

16.1-1

(a) Cylindrical part; x-direction axial:

$$\sigma_x = \frac{PR}{2t} \quad \sigma_y = \frac{PR}{t} \quad \tau_{xy} = 0$$

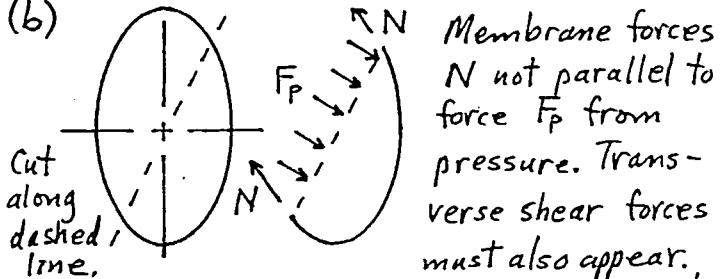
$$N_x = \sigma_x t = \frac{PR}{2}, \quad N_y = \sigma_y t = PR, \quad N_{xy} = 0$$

Hemispherical cap: if n and s are any two surface-tangent directions,

$$\sigma_n = \sigma_s = \frac{PR}{2t}, \quad \tau_{ns} = 0$$

$$N_n = N_s = \frac{PR}{2}, \quad N_{ns} = 0$$

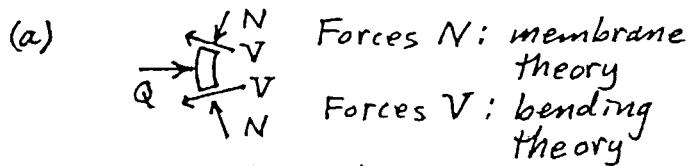
(b)



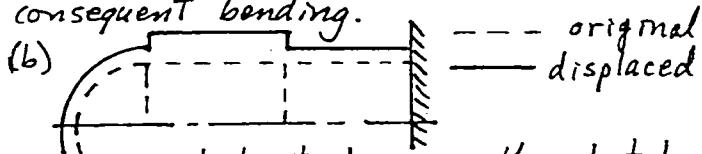
Membrane forces
not parallel to
force F_p from
pressure. Trans-
verse shear forces
must also appear.

Transverse shear forces, and consequent
bending, vanish only for circular cylinder.

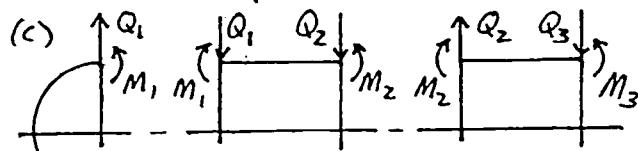
16.1-2



Shrink slice shown to zero size to span only point load Q . Then forces N are parallel and must become infinite to carry Q . Can't happen. Q carried by V , with consequent bending.



Also, but not shown in the sketch, is axial disp. associated with axial ϵ .



Q_i & M_i are line loads uniformly distributed around parallels. Direction of M_2 guessed at. Analysis will show $M_1=0$.

16.2-1

$$\epsilon_s = \frac{d}{ds}(\delta_a + \delta_c) + \frac{w}{R} = \frac{d}{ds}\left[u\left(1 + \frac{z}{R}\right) - zw_s\right] + \frac{w}{R}$$

$$\epsilon_s = u_{,s} + z \frac{u_{,s}}{R} - z \frac{u}{R^2} R_{,s} - zw_{,ss} + \frac{w}{R}$$

$$\epsilon_s = u_{,s} + \frac{w}{R} + z\left(\frac{u_{,s}}{R} - w_{,ss} - \frac{u}{R^2} R_{,s}\right)$$

Term not in Eq. 16.2-2 \rightarrow

16.2-2

$$(a) \epsilon_m = 0 \text{ gives } u_{ss} = -\frac{w}{R} \text{ or } \frac{u_{ss}}{R} = -\frac{w}{R^2}$$

$$K = 0 \text{ gives } \frac{u_{ss}}{R} - w_{ss} = 0 \text{ or } w_{ss} + \frac{w}{R^2} = 0$$

