phi(x, t), 0, t) \le -c_3 \|y\|^2 \qquad (16) \qquad g(x, \varphi_0(x)) = 0, \text{ that is, } \phi(x, 0) = \varphi_0(x) \qquad (24)
$$

over the domain of interest, where c_1 to c_3 are positive con- terms, we get stants independent of (x, t) .

Slow Manifold

= 0 forces \bar{x} and \bar{z} to lie in an n_s -dimensional quasi-steady-singular, so that state manifold \overline{M} , explicitly described by Eq. (5). It can be shown that, under the conditions of Tikhonov's Theorem, shown that, under the conditions of Tikhonov's Theorem,
there exists an $\epsilon^* > 0$ such that for all $\epsilon \in (0, \epsilon^*]$, the system in Eq. (3) possesses an integral manifold M_{ϵ} that is invariant: whenever $x(t_0)$, $z(t_0) \in M_e$, then $x(t)$, $z(t) \in M_e$ for all $t \in [t_0$. This recursive process can be repeated to find the higher-
 t_1 . The slow manifold M_e is in the e-neighborhood of the quasi-

steady-state manifo cial case when *f* and *g* in the system in Eq. (3) do not depend on *t* and ϵ :

$$
\dot{x} = f(x, z) \tag{17a}
$$

$$
\epsilon \dot{z} = g(x, z) \tag{17b}
$$

A derivation of slow manifolds for systems with *f* and *g* also dependent on t and ϵ is given in Ref. (26).

 $\phi(x, \epsilon)$. The existence of M_{ϵ} gives a clear geometric meaning that is, the equilibrium manifold of Eq. (28b). The geometry $\phi(x, \epsilon)$ and $x \in R^2$ and $z \in R^1$ is to the slow subsystem of the full-order model in Eq. (17): it is of a third-order system in Eq. (17) with $x \in R^2$ and $z \in R^1$ is
the restriction of the model in Eq. (17) to the slow manifold illustrated in Fig. 3. Start the restriction of the model in Eq. (17) to the slow manifold *M*, given by

$$
\dot{x} = f(x, \phi(x, \epsilon))\tag{18}
$$

To find M_e , we differentiate the manifold $z = \phi(x, \epsilon)$ with respect to *t*

$$
\dot{z} = \frac{d}{dt}\phi(x,\epsilon) = \frac{\partial\phi}{\partial x}\dot{x}
$$
\n(19)

and, upon the multiplication by ϵ and the substitution for \dot{x} and *z˙*, we obtain the *slow manifold condition*

$$
\epsilon \frac{\partial \phi}{\partial x} f(x, \phi(x, \epsilon)) = g(x, \phi(x, \epsilon)) \tag{20}
$$

hold uniformly for $t \in [t_0, t_f]$, where $\hat{y}(\tau)$ is the solution of the which $\phi(x, \epsilon)$ must satisfy for all *x* of interest and all $\epsilon \in (0, 1]$ system in Eq. (9). Moreover, given any $t_b > t_0$, the approxima- ϵ^* . This is a partial differential equation, which, in general, tion is difficult to solve. However, its solution can be approximated by the power series

$$
\phi(x,\epsilon) = \varphi_0(x) + \epsilon \varphi_1(x) + \epsilon^2 \varphi_2(x) + \cdots \tag{21}
$$

where the functions $\varphi_0(x)$, $\varphi_1(x)$, . . ., can be found by equatcan be guaranteed with the eigenvalue condition ing the terms with like powers in ϵ . To this end, we expand *f* and *g* as power series of ϵ

$$
f(x, \varphi_0(x) + \epsilon \varphi_1(x) + \cdots) = f(x, \varphi_0(x)) + \epsilon \frac{\partial f}{\partial z} \varphi_1(x) + \cdots (22)
$$

$$
g(x, \varphi_0(x) + \epsilon \varphi_1(x) + \dots) = g(x, \varphi_0(x)) + \epsilon \frac{\partial g}{\partial z} \varphi_1(x) + \dots
$$
 (23)

We substitute Eqs. (22) and (23) into Eq. (20). The terms with ϵ^0 yield

$$
g(x, \varphi_0(x)) = 0, \text{ that is, } \phi(x, 0) = \varphi_0(x) \tag{24}
$$

which is the quasi-steady-state manifold \overline{M} . Equating the ϵ^1

$$
\frac{\partial g}{\partial z}\varphi_1(x) = \frac{\partial \varphi_0(x)}{\partial x} f(x, \varphi_0(x))
$$
\n(25)

In the state-space $R^{n_s} \times R^{n_f}$ of (x, z) , the equation $g(\bar{x}, \bar{z}, 0, t)$ For the standard model, $\det(\partial g/\partial z) \neq 0$, that is, $\partial g/\partial z$ is non-

$$
\varphi_1(x) = \left(\frac{\partial g}{\partial z}\right)^{-1} \frac{\partial \varphi_0(x)}{\partial x} f(x, \varphi_0(x)) \tag{26}
$$

$$
y = z - \phi(x, \epsilon) \tag{27}
$$

In the x, y coordinates, the system in Eq. (17) becomes

$$
\dot{x} = f(x, \phi(x, \epsilon) + y) \tag{28a}
$$

$$
\epsilon \dot{y} = g(x, \phi(x, \epsilon) + y) - \epsilon \frac{\partial \phi}{\partial x} f(x, \phi(x, \epsilon) + y)
$$
 (28b)

We will seek the graph of M_{ϵ} in the explicit form M_{ϵ} : $z =$ In these coordinates, the slow manifold M_{ϵ} is simply $y = 0$,
i. ϵ) The existence of M gives a clear geometric meaning that is, the equilibrium

Figure 3. Trajectory converging to a slow manifold.

dition $(x(0), z(0))$, the state trajectory rapidly converges to M_c applied to Eq. (32) results in the block-diagonal system and then slowly evolves along *M*.

Linear Two-Time-Scale Systems

The manifold condition in Eq. (20) is readily solvable for linear two-time-scale systems

$$
\dot{x} = A_{11}x + A_{12}z + B_1u, \quad x(t_0) = x_0 \tag{29a}
$$

$$
\epsilon \dot{z} = A_{21} x + A_{22} z + B_2 u, \quad z(t_0) = z_0 \eqno(29b)
$$

where A_{22} is nonsingular, which corresponds to $det(\partial g/\partial z) \neq$ 0, and $u \in \mathbb{R}^m$ is the control input vector. The change of variables

$$
y = z + L(\epsilon)x \tag{30}
$$

where the $n_f \times n_s$ matrix $L(\epsilon)$ satisfies the matrix quadratic equation $H(\epsilon) = A_{12}A_{22}^{-1}$

$$
A_{21}-A_{22}L+\epsilon LA_{11}-\epsilon LA_{12}L=0 \eqno(31)
$$

transforms the system in Eq. (29) into a block-triangular sys-
tem $H_{k+1} = A_{12}A_{22}^{-1}$

$$
\begin{bmatrix} \dot{x} \\ \epsilon \dot{y} \end{bmatrix} = \begin{bmatrix} A_{11} - A_{12}L & A_{12} \\ 0 & A_{22} + \epsilon LA_{12} \end{bmatrix} \begin{bmatrix} x \\ y \end{bmatrix} + \begin{bmatrix} B_1 \\ B_2 - \epsilon LB_1 \end{bmatrix} u
$$
\n(32)

$$
\begin{bmatrix} x(t_0) \\ y(t_0) \end{bmatrix} = \begin{bmatrix} x_0 \\ z_0 + Lx_0 \end{bmatrix}
$$
 (33) $\dot{\xi} = A_0 \xi + B_0 u, \quad \xi(t_0) = x_0$ (43)
where $A = A_0 = A_0 A^{-1} A$ and $B = B_0 = A_0 A^{-1} B$. The fact

Note that Eq. (31) is the slow manifold condition in Eq. (20) subsystem is approximated to $O(\epsilon)$ by for linear systems.