Integrate; let $\phi = \frac{s}{R}$:

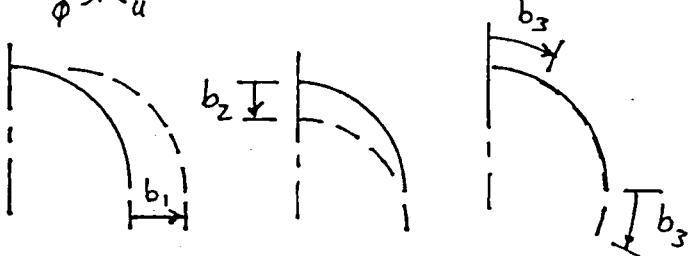
$$w = b_1 \sin \phi - b_2 \cos \phi; \text{ hence } u_{ss} = \frac{-b_1 \sin \phi + b_2 \cos \phi}{R}$$

$$\text{or } u_{ss} = b_1 \sin \phi + b_2 \cos \phi$$

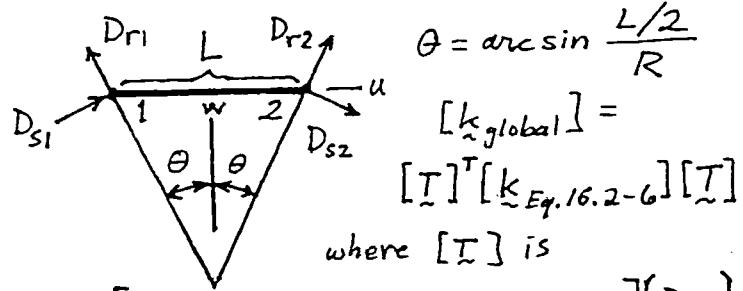
$$u = b_1 \cos \phi + b_2 \sin \phi + b_3$$

$$(b) D_2 \quad D_1 = u \cos \phi + w \sin \phi = b_1 + b_3 \cos \phi$$

$$D_2 = -u \sin \phi + w \cos \phi = -b_2 - b_3 \sin \phi$$



16.2-3



$$\theta = \arcsin \frac{L/2}{R}$$

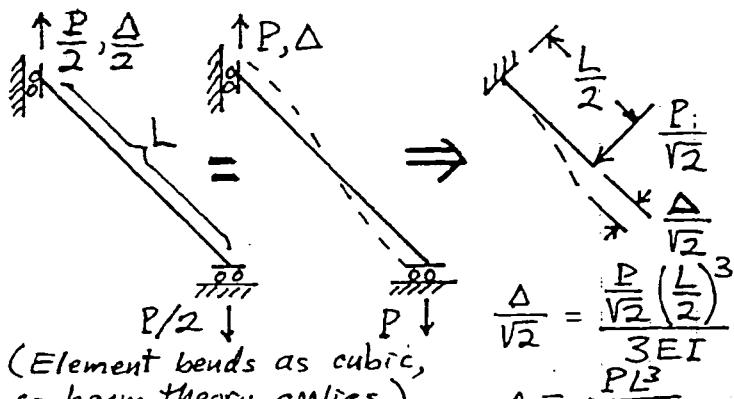
$$[k_{\text{global}}] = [T]^T [k_{\text{Eq. 16.2-6}}] [T]$$

where $[T]$ is

$$\begin{Bmatrix} u_1 \\ u_z \\ w_1 \\ \psi_1 \\ w_z \\ \psi_z \end{Bmatrix} = \begin{bmatrix} \cos \theta & -\sin \theta & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & \cos \theta & \sin \theta & 0 \\ \sin \theta & \cos \theta & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & -\sin \theta & \cos \theta & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} D_{s1} \\ D_{r1} \\ \psi_1 \\ D_{s2} \\ D_{r2} \\ \psi_z \end{Bmatrix}$$

16.2-4

$$\text{Exact: } \Delta = 0.1488 PR^3/EI$$



(Element bends as cubic,
so beam theory applies.)