Given that A_{22} is nonsingular, the implicit function theorem implies that Eq. (31) admits a solution $L(\epsilon)$ for ϵ sufficiently small. Furthermore, an asymptotic expansion of the Thus as $\epsilon \to 0$, the slow eigenvalues of Eq. (29) are approxi-

$$
L(\epsilon) = A_{22}^{-1}A_{21} + \epsilon A_{22}^{-2}A_{21}(A_{11} - A_{12}A_{22}^{-1}A_{21}) + O(\epsilon^2)
$$
 (34)

$$
A_{21} - A_{22}L(0) = 0 \tag{35}
$$

whose solution is the first term in Eq. (34). Furthermore, to solve for the higher-order terms, an iterative scheme **EXAMPLES**

$$
L_{k+1} = A_{22}^{-1}A_{21} + \epsilon A_{22}^{-1}L_k(A_{11} - A_{12}L_k), \quad L_0 = A_{22}^{-1}A_{21} \quad (36)
$$

Eq. (32) is now decoupled from the slow variable *x*, the slow variable x is still driven by the fast variable y . To remove this influence, the change of variables

$$
\xi = x - \epsilon H(\epsilon) y \tag{37}
$$

where the $n_s \times n_f$ matrix $H(\epsilon)$ satisfies the linear matrix equation which yields the reduced-order model

$$
\epsilon (A_{11} - A_{12}L)H - H(A_{22} + \epsilon LA_{12}) + A_{12} = 0 \qquad (38)
$$

$$
RC\dot{\overline{x}} = -\overline{x} + u \qquad (46)
$$

$$
\begin{bmatrix} \dot{\xi} \\ \epsilon \dot{y} \end{bmatrix} = \begin{bmatrix} A_{11} - A_{12}L & 0 \\ 0 & A_{22} + \epsilon LA_{12} \end{bmatrix} \begin{bmatrix} \xi \\ y \end{bmatrix}
$$

$$
+ \begin{bmatrix} B_1 - H(B_2 + \epsilon LB_1) \\ B_2 - \epsilon LB_1 \end{bmatrix} u
$$
(39)

with the initial condition

$$
\begin{bmatrix} \xi(t_0) \\ y(t_0) \end{bmatrix} = \begin{bmatrix} x_0 - \epsilon H(z_0 + Lx_0) \\ z_0 + Lx_0 \end{bmatrix}
$$
 (40)

For ϵ sufficiently small, Eq. (38) admits a unique solution $H(\epsilon)$ that can be expressed as

$$
H(\epsilon) = A_{12}A_{22}^{-1} + O(\epsilon)
$$
 (41)

The solution $H(\epsilon)$ can also be computed iteratively as

$$
H_{k+1} = A_{12}A_{22}^{-1} + \epsilon((A_{11} - A_{12}L)H_k + H_kLA_{12})A_{22}^{-1},
$$

$$
H_0 = A_{12}A_{22}^{-1} \quad (42)
$$

If *L* is available from the recursive formula (36), we can use L_{k+1} instead of L in Eq. (42).

From the block-diagonal form in Eq. (39), it is clear that with the initial condition the slow subsystem of Eq. (29) is approximated to $O(\epsilon)$ by

$$
\dot{\xi} = A_0 \xi + B_0 u, \quad \xi(t_0) = x_0 \tag{43}
$$

where $A_0 = A_{11} - A_{12}A_{22}^{-1}A_{21}$ and $B_0 = B_1 - A_{12}A_{22}^{-1}B_2$. The fast

$$
\epsilon \dot{y} = A_{22}y + B_2 u, \quad y(t_0) = z_0 - A_{22}^{-1} A_{21} x_0 \tag{44}
$$

solution to Eq. (31) is given by mated by $\lambda(A_0)$, and the fast eigenvalues are approximated by $\lambda(A_{22})/\epsilon.$ It follows that if ${\rm Re}\{\lambda(A_0)\} < 0$ and ${\rm Re}\{\lambda(A_{22})\} < 0,$ $L(\epsilon) = A_{22}^{-1}A_{21} + \epsilon A_{22}^{-2}A_{21}(A_{11} - A_{12}A_{22}^{-1}A_{21}) + O(\epsilon^2)$ (34) then there exists an $\epsilon^* > 0$ such that (29) is asymptotically stable for all $\epsilon \in (0, \epsilon^*]$. Furthermore, if the pair (A_0, B_0) and We can readily verify that for $\epsilon = 0$, Eq. (31) reduces to the pair (A_{22}, B_2) are each completely controllable (stabiliza- $A_{21} - A_{22}L(0) = 0$ (35) ble), then there exists an $\epsilon^* > 0$ such that Eq. (29) is com-
pletely controllable (stabilizable) for all $\epsilon \in (0, \epsilon^*]$.

$Example 1.$ An RLC Circuit

To complete our introductory example, we represent the *RLC* circuit in Eq. (1) using the state variables $x = v$ and $z = \dot{v}$:
Although the fast variable *y* in the triangular system in

$$
\dot{x} = z \tag{45a}
$$

$$
\epsilon \dot{z} = -z - \frac{1}{RC}(x - u) \tag{45b}
$$

where ϵ is the small time constant L/R . The unique solution $\mathbf{z}_s \times n_f$ matrix $H(\epsilon)$ satisfies the linear matrix equa- of the quasi-steady-state equation (5) is $\bar{z} = -(\bar{x} - u)/(RC)$,

$$
RC\overline{x} = -\overline{x} + u\tag{46}
$$

As expected, this is the *RC* equation (2). The boundary-layer system Eq. (9) for $y = z + (x - u)/(RC)$ is

$$
\frac{dy}{d\tau} = -y, \quad y(t_0) = z(t_0) + \frac{1}{RC}(x(t_0) - u(t_0))
$$
(47)

Its solution $y = e^{-\tau}y(t_0) = -e^{-(R/L)t}y(t_0)$ approximates the fast transient neglected in the slow subsystem in Eq. (46). Tikhonov's Theorem is satisfied, because the fast subsystem in Eq. (47) is exponentially stable.

Example 2. A dc Motor

A common model for dc motors, shown in Fig. 4, under constant field excitation, consists of a mechanical torque equation and an equation for the electrical transient in the armature circuit, namely,

$$
J\dot{\omega} = Ki - T_{\rm L} \tag{48a}
$$

$$
Li = -K\omega - Ri + u \tag{48b}
$$

 t ance, and *inductance*, respectively, J is the combined moment of inertia of the motor and the load, ω is the angular state condition, $T_L = 14.3$ N m, $i = 3.38$ A, and $u = 246.8$ V.
speed, T_i is the load torque, and K is a motor design constant. The time constants are $\tau_e =$ speed, T_L is the load torque, and K is a motor design constant The time constants are $\tau_e = 0.0073$ s and $\tau_m = 0.115$ s, re-
such that Ki and K₀ are, respectively, the motor torque and sulting in $\epsilon = 0.063$. The res such that Ki and $K\omega$ are, respectively, the motor torque and

We consider the case in which the electrical time constant $\tau_e = L/R$ is much smaller than the mechanical time constant Note that \bar{x} is a good approximation of *x*. Initially there is a $\tau_m = JR/R^2(3)$. Defining $\epsilon = \tau_e/\tau_m$, $x = \omega$, and $z = i$, we rewrite fast transient in *z*. Af $\tau_m = J R / K^2$ (3). Defining $\epsilon = \tau_e / \tau_m$, $x = \omega$, and $z = i$, we rewrite Eq. (48) as \overline{z} becomes a good approximation of *z*.

$$
\dot{x} = \frac{R}{\tau_{\rm m} K} z - \frac{1}{J} T_{\rm L} \tag{49a}
$$

$$
\epsilon \dot{z} = -\frac{K}{\tau_{\rm m} R} x - \frac{1}{\tau_{\rm m}} z + \frac{1}{\tau_m R} u \tag{49b}
$$

Setting $\epsilon = 0$, we obtain from Eq. (49b) which has the unique solution

$$
0 = -K\overline{x} - R\overline{z} + u \tag{50}
$$

Thus the quasi-steady state of *z* is

$$
\overline{z} = \frac{u - K\overline{x}}{R} \tag{51}
$$

which, when substituted in Eq. (49a), yields the slow mechan-

ical subsystem *x*ⁱ = $x^2(1 + t)/z$ (55a)

$$
\overline{m}\dot{\overline{x}} = -\overline{x} + \frac{1}{K}u - \frac{\tau_m}{J}T_L
$$
\n(52)

Figure 5. dc motor step response.

The parameters of a 1 hp (746 W) dc motor with a rated where *i*, *u*, *R*, and *L* are the armature current, voltage, resis-
tance. and inductance, respectively. *J* is the combined mo-
 $K = 4.23$ Vs/rad, and $J = 0.136$ kg m². At the rated steady the back emf (electromotive force). of the full model in Eq. (48) (solid curves) and the slow sub-
We consider the case in which the electrical time constant system in Eqs. (51) and (52) (dashed curves) is shown Fig. 5.

> The fast electrical transient is approximated by the boundary-layer system

$$
\tau_{\rm m} \frac{dy}{d\tau} = -y, \quad y(0) = z(0) - \frac{u(0) - Kx(0)}{R} \tag{53}
$$

$$
0 = -K\overline{x} - R\overline{z} + u \qquad (50) \qquad \qquad y = e^{-\tau/\tau_{\text{m}}}y(0) = e^{-t/\tau_{\text{e}}}y(0) \qquad (54)
$$

Example 3. Multiple Slow Subsystems

To illustrate the possibility of several reduced-order models, we consider the singularly perturbed system

$$
\dot{x} = x^2(1+t)/z \tag{55a}
$$

$$
\epsilon \dot{z} = -[z + (1+t)x]z[z - (1+t)] \tag{55b}
$$

 $\tau_m \overline{x} = -\overline{x} + \frac{\overline{x}}{K}u - \frac{\overline{x}}{J}T_L$ (52) where the initial conditions are $x(0) = 1$ and $z(0) = z_0$. Setting $\epsilon = 0$ results in

$$
0 = -[\overline{z} + (1+t)\overline{x}]\overline{z}[\overline{z} - (1+t)]\tag{56}
$$

which has three distinct roots

$$
\overline{z} = -(1+t)\overline{x}; \quad \overline{z} = 0; \quad \overline{z} = 1+t \tag{57}
$$

Consider first the root $z = -(1 + t)\overline{x}$. The boundary-layer system in Eq. (10) is

$$
\frac{dy}{dt} = -y[y - (1+t)x][y - (1+t)x - (1+t)]
$$
 (58)

Taking $W(y) = y^2/2$, it can be verified that *W* satisfies the inequalities in Eqs. (15) and (16) for $y < (1 + t)x$. The reduced system

$$
\dot{\overline{x}} = -\overline{x}, \quad \overline{x}(0) = 1 \tag{59}
$$

has the unique solution $\bar{x}(t) = e^{-t}$ for all $t \ge 0$. The boundarylayer system with $t = 0$ and $x = 1$ is

$$
\frac{d\hat{y}}{d\tau} = -\hat{y}(\hat{y} - 1)(\hat{y} - 2), \quad \hat{y}(0) = z_0 + 1 \tag{60}
$$

and has a unique exponentially decaying solution $\hat{y}(\tau)$ for z_0 + $1 \leq 1 - a$, that is, for $z_0 \leq -a < 0$ where $a > 0$ can be arbitrarily small.

Consider next the root $\bar{z} = 0$. The boundary-layer system in Eq. (10) is

$$
\frac{dy}{d\tau} = -[y + (1+t)x]y[y - (1+t)]\tag{61}
$$

By sketching the right-hand side function, it can be seen that full model (solid curves) and two for each reduced model
the origin is unstable. Hence, Tikhonov's Theorem does not (dashed curves).
apply to this case and we

Furthermore, x is not defined at $z = 0$.
Finally, the boundary-layer system for the root $\overline{z} = 1 + t$
Is and m_u are the car body and tire masses, k_s and k_t are the

$$
\frac{dy}{d\tau} = -[y + (1+t) + (1+t)x][y + (1+t)]y \tag{62}
$$

nentially stable uniformly in (x, t) . The reduced system

$$
\dot{\overline{x}} = \overline{x}^2, \ \overline{x}(0) = 1 \tag{63}
$$

has the unique solution $\bar{x}(t) = 1/(1 - t)$ for all $t \in [0, 1)$. Notice that $\bar{x}(t)$ has a finite escape time at $t = 1$. However, Tikhonov's Theorem still holds for $t \in [0, t_f]$ with $t_f < 1$. The *In a typical car*, the natural frequency $\sqrt{k_t/m_u}$ of the tire boundary-layer system, with $t = 0$ and $x = 1$, is much higher than the natural frequency $\sqrt{k_s/m_s}$ of the car

$$
\frac{d\hat{y}}{d\tau} = -(\hat{y} + 2)(\hat{y} + 1)\hat{y}, \quad \hat{y}(0) = z_0 - 1 \tag{64}
$$

has a unique exponentially decaying solution $\hat{y}(\tau)$ for $z_0 >$ $a > 0$.

rise to valid reduced models. Tikhonov's Theorem applies to the root $\phi = -(1 + t)x$ if $z_0 < 0$ and to the root $\phi = 1 + t$ if

Figure 6. Response of system in Eq. (55) illustrating two slow subsystems.

Figure 7. Quarter-car model. *dy*

spring constants of the strut and the tire, and b_s is the damper constant of the shock. The distances d_s , d_u , and d_r are the elevations of the car, the tire, and the road surface, re-As in the first case, it can be shown that the origin is expo-
neutively. From Newton's Law, the balance of forces acting
neutially stable uniformly in (x, t) . The reduced system
on m_s and m_u results in the modeling e

$$
m_{\rm s}d_{\rm s} + b_{\rm s}(d_{\rm s} - d_{\rm u}) + k_{\rm s}(d_{\rm s} - d_{\rm u}) = 0 \qquad (65a)
$$

$$
m_{\rm u}\ddot{d}_{\rm u} + b_{\rm s}(\dot{d}_{\rm u} - \dot{d}_{\rm s}) + k_{\rm s}(d_{\rm u} - d_{\rm s}) + k_{\rm t}(d_{\rm u} - d_{\rm r}) = 0 \qquad (65b)
$$

body and the strut. We therefore define the parameter

$$
\epsilon = \sqrt{\frac{k_s/m_s}{k_t/m_u}} = \sqrt{\frac{k_s m_u}{k_t m_s}}\tag{66}
$$