$$\frac{\Delta}{\sqrt{2}} = \frac{P}{3EI} \left(\frac{L}{2}\right)^3$$

$$\Delta = \frac{PL^3}{24EI}$$

$$(a) L = \sqrt{2}R, \Delta = 0.1179 PR^3/EI \quad 20.8\% \text{ low}$$

$$(b) L = \frac{\pi R}{2}, \Delta = 0.1615 PR^3/EI \quad 8.5\% \text{ high}$$

16.2-5

$$u = a_1 + a_2 s \quad \epsilon_m = a_2 + \frac{a_3 + a_4 s + a_5 s^2 + a_6 s^3}{R}$$
$$w = a_3 + a_4 s + a_5 s^2 + a_6 s^3 \quad \text{or}$$

$$\epsilon_m = b_1 + b_2 s + b_3 s^3 + b_4 s^4$$

where $b_1 = a_2 + \frac{a_3}{R}$, $b_2 = \frac{a_4}{R}$, $b_3 = \frac{a_5}{R}$, $b_4 = \frac{a_6}{R}$

$$\epsilon_m^2 = b_1^2 + b_2^2 s^2 + b_3^2 s^4 + b_4^2 s^6 + 2b_1 b_3 s^2 + 2b_2 b_4 s^4$$

+ (terms with odd powers of s)

Odd powers will integrate to zero, so ignore.

Let β_i = constants; thus

$$\int_{-L/2}^{L/2} \epsilon_m^2 ds = \beta_1 b_1^2 + \beta_2 b_2^2 + (\beta_3 b_3^2 + \beta_4 b_4^2 + \beta_5 b_1 b_3 + \beta_6 b_2 b_4)$$

$$\int_{-L/2}^{L/2} \epsilon_m^2 ds = 0 \text{ implies } b_1 = b_2 = b_3 = b_4 = 0$$

i.e. that $a_2 + \frac{a_3}{R} = a_4 = a_5 = a_6 = 0$

16.2-6

$$\text{Eqs. 16.2-9.}$$

$$\Psi = w_{ss} = a_4 + 2a_5 s + 3a_6 s^2$$

node	<u>s</u>
1	-L/2
2	+L/2

$$\begin{Bmatrix} u_1 \\ w_1 \\ \Psi_1 \\ u_2 \\ w_2 \\ \Psi_2 \end{Bmatrix} = \begin{bmatrix} 1 & -\frac{L}{2} & 0 & 0 & 0 & \frac{0}{L^3} \\ 0 & 0 & 1 & -\frac{L}{2} & \frac{L^2}{4} & -\frac{L^3}{8} \\ 0 & 0 & 0 & 1 & -L & -\frac{3L^2}{2} \\ 1 & \frac{L}{2} & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & \frac{L}{2} & \frac{L^2}{4} & \frac{L^3}{8} \\ 0 & 0 & 0 & 1 & L & \frac{3L^2}{2} \end{bmatrix} \begin{Bmatrix} a_1 \\ a_2 \\ a_3 \\ a_4 \\ a_5 \\ a_6 \end{Bmatrix}$$

(b) Eqs. 16.2-12

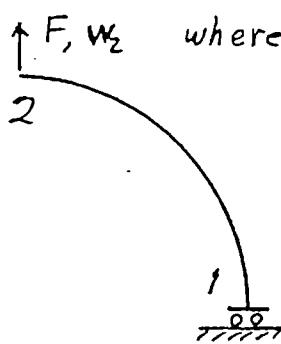
$$\Psi = w_{ss} = -\frac{1}{R}(2a_3 + 6a_4\phi + 12a_5\phi^2 + 20a_6\phi^3)$$