In summary, only two of the three roots in Eq. (57) give The mass-spring system in Eq. (65) is of interest because it
e to valid reduced models Tikbonov's Theorem annies to cannot be transformed into a standard model with dependent scaling. From Eq. (66), the tire stiffness $k_t = O(1/\epsilon^2)$ tends to infinity as $\epsilon \to 0$. For the tire potential en $z_0 > 0$. Figure 6 shows *z* for four different values of z_0 of the $O(1/\epsilon^2)$ tends to infinity as $\epsilon \to 0$. For the tire potential energy $k_{\rm t} (d_{\rm u} - d_{\rm r})^2/2$ to remain bounded, the displacement $d_{\rm u}$ – $d_{\rm r}$ must be $O(\epsilon)$, that is, the scaled displacement $(d_{\rm u}-d_{\rm r})/\epsilon$ must remain finite. Thus to express Eq. (65) in the standard singularly perturbed form, we introduce the slow and fast variables as

$$
x = \begin{bmatrix} d_s - d_u \\ \dot{d}_s \end{bmatrix}, \quad z = \begin{bmatrix} (d_u - d_r)/\epsilon \\ \dot{d}_u \end{bmatrix}
$$
(67)

and $u = d_r$ as the disturbance input. The resulting model is

$$
\dot{x} = A_{11}x + A_{12}z + B_1u \tag{68a}
$$

$$
\epsilon \dot{z} = A_{21}x + A_{22}z + B_2u \tag{68b}
$$

$$
A_{11} = \begin{bmatrix} 0 & 1 \\ -k_s/m_s & -b_s/m_s \end{bmatrix}, \quad A_{12} = \begin{bmatrix} 0 & -1 \\ 0 & b_s/m_s \end{bmatrix}, \quad B_1 = \begin{bmatrix} 0 \\ 0 \end{bmatrix}
$$

$$
A_{21} = \begin{bmatrix} 0 & 0 \\ \alpha k_s/m_s & \alpha b_s/m_s \end{bmatrix}, \quad A_{22} = \begin{bmatrix} 0 & 1 \\ -k_s/m_s & -\alpha b_s/m_s \end{bmatrix},
$$

$$
B_2 = \begin{bmatrix} -1 \\ 0 \end{bmatrix}
$$
(69)

$$
\alpha = \sqrt{\frac{k_{\rm s} m_{\rm s}}{k_{\rm t} m_{\rm u}}} \tag{70}
$$

The parameters of a typical passenger car are $m_s = 504.5$ kg, $m_u = 62$ kg, $b_s = 1,328$ Ns/m, $k_s = 13,100$ N/m, $k_t =$ 252,000 N/m, and $\alpha = 0.65$. In this case, $\epsilon = 0.08$ and the
time scales of the body (slow) and the tire (fast) are well sepa-
rated.
Recomes

To illustrate the approximation provided by the two-timescale analysis, the slow and fast eigenvalues of the uncorrected subsystems in Eqs. (43) and (44) are found to be $-2.632 \pm j6.709$ and $-2.632 \pm j6.709$ and $-10.710 \pm j62.848$, respectively, which which yields, as $\epsilon \to 0$, the quasi-steady state are within 4% of the eigenvalues $-2.734 \pm j7.018$ and -9.292 \pm *j*60.287 of the full-order model. If a higher accuracy is desired, the series expansion in Eq. (34) can be used to add the first-order correction terms to the diagonal blocks of Eq. (39), resulting in the approximate slow eigenvalues $-2.705 \pm$ *j*6.982 and fast eigenvalues $-9.394 \pm j60.434$, and thus re-

Example 5. A High-Gain Power Rectifier

fiers as actuators. An example is a static excitation system We now proceed to illustrate the use of the slow manifold con-
that controls the field voltage E_{fit} of a synchronous machine cept as a modeling tool. In mo that controls the field voltage E_{fd} of a synchronous machine shown in Fig. 8 (4). The synchronous machine is modeled as manifold condition in Eq. (20) is evaluated approximately as

$$
\dot{x} = f(x, E_{\text{fd}}) \tag{71a}
$$

$$
V_{\rm T} = h(x) \tag{71b}
$$

where x is the machine state vector including the flux vari-
The synchronous machine model with one damper winding ables and the scalar output V_T is the generator terminal volt- in the quadrature axis is age. Here we focus on the exciter system which, from Fig. 8, is described by

$$
T_{\rm M}\dot{E} = -E - K_{\rm M}K_{\rm G}E_{\rm fd} + K_{\rm M}V_{\rm R}
$$
 (72a)

$$
E_{\rm fd} = V_{\rm B}(x)E\tag{72b}
$$

where T_M is a time constant and K_M and K_G are gains.

where \blacksquare Following the input signal V_{R} , the voltage E modulates the supply voltage $V_B(x)$ to control E_{fd} . The supply voltage $V_B(x)$ is a function of *x* and is typically within the range of 4–8 per unit on the base field voltage.

> Under normal operating conditions, the product of the rectifier gain and the supply voltage $K_M V_B(x)$ is very high. This motivates us to define ϵ as

(69)
$$
\epsilon = \frac{1}{K_{\rm M} \overline{V}_{\rm B}} \ll 1
$$
 (73)

and where \overline{V}_B is a constant of the order of magnitude of $V_B(x)$, that is, the ratio $\beta(x) = V_{\text{B}}(x)/\overline{V}_{\text{B}} = O(1)$. Using E_{fd} as the state variable instead of E , Eq. (72) is rewritten as

$$
T_{\rm M} \dot{E}_{\rm fd} = -\left[1 + K_{\rm G} K_{\rm M} V_{\rm B}(x) - \frac{T_{\rm M}}{V_{\rm B}(x)} \gamma(x, E_{\rm fd})\right] E_{\rm fd} + K_{\rm M} V_{\rm B}(x) V_{\rm R}
$$
(74)

$$
\epsilon \dot{E}_{\rm fd} = -\left[\frac{K_{\rm G}\beta(x) + \epsilon}{T_{\rm M}} - \frac{\epsilon}{V_{\rm B}(x)}\gamma(x, E_{\rm fd})\right] E_{\rm fd} + \frac{\beta(x)}{T_{\rm M}}V_{\rm R} \quad (75)
$$

$$
\overline{E}_{\text{fd}} = \frac{1}{K_{\text{G}}} V_{\text{R}} \tag{76}
$$

In a typical rectifier system, $K_G = 1, K_M = 7.93$, and $T_M =$ 0.4 s. For $V_B = 6$, we obtain $\epsilon = 0.021$. The time-scales are ducing the errors to less than 0.5%. well separated, which allows us to achieve high accuracy with singular perturbation approximations.

Many modern control systems include power electronic recti- *Example 6. Slow Manifold in a Synchronous Machine*

a power series in ϵ . However, for a synchronous machine model, the slow manifold, which excludes stator circuit transients, can be calculated exactly, as shown by Kokotovic and Sauer (5).