Node 1: $\phi = -L/2R$; Node 2: $\phi = +L/2R$

Let $\alpha = L/2R$

$$\begin{Bmatrix} u_1 \\ w_1 \\ \Psi_1 \\ u_2 \\ w_2 \\ \Psi_2 \end{Bmatrix} = \begin{bmatrix} 1 & -\alpha & \alpha^2 & -\alpha^3 & \alpha^4 & -\alpha^5 \\ 0 & -1 & 2\alpha & -3\alpha^2 & 4\alpha^3 & -5\alpha^4 \\ 0 & 0 & -\frac{2}{R} & \frac{6\alpha}{R} & -\frac{12\alpha^2}{R} & \frac{20\alpha^3}{R} \\ 1 & \alpha & \alpha^2 & \alpha^3 & \alpha^4 & \alpha^5 \\ 0 & -1 & -2\alpha & -3\alpha^2 & -4\alpha^3 & -5\alpha^4 \\ 0 & 0 & -\frac{2}{R} & -\frac{6\alpha}{R} & -\frac{12\alpha^2}{R} & -\frac{20\alpha^3}{R} \end{bmatrix} \begin{Bmatrix} a_1 \\ a_2 \\ a_3 \\ a_4 \\ a_5 \\ a_6 \end{Bmatrix}$$

16.2-7



where $F = \frac{P}{2}$ (P = a diametral force on the complete ring)

$$u_1 = u_2 = \psi_1 = \psi_2 = 0$$

$$\frac{EI}{R^3} \begin{bmatrix} k_{22} & k_{25} \\ k_{52} & k_{55} \end{bmatrix} \begin{Bmatrix} w_1 \\ w_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ F \end{Bmatrix} \quad k_{52} = k_{25}, k_{22} = k_{55}$$

$$\text{First eq. gives } w_1 = -\frac{k_{25}}{k_{22}} w_2$$

Second eq. becomes

$$\left(-\frac{k_{25}^2}{k_{22}} + k_{55} \right) w_2 = F \frac{R^3}{EI}$$

$$\beta = \frac{\pi R/2}{2R} = \frac{\pi}{4} \quad \text{hence} \quad k_{22} = k_{25} = 42.9128$$

$$k_{25} = 39.3948$$

$$\text{Therefore } w_2 = 0.1482 F \frac{R^3}{EI}$$

$$\text{Change in dia. of ring} = 2w_2 = 2 \left(0.1482 \frac{P}{2} \frac{R^3}{EI} \right) = 0.1482 \frac{PR^3}{EI}$$

(0.4% low)

$$\text{exact: } 0.1488 \frac{PR^3}{EI}$$

16.2-8

In elements of Eqs. 16.2-9,
constraint $\epsilon_m = 0$ is not explicit; it is
implicitly imposed (e.g. Eq. 16.2-11). If
 ϵ_m were ignored, membrane stiffness would
be zero, not infinite ($\epsilon_m = 0$); singular $[k]$.

16.2-9

$$(a) \quad \epsilon_m = u_{,s} + \frac{w}{R} = a_2 + \frac{a_3 + a_4 s}{R}$$

$$\gamma_{zs} = w_{,s} - \beta = a_4 - (a_5 + a_6 s)$$

$\epsilon_m = 0$ for all s implies

$$a_2 + \frac{a_3}{R} = a_4 = 0$$

$$\gamma_{zs} = 0 \text{ for all } s \text{ implies } \left. \begin{array}{l} a_2 + \frac{a_3}{R} = a_4 = 0 \\ a_4 - a_5 = a_6 = 0 \end{array} \right\} \therefore a_5 = 0$$

For a straight el., $R \rightarrow \infty$, and

$$a_2 = a_4 - a_5 = a_6 = 0.$$

(b) Evaluate at $s=0$:

$$\left. \begin{array}{l} \epsilon_m = a_2 + \frac{a_3}{R} \\ \gamma_{zs} = a_4 - a_5 \end{array} \right\} \text{constraints } a_2 + \frac{a_3}{R} = a_4 - a_5 = 0$$

16.2-10

$$\gamma_{25} = w_{ss} - \beta = \frac{2a_5}{L} + \frac{4a_6}{L}\xi - a_7 - a_8\xi - a_9\xi^2$$

$$\gamma_{25} = \left(\frac{2a_5}{L} - a_7\right) + \left(\frac{4a_6}{L} - a_8\right)\xi - (a_9)\xi^2 \quad (A)$$

$\gamma_{25} = 0$ for all ξ implies that all three
parenthetic expressions in Eq. (A) vanish.

Note; $a_9 = 0$ implies $\beta_{ss} = 0$.

Rewrite Eq. (A):

$$0 = \left(\frac{2a_5}{L} - a_7 - \frac{a_9}{3}\right) + \left(\frac{4a_6}{L} - a_8\right)\xi - \left[\xi^2 - \frac{1}{3}\right]a_9$$

At Gauss pts, $\xi = \pm \frac{1}{\sqrt{3}}$, $\left[\xi^2 - \frac{1}{3}\right] = 0$.

Constraints are then $\frac{2a_5}{L} - a_7 - \frac{a_9}{3} = 0$

$$\frac{4a_6}{L} - a_8 = 0$$

16.3-1

$$U_m = \frac{1}{2} \int_{-L/2}^{L/2} \frac{Et}{1-\nu^2} \begin{Bmatrix} \epsilon_{ms} \\ \epsilon_{mo} \end{Bmatrix}^T \begin{bmatrix} 1 & -\nu \\ -\nu & 1 \end{bmatrix} \begin{Bmatrix} \epsilon_{ms} \\ \epsilon_{mo} \end{Bmatrix} 2\pi R ds$$

Unrestrained axially: $\epsilon_{ms} = -\nu \epsilon_{mo} = -\nu \frac{w}{R}$

$$U_m = \frac{1}{2} \int_{-L/2}^{L/2} \frac{Et}{1-\nu^2} \frac{w^2}{R^2} \begin{Bmatrix} -\nu \\ 1 \end{Bmatrix}^T \begin{bmatrix} 1 & -\nu \\ -\nu & 1 \end{bmatrix} \begin{Bmatrix} -\nu \\ 1 \end{Bmatrix} 2\pi R ds$$

$$U_m = \frac{1}{2} \int_{-L/2}^{L/2} \frac{Et}{1-\nu^2} \frac{w^2}{R^2} (1-\nu^2) 2\pi R ds = \frac{1}{2} \int \frac{Et}{R^2} w^2 dA$$

Compare with Eq. 8.8-1:

foundation modulus is $B = \frac{Et}{R^2}$

16.3-2

$$U = \frac{1}{2} \int_{-L/2}^{L/2} E \epsilon_{m\theta}^2 (2\pi R t) ds + \frac{1}{2} \int_{-L/2}^{L/2} D K_s^2 (2\pi R) ds \quad (\text{see Fig. 16.3-1b})$$

where, from Eq. 16.3-3, $\epsilon_{m\theta} = \frac{w}{R}$, $K_s = \frac{d^2 w}{ds^2}$. Also $D = \frac{Et^3}{12(1-\nu^2)}$

$$[k] = [k_m] + [k_b] = \frac{2\pi Et}{R} \int_{-L/2}^{L/2} [N]^T [N] ds + 2\pi DR \int_{-L/2}^{L/2} [N_{ss}]^T [N_{ss}] ds$$

$[k_m]$ has the form of a mass matrix (Eq. 11.2-6) with $2\pi Et/R$ in place of m ; thus

$$[k_m] = \frac{2\pi EtL}{420R} \begin{bmatrix} 156 & 22L & 54 & -13L \\ & 4L^2 & 13L & -3L^2 \\ & & 156 & -22L \\ & & & 4L^2 \end{bmatrix}$$

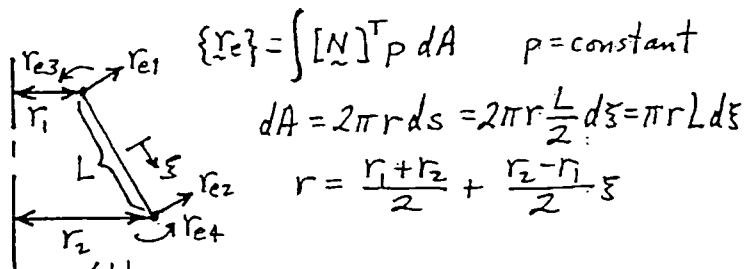
symm.