$$
\frac{d\delta}{dt} = \omega - \omega_{\rm s} \tag{77a}
$$

$$
\frac{2H}{\omega_{\rm s}}\frac{d\omega}{dt} = T_{\rm m} + \left(\frac{1}{L'_{\rm q}} - \frac{1}{L'_{\rm d}}\right)\psi_{\rm d}\psi_{\rm q} + \frac{1}{L'_{\rm q}}\psi_{\rm d}E'_{\rm d} + \frac{1}{L'_{\rm d}}\psi_{\rm q}E'_{\rm q} \tag{77b}
$$

$$
T'_{\rm do} \frac{dE'_{\rm q}}{dt} = -\frac{L_{\rm d}}{L'_{\rm d}} E'_{\rm q} - \frac{L_{\rm d} - L'_{\rm d}}{L'_{\rm d}} \psi_{\rm d} + E_{\rm fd} \tag{77c}
$$

$$
T'_{\rm qo} \frac{dE'_{\rm d}}{dt} = -\frac{L_{\rm q}}{L'_{\rm q}} E'_{\rm d} - \frac{L_{\rm q} - L'_{\rm q}}{L'_{\rm q}} \psi_{\rm q} \tag{77d}
$$

$$
\frac{1}{\omega_{\rm s}} \frac{d\psi_{\rm d}}{dt} = -\frac{R_{\rm a}}{L_{\rm d}'} \psi_{\rm d} + \frac{R_{\rm a}}{L_{\rm d}'} E_{\rm q}' + \frac{\omega}{\omega_{\rm s}} \psi_{\rm q} + V \sin \delta \tag{77e}
$$

$$
\frac{1}{\sqrt{1+sT_{\text{M}}}} \xrightarrow{\text{TM}} \frac{1}{\sqrt{1+sT_{\text{M}}}} \times \frac{1}{\sqrt{1+sT_{\text{M}}}} \xrightarrow{\text{Synchronous}} \frac{1}{\sqrt{1+s}} \frac{d\psi_{\text{q}}}{dt} = -\frac{R_{\text{a}}}{L_{\text{q}}'} \psi_{\text{q}} - \frac{R_{\text{a}}}{L_{\text{q}}'} \psi_{\text{d}} - \frac{\omega}{\omega_{\text{s}}} \psi_{\text{d}} + V \cos \delta \tag{77f}
$$

where δ , ω , and *H* are the generator rotor angle, speed, and **Figure 8.** Static excitation system. inertia, respectively, (E'_d, E'_q) , (ψ_d, ψ_q) , (T'_{d0}, T'_{q0}) , (L_d, L_q) , and (L_d, L_u) are the *d*- and *q*-axis voltages, flux linkages, opencircuit time constants, synchronous reactances, and transient reactances, respectively, R_a is the stator resistance, T_m is the input mechanical torque, E_{fd} is the excitation voltage, and ω_{s} is the system frequency.

In the model shown in Eq. (77), the slow variables are δ , ω , E_d' , and E_s' , and the fast variables are ψ_d and ψ_d . The singular perturbation parameter can be defined as $\epsilon = 1/\omega_s$.

If the stator resistance is neglected, that is, $R_a = 0$, it can be readily verified that the slow manifold condition in Eq. (20) gives the exact slow invariant manifold

$$
\psi_{\rm d} = V \cos \delta, \quad \psi_{\rm q} = -V \sin \delta \tag{78}
$$

Figure 9. Phase portrait of the Van der Pol oscillator.
These expressions can be substituted into Eqs. (77a)–(77d) to obtain a fourth-order slow subsystem.

If the initial condition $[\psi_a(0), \psi_a(0)]$ is not on the manifold shown in Eq. (78), then using the fast variables than boundary) layers, from *B* to *D'* and from *C* to *A'* is clear

$$
y_d = \psi_d - V \cos \delta, \quad y_q = \psi_q + V \sin \delta \tag{79}
$$

$$
\epsilon \frac{dy_{\rm d}}{dt} = \frac{\omega}{\omega_{\rm s}} y_{\rm q}, \qquad y_{\rm d}(0) = \psi_{\rm d}(0) - V \cos \delta(0) \tag{80a}
$$

$$
\epsilon \frac{dy_{\mathbf{q}}}{dt} = -\frac{\omega}{\omega_{\mathbf{s}}} y_{\mathbf{d}}, \quad y_{\mathbf{q}}(0) = \psi_{\mathbf{q}}(0) + V \sin \delta(0) \tag{80b}
$$

where the state ω appears as a time-varying coefficient. **STABILITY ANALYSIS**

When the stator resistance R_a is nonzero, the slow mani-
fold condition can no longer be solved exactly. Instead, the We consider the autonomous singularly perturbed system in
leading terms in the power series expansion manifold. \blacksquare domain that contains the origin. We want to analyze stability

$$
\dot{x} = z \tag{81a}
$$

$$
\epsilon \dot{z} = -x + z - \frac{1}{3}z^3 \tag{81b}
$$

For ϵ small, the slow manifold is approximated by

$$
g(x, z) = -x + z - \frac{1}{3}z^3 = 0
$$
 (82)

$$
\frac{\partial g}{\partial z} = 1 - z^2 < 0 \tag{83}
$$

fied because $z^2 > 1$. Therefore, the branches AB and CD are attractive, that is, trajectories converging to these two ary-layer systems, we form a composite Lyapunov function branches will remain on them, moving toward either the candidate for the full system as a linear combination of the point *B* or *C*. However, the root on the branch *BC* is unstable Lyapunov functions for the reduced and boundary-layer sysbecause z^2 < 1; hence, this branch of the slow manifold is re- tems. We then proceed to calculate the derivative of the compulsive. posite Lyapunov function along the trajectories of the full sys-

and *CD*, because $\epsilon \neq 0$. The mechanism of two interior (rather *g*, that it is negative definite for sufficiently small ϵ .

from this phase portrait. The relaxation oscillation forming *the limit cycle <i>A'*-*B-D'*-*C* consists of the slow motions from A' to B and D' to C , connected with fast jumps (layers) from we obtain the fast subsystem *B* to *D'* and from *C* to *A'*. When ϵ is increased to a small positive value, this limit cycle is somewhat deformed, but its main character is preserved. This observation is one of the cornerstones of the classical nonlinear oscillation theory (7), which has many applications in engineering and biology.

of the origin by examining the reduced and boundary-layer *Example 7. Van der Pol Oscillator* models. Let $z = \phi(x)$ be an isolated root of $0 = g(x, z)$ defined A classical use of the slow manifold concept is to demonstrate in a domain $D_1 \in \mathbb{R}^n$ that contains $x = 0$, such that $\phi(x)$ is A classical use of the slow manifold concept is to demonstrate in a domain $D_1 \in \mathbb{R}^n$ that contains $x = 0$, such that $\phi(x)$ is the relaxation oscillation phenomenon in a Van der Pol oscil-
continuous and $\phi(0) = 0$. lator, modeled in the state-space form as (6) $z - \phi(x)$, the singularly perturbed system is represented in the new coordinates as

$$
\frac{1}{3}z^3 \t\t (81b) \t\t \dot{x} = f(x, y + \phi(x)) \t\t (84a)
$$

$$
\epsilon \dot{y} = g(x, y + \phi(x)) - \epsilon \frac{\partial \phi}{\partial x} f(x, y + \phi(x)) \tag{84b}
$$

The reduced system $\dot{x} = f(x, \phi(x))$ has equilibrium at $x = 0$, and the boundary-layer system $dy/d\tau = g(x, y + \phi(x))$ has which is shown as the curve *ABCD* in Fig. 9. For the roots equilibrium at $y = 0$. The main theme of the two-time-scale $z = \phi(x)$ on the branches *AB* and *CD* stability analysis is to assume that, for each of the two systems, the origin is asymptotically stable and that we have a Lyapunov function that satisfies the conditions of Lyapunov's Theorem. In the case of the boundary-layer system, we re quire asymptotic stability of the origin to hold uniformly in and the eigenvalue condition of Tikhonov's Theorem is satis- the frozen parameter *x*. Viewing the full singularly perturbed system (84) as an interconnection of the reduced and bound-Figure 9 shows vertical trajectories converging toward *AB* tem and verify, under reasonable growth conditions on *f* and

$$
\frac{\partial V}{\partial x} f(x, \phi(x)) \le -\alpha_1 \psi_1^2(x), \quad \forall \, x \in D_1 \tag{85}
$$

Lyapunov function for the boundary-layer system such that

$$
\frac{\partial W}{\partial y}g(x, y + \phi(x)) \le -\alpha_2 \psi_2^2(y), \quad \forall (x, y) \in D_1 \times D_2 \qquad (86)
$$

where $D_2 \subset R^m$ is a domain that contains $y = 0$, and $\psi_2(y)$ is a positive definite function. We allow the Lyapunov function *W* It follows that the origin is asymptotically stable for all $\epsilon < \epsilon^*$.
It follows that the functions $\psi_1(x)$ and $\psi_2(x)$ and the interconnection to depend on x because x is a parameter of the system and
Lyapunov functions may, in general, depend on the system's
Lyapunov functions may, in general, depend on the system's
conditions in Eqs. (90) and (91) should be ca must keep track of the effect of the dependence of *W* on *x*. To As an illustration of the preceding α and α and α *x*. To α *x*. The second-order system ensure that the origin of the boundary-layer system is asymptotically stable uniformly in *x*, we assume that $W(x, y)$ satis-
fies $\dot{x} = f(x, z) = x - x^3 + z$ (95a)

$$
W_1(y)\leq W(x,y)\leq W_2(y),\quad \forall (x,y)\in D_1\times D_2\qquad \quad \ (87)
$$

for some positive definite continuous functions W_1 and W_2 . Now consider the composite Lyapunov function candidate

$$
\nu(x, y) = (1 - d)V(x) + dW(x, y), \quad 0 < d < 1 \tag{88}
$$

we obtain

$$
\dot{v} = (1 - d) \frac{\partial V}{\partial x} f(x, \phi(x)) + \frac{d}{\epsilon} \frac{\partial W}{\partial y} g(x, y + \phi(x)) \n+ (1 - d) \frac{\partial V}{\partial x} [f(x, y + \phi(x)) - f(x, \phi(x))] \n+ d \left(\frac{\partial W}{\partial x} - \frac{\partial W}{\partial y} \frac{\partial \phi}{\partial x} \right) f(x, y + \phi(x))
$$
\n(89)

We have represented the derivative ν as the sum of four
terms. The first two terms are the derivatives of V and W bounded, the origin is globally asymptotically stable for ϵ
along the trajectories of the reduced

$$
\frac{\partial V}{\partial x}[f(x, y + \phi(x)) - f(x, \phi(x))] \le \beta_1 \psi_1(x)\psi_2(y) \qquad (90)
$$

$$
\left(\frac{\partial W}{\partial x}-\frac{\partial W}{\partial y}\frac{\partial \phi}{\partial x}\right)f(x,y+\phi(x))\leq \beta_2\psi_1(x)\psi_2(y)+\gamma\psi_2^2(y)\eqno(91)
$$

for some nonnegative constants β_1 , β_2 , and γ . Using the in-
for a given t_f . equalities in Eqs. (85), (86), (90), and (91), we obtain **COMPOSITE FEEDBACK CONTROL**

$$
\dot{\nu} \leq -\begin{bmatrix} \psi_1(x) \\ \psi_2(y) \end{bmatrix}^T \begin{bmatrix} (1-d)\alpha_1 & -\frac{1}{2}(1-d)\beta_1 - \frac{1}{2}d\beta_2 \\ -\frac{1}{2}(1-d)\beta_1 - \frac{1}{2}d\beta_2 & d\left(\frac{\alpha_2}{\epsilon} - \gamma\right) \end{bmatrix}
$$

$$
\begin{bmatrix} \psi_1(x) \\ \psi_2(y) \end{bmatrix}
$$
(92)

Let $V(x)$ be a Lyapunov function for the reduced system The right-hand side of the last inequality is a quadratic form such that in $(\psi_1(x), (\psi_2(y)))$. The quadratic form is negative definite when

$$
\frac{\partial V}{\partial x} f(x, \phi(x)) \le -\alpha_1 \psi_1^2(x), \quad \forall x \in D_1 \tag{85} \qquad \qquad d(1-d)\alpha_1 \left(\frac{\alpha_2}{\epsilon} - \gamma\right) > \frac{1}{4} [(1-d)\beta_1 + d\beta_2]^2 \tag{93}
$$

where $\psi_1(x)$ is a positive definite function. Let $W(x, y)$ be a For any *d*, there is an ϵ_d such that Eq. (93) is satisfied for $_{1}/(\beta_{1} + \beta_{2})$ and is given by

$$
\epsilon^* = \frac{\alpha_1 \alpha_2}{\alpha_1 \gamma + \beta_1 \beta_2} \tag{94}
$$

$$
\dot{x} = f(x, z) = x - x^3 + z \tag{95a}
$$

$$
\epsilon \dot{z} = g(x, z) = -x - z \tag{95b}
$$

which has a unique equilibrium point at the origin. Let $y =$ $-\phi(x) = z + x$. For the reduced system $\dot{x} = -x^3$, we take $\frac{1}{4}x^4$, which satisfies Eq. (85) with $\psi_1(x) = |x|^3$ and $\alpha_1 =$ 1. For the boundary-layer system $dy/d\tau = -y$, we take $v(x, y) = (1 - d)V(x) + dW(x, y), \quad 0 < d < 1$ (88) $\overline{W(y)} = \frac{1}{2}y^2$, which satisfies Eq. (86) with $\psi_2(y) = |y|$ and $\alpha_2 =$ where the constant *d* is to be chosen. Calculating the deriva-
tive of ν along the trajectories of the full system in Eq. (84),
 α have

$$
\frac{\partial V}{\partial x}[f(x, y + \phi(x)) - f(x, \phi(x))] = x^3 y \le \psi_1 \psi_2 \tag{96}
$$

$$
\frac{\partial W}{\partial y} f(x, y + \phi(x)) = y(-x^3 + y) \le \psi_1 \psi_2 + \psi_2^2 \tag{97}
$$

Note that $\partial W/\partial x = 0$. Hence, Eqs. (90) and (91) are satisfied with $\beta_1 = \beta_2 = \gamma = 1$. Therefore, the origin is asymptotically stable for $\epsilon < \epsilon^* = 0.5$. Because all the conditions are satisfied globally and $v(x, y) = (1 - d)V(x) + dW(y)$ is radially un-

tem in Eq. (3) is exponentially stable for sufficiently small ϵ . The same conditions ensure the validity of Tikhonov's Theo rem for all $t \geq t_0$. Note that the earlier statement of Tikhonov's Theorem, which does not require exponential stability of the reduced system, is valid only on a compact time interval

The impetus for the systematic decomposition of the slow and fast subsystems in singularly perturbed systems can be readily extended to the separate control design of the slow and fast dynamics. As will be shown, the crucial idea is to compensate for the quasi-steady state in the fast variable.

$$
\dot{x} = f(x, z, u) \tag{98a}
$$

$$
\epsilon \dot{z} = g(x, z, u) \tag{98b}
$$

and suppose the equation $0 = g(x, z, u)$ has a unique root $z =$ where $\phi(x, u)$ in a domain *D* that contains the origin. A separated slow and fast design is succinctly captured in the composite control

$$
u = u_s + u_f \tag{99}
$$

where u_s is a slow control function of x (109) and (110), respectively.

$$
u_{\rm s} = \Gamma_{\rm s}(x) \tag{100}
$$

$$
u_{\rm f} = \Gamma_{\rm f}(x, z) \tag{101}
$$

Applying the control in Eq. (99) to the full model in Eq. these eigenvalues as ϵ tends to zero. (98), we obtain The composite feedback control is also fundamental in

$$
\dot{x} = f(x, z, \Gamma_{\rm s}(x) + \Gamma_{\rm f}(x, z)) \tag{102a}
$$

$$
\epsilon \dot{z} = g(x, z, \Gamma_{\rm s}(x) + \Gamma_{\rm f}(x, z)) \tag{102b}
$$

The fast control $\Gamma_f(x, z)$ must guarantee that $z = \phi(x, \Gamma_s(x))$ is a unique solution to the equation

$$
0 = g(x, z, \Gamma_{\rm s}(x) + \Gamma_{\rm f}(x, z))
$$
\n(103)

in the domain *D*. Furthermore, we require the fast control to be inactive on the manifold in Eq. (103), that is,

$$
\Gamma_{\mathbf{f}}(x, \phi(x, \Gamma_{\mathbf{s}}(x))) = 0 \tag{104}
$$

Then the slow and fast subsystems become, respectively, ponents as

$$
\dot{x} = f(x, \phi(x, \Gamma_s(x)), \Gamma_s(x))
$$
\n(105)

$$
\epsilon \dot{z} = g(x, z, \Gamma_{\rm s}(x) + \Gamma_{\rm f}(x, z)) \tag{106}
$$

To obtain controllers so that the equilibrium $(x = 0, z = 0)$ is asymptotically stable, $\Gamma_s(x)$ must be designed so that a Lyapunov function *V*(*x*) satisfying Eq. (85) can be found for the slow subsystem in Eq. (105), and $\Gamma_f(x, z)$ must be designed so that a Lyapunov function $W(x, z)$ satisfying Eq. (86) can be that a Lyapunov function $W(x, z)$ satisfying Eq. (86) can be
found for the fast subsystem in Eq. (106). Furthermore, the
found for the fast subsystem in Eq. (106). Furthermore, the interconnection conditions corresponding to Eqs. (90) and (91) and must be satisfied, so that a composite Lyapunov function similar to Eq. (88) can be used to establish the asymptotic stability of the equilibrium.

Specializing the composite control design to the linear sin- From the subsystems in Eqs. (43) and (44) and the decomposigularly perturbed system in Eq. (29), we design the slow and tion in Eq. (115), the linear quadratic regulator problem in fast controls as Eq. (113) can be solved from two lower-order subproblems.

$$
\Gamma_{\rm s} = G_0 x = G_0 \xi + O(\epsilon) \tag{107}
$$

$$
\Gamma_{\rm f} = G_2[z + A_{22}^{-1}(A_{21}x + B_{22}G_0x)] \triangleq G_2y \tag{108}
$$

such that the closed-loop subsystems in Eqs. (43) and (44) to $O(\epsilon)$

$$
\xi = (A_0 + B_0 G_0)\xi \tag{109}
$$

$$
\epsilon \dot{y} = (A_{22} + B_2 G_2) y \tag{110}
$$

Consider the nonlinear singularly perturbed system have the desired properties. Thus the composite control in Eq. (99) is

$$
u = G_0 x + G_2 [z + A_{22}^{-1} (A_{21} + B_2 G_0) x] = G_1 x + G_2 z \tag{111}
$$

$$
G_1 = (I_m + G_2 A_{22}^{-1} B_2) G_0 + G_2 A_{22}^{-1} A_{21}
$$
 (112)

When Eq. (111) is applied to the full-order model in Eq. (29), the slow and fast dynamics of the closed-loop system are approximated to $O(\epsilon)$ by the slow and fast subsystems in Eqs.

If the pairs (A_0, B_0) and (A_{22}, B_2) are completely controlla-
ble, the results here point readily to a two-time-scale poleand u_f is a fast control function of both x and z
placement design in which G_0 and G_2 are designed separately
to place the slow eigenvalues of $A_0 + B_0 G_0$ and the fast eigen*values of* $(A_{22} + B_2G_2)/\epsilon$ at the desired locations. Then the eigenvalues of the closed-loop full-order system will approach

> near-optimal control design of linear quadratic regulators for two-time-scale systems. Consider the optimal control of the linear singularly perturbed system in Eq. (29) to minimize the performance index

$$
J(u) = \frac{1}{2} \int_0^{\infty} (q^T q + u^T R u) dt, \quad R > 0
$$
 (113)

where

$$
q(t) = C_1 x(t) + C_2 z(t)
$$
 (114)

Following the slow and fast subsystem decomposition in Eqs. (43) and (44), we separate $q(t)$ into its slow and fast com-

$$
\dot{x} = f(x, \phi(x, \Gamma_s(x)), \Gamma_s(x))
$$
\n(105)\n
\n
$$
q(t) = q_s(t) + q_f(t) + O(\epsilon)
$$
\n(115)

where

$$
q_{s}(t) = C_0 \xi + D_0 u_s \tag{116}
$$

 $C_0 = C_1 + C_2 A_{22}^{-1} A_{21}, \quad D_0 = -C_2 A_{22}^{-1} B_2$

$$
q_{\rm f}(t) = C_2 y \tag{118}
$$

S low Regulator Problem

Find the slow control u_s for the slow subsystem in Eqs. (43) and (116) to minimize

$$
J_s(u_s) = \frac{1}{2} \int_0^\infty (q_s^T q_s + u_s^T R u_s) dt, \quad R > 0
$$

=
$$
\frac{1}{2} \int_0^\infty (\xi^T C_0^T C_0 \xi + 2 u_s^T D_0^T C_0^T \xi + u_s^T R_0 u_s) dt
$$
 (119)

$$
R_0 = R + D_0^T D_0 \tag{120}
$$

If the triple (C_0, A_0, B_0) is stabilizable and detectable (observable), then there exists a unique positive-semidefinite In Theorem 1.1, an asymptotic expansion exists for the so-

$$
0 = -K_{\rm s}(A_0 - B_0 R_0^{-1} D_0^T C_0) - (A_0 - B_0 R_0^{-1} D_0^T C_0)^T K_{\rm s}
$$

$$
+ K_{\rm s} B_0 R_0^{-1} B_0 K_{\rm s} - C_0^T (I - D_0 R_0^{-1} N_0^T) C_0 \tag{121}
$$

$$
u_{\rm s} = -R_0^{-1} (D_0^T C_0 + B_0^T K_{\rm s}) \xi = G_0 \xi \tag{122}
$$

$$
J_{\rm f}(u_{\rm f}) = \frac{1}{2} \int_0^\infty (q_{\rm f}^T q_{\rm f} + u_{\rm f}^T R u_{\rm f}) dt, \quad R > 0
$$

=
$$
\frac{1}{2} \int_0^\infty (y^T C_2^T C_2 y + u_{\rm f}^T R u_{\rm f}) dt
$$
 (123)

If the triple (C_2, A_{22}, B_2) is stabilizable and detectable (ob-
servable), then there exists a unique positive-semidefinite P -machine power system in the second-order form with damp-(positive-definite) stabilizing solution K_f of the matrix Riccati ing neglected equation

$$
0 = -K_f A_{22} - A_{22}^T K_f + K_f B_2 R^{-1} B_2 K_f - C_2^T C_2 \tag{124}
$$

$$
u_{\rm f} = -R^{-1}B_2^T K_{\rm f} y = G_{22} y \tag{125}
$$

The following results are from Reference (8). case, we decompose *K* into

- and detectable (observable), then there exists an $\epsilon^* > 0$ for the linear regulator problem (113) with an optimal ternal connection strength.
-

$$
u_{c} = -[(I - R^{-1}B_{2}^{T}K_{f}A_{22}^{-1}B_{2})R_{0}^{-1}(D_{0}^{T}C_{0} + B_{0}^{T}K_{s})
$$

+ $R^{-1}B_{2}^{T}K_{f}A_{22}^{-1}A_{21}]x - R^{-1}B_{2}^{T}K_{f}z$ (126) $K^{I} = \text{block-diag}(K_{1}^{I}, K_{2}^{I})$ (132)

applied to the system in Eq. (29) achieves an $O(\epsilon^2)$ ap-

$$
J(u_{\rm c}) = J_{\rm opt} + O(\epsilon^2)
$$
 (127)

3. If A_{22} is stable, then the slow control in Eq. (122)

$$
u_{\rm s} = -R_0^{-1} (D_0^T C_0 + B_0^T K_{\rm s}) x \tag{128}
$$

where applied to the system in Eq. (29) achieves an $O(\epsilon)$ approximation of J_{opt} , that is,

$$
J(us) = Jopt + O(\epsilon)
$$
 (129)

(positive-definite) stabilizing solution K_s of the matrix Riccati lution to the matrix Riccati equation associated with the full equation linear regulator problem. Theorem 1.3 is one of the robustness results with respect to fast unmodeled dynamics, that is, if the fast dynamics is asymptotically stable, a feedback control containing only the slow dynamics would not destabilize the fast dynamics.

and the optimal control is **APPLICATIONS TO LARGE POWER SYSTEMS**

In this section, we analyze a large power system as an example of time-scales arising in an interconnected systems. A **Fast Regulator Problem** power system dispersed over a large geographical area tends Find the fast control u_f for the fast subsystem in Eqs. (44) to have dense meshes of power networks serving heavily pop-
and (118) to minimize the fast subsystem in Eqs. (44) to have dense meshes of power transmission l disturbance, it is observed that groups of closely located machines would swing coherently at a frequency that is lower than the frequency of oscillation within the coherent groups. Singular perturbation techniques have been successfully applied to these large power networks to reveal this two-timescale behavior (9).

n-machine power system in the second-order form with damp-

$$
M\ddot{\delta} = K\delta \tag{130}
$$

where $\delta \in R^n$ is the machine rotor angle vector, *M* is the diagand the optimal control is **and** the optimal control is **and the stiffness matrix** determined by the network impedances. Assume that the sys m tem in Eq. (130) has *r* tightly connected areas, with the connections between the areas being relatively fewer. In this

Theorem 1.
$$
K = K^{\mathrm{I}} + \epsilon K^{\mathrm{E}} \tag{131}
$$

where K^I is the stiffness matrix due to the impedances inter-1. If the triples (C_0, A_0, B_0) and (C_2, A_{22}, B_2) are stabilizable nal to the areas, and K^E is the stiffness matrix due to the and detectable (observable), then there exists an $\epsilon^* > 0$ impedances external to the areas and scaled by the small pa-
such that for all $\epsilon \in (0, \epsilon^*]$, an optimal control exists rameter ϵ that represents the ratio o rameter ϵ that represents the ratio of the external to the in-

performance J_{opt} . For illustrative purposes, we let $r = 2$, with r_1 machines in 2. The composite control in Eq. (111) area 1 and r_2 machines in area 2. Arranging the machines in area 1 to appear first in δ , K^{I} has a block-diagonal structure

$$
KI = block-diag(K1I, K2I)
$$
 (132)

A particular property of the $r_i \times r_i$ matrix K^{I}_i , $i=1,\,2,$ is that each of the rows in K_i^{I} will sum to zero. If $\epsilon = 0$, this conservaproximation of *J*_{ont}, that is, tion property results in a slow mode in each area. When $\epsilon \neq 0$, the slow mode from each area will interact to form the I_0 *low-frequency interarea oscillatory mode.*

To reveal this slow dynamics, we define a grouping matrix

$$
U = \begin{bmatrix} \mathbf{1}_{r_1} & 0\\ 0 & \mathbf{1}_{r_2} \end{bmatrix} \tag{133}
$$

where $\mathbf{1}_r$ is an $r_i \times 1$ column vector of all ones. Using *U*, we The composite control has also been applied to solve opti-
introduce the aggregate machine angles weighted according mal regulator problems for nonli to the inertia as the slow variables systems (15). Recently, composite control results for *H*_n opti-

$$
\delta_{\mathbf{a}} = (U^T M U)^{-1} U^T M \delta = C \delta \tag{134}
$$

and the difference angles with respect to the first machine in by a high-gain control (18).
each area as the fast variables Singular perturbation m

$$
\delta_{\rm d} = G \delta \tag{135}
$$

$$
G = \begin{bmatrix} -\mathbf{1}_{r_1-1} & I_{r_1-1} & 0 & 0\\ 0 & 0 & -\mathbf{1}_{r_2-1} & I_{r_2-1} \end{bmatrix}
$$
 (136)

 $\text{Noting that } CM^{-1}K^{\text{I}} = 0 \text{ and } K^{\text{I}}$ null space of $M^{-1}K^{\text{I}}$ and *U* is in the right null space of K^{I} , the in Ref. 23. system in Eq. (130) in the new variables become A topic not covered here is the filtering and stochastic con-

$$
\ddot{\delta}_a = \epsilon C M^{-1} K^E U \delta_a + \epsilon C M^{-1} K^E G^+ \delta_d \tag{137a}
$$

$$
\ddot{\delta}_d = \epsilon G M^{-1} K^E U \delta_a + (G M^{-1} K^{\mathrm{T}} G^+ + \epsilon G M^{-1} K^{\mathrm{E}} G^+) \delta_d \quad (137b)
$$

where $G^+ = G^T(GG^T)^{-1}$. The system in Eq. (137) clearly points
to the two-time-scale behavior in which the right-hand side of
Eq. (137a) is $O(\epsilon)$, indicating that δ_a is a slow variable. The
method can readily be exte

method can readily be extended to systems with $r > 2$ areas.

Based on this time-scale interpretation, a grouping algo-

Based on this time-scale interpretation, a grouping algo-

time-scale methods (25). More development

than those discussed in this article. For our concluding re- these new problems. marks, we briefly comment on some of these applications as extensions of the results already discussed. **BIBLIOGRAPHY** The two-time-scale properties can also be used to charac-

terize transfer functions of singularly perturbed systems (11).
In a linear discrete-time two-time-scale system, the slow dy-
In a linear discrete-time two-time-scale system, the slow dy-
tion Methods in Control: Analysis namics arise from the system eigenvalues close to the unit Press, 1986.
circle, whereas the fast dynamics are a result of those eigen-
 $\frac{1}{2}$ P K Keket. circle, whereas the fast dynamics are a result of those eigen-
values close to the origin. The two-time-scale analysis for con-
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timeous-time systems can be readily ext

An application of the stability results is to determine the 4. IEEE Committee Report, Excitation system models for power system robustness of a control design. For example, Khalil (13) shows tem stability studies, IEEE Tra a simple example where a static output feedback designed 494–509, 1981. without considering the parasitic effects would lead to an in-
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mal regulator problems for nonlinear singularly perturbed mal control of singularly perturbed systems have been obtained (16, 17). The composite control idea can also be used to establish the time scales in a closed-loop system induced

Singular perturbation methods also have significant applications in flight control problems (19, 20). For example, two-
point boundary-value problems arising from trajectory optimization can be solved by treating the fast maneuvers and the slower cruise dynamics separately.

> The slow coherency and aggregation technique is also applicable to other large-scale systems, such as Markov chains (21) and multi-market economic systems (22). These systems belong to the class of singularly perturbed systems in the nonstandard form, of which an extended treatment can be found in Ref. 23.

trol of singularly perturbed systems with input noise. As $\epsilon \rightarrow 0$, the fast dynamics will tend to a white noise. Although the problem can be studied in two-time-scales, the convergence of the optimal solution requires that the noise input in where $G^+ = G^T(GG^T)^{-1}$. The system in Eq. (137) clearly points G_1 (94)

contribute to the advances of modern control theory. As new **FURTHER READING control problems are proposed and new applications are dis**covered for systems with time-scales, we expect that singular Singular perturbation techniques have been successfully ap- perturbation methods will also be extended accordingly to plied to the analysis and design of many control systems other provide simpler models to gain useful design insights into

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