$[k_b]$ has the form of a standard beam element matrix with $2\pi RD$ in place of EI ; thus

$$[k_b] = \frac{2\pi RD}{L^3} \begin{bmatrix} 12 & GL & -12 & GL \\ & 4L^2 & -6L & 2L^2 \\ & & 12 & -6L \\ & & & 4L^2 \end{bmatrix}$$

symm.

16.3-3



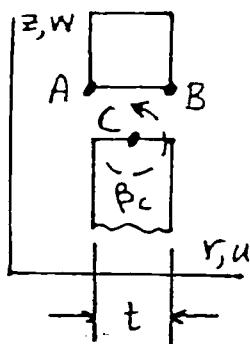
$$\{r_e\} = \begin{cases} +1 & \\ -1 & \end{cases} \frac{P\pi L}{2} \left\{ \begin{array}{l} (1-\xi)/2 \\ (1+\xi)/2 \\ (1-\xi^2)L/8 \\ -(1-\xi^2)L/8 \end{array} \right\} [(r_1+r_2) + (r_2-r_1)\xi] d\xi$$

$$\{r_e\} = \frac{P\pi L}{2} (r_1+r_2) \left\{ \begin{array}{l} 1 \\ \frac{L}{8} \left(\frac{4}{3} \right) \\ -\frac{L}{8} \left(\frac{4}{3} \right) \end{array} \right\} + \frac{P\pi L}{2} (r_2-r_1) \left\{ \begin{array}{l} -\frac{1}{3} \\ +\frac{1}{3} \\ 0 \\ 0 \end{array} \right\}$$

$$\{r_e\} = \frac{P\pi L}{2} (r_1+r_2) \left\{ \begin{array}{l} 1 \\ 1 \\ L/6 \\ -L/6 \end{array} \right\} + \frac{P\pi L (r_2-r_1)}{6} \left\{ \begin{array}{l} -1 \\ 1 \\ 0 \\ 0 \end{array} \right\}$$

16.3-4

(a)



Axially symmetric:

$$\begin{Bmatrix} w_c \\ u_c \\ \beta_c \end{Bmatrix} = [T] \begin{Bmatrix} w_A \\ u_A \\ w_B \\ u_B \end{Bmatrix}$$

3x4

where, following
arguments in Section
8.5, $[T]$ is

$$\begin{bmatrix} \frac{1}{2} & 0 & \frac{1}{2} & 0 \\ 0 & \frac{1}{2} & 0 & \frac{1}{2} \\ -\frac{1}{t} & 0 & \frac{1}{t} & 0 \end{bmatrix}$$

(b) Circumferential displacement vectors.

$$\begin{Bmatrix} w_c \\ u_c \\ v_c \\ \beta_c \end{Bmatrix} = \begin{bmatrix} \frac{1}{2} & 0 & 0 & \frac{1}{2} & 0 & 0 \\ 0 & \frac{1}{2} & 0 & 0 & \frac{1}{2} & 0 \\ 0 & 0 & \frac{1}{2} & 0 & 0 & \frac{1}{2} \\ -\frac{1}{t} & 0 & 0 & \frac{1}{t} & 0 & 0 \end{bmatrix} \begin{Bmatrix} w_A \\ u_A \\ v_A \\ w_B \\ u_B \\ v_B \end{Bmatrix}$$

(c) For the case without axial symmetry, we write

$$\begin{bmatrix} \frac{1}{2} & 0 & 0 & \frac{1}{2} & 0 & 0 & -1 & 0 & 0 & 0 \\ 0 & \frac{1}{2} & 0 & 0 & \frac{1}{2} & 0 & 0 & -1 & 0 & 0 \\ 0 & 0 & \frac{1}{2} & 0 & 0 & \frac{1}{2} & 0 & 0 & -1 & 0 \\ -\frac{1}{t} & 0 & 0 & \frac{1}{t} & 0 & 0 & 0 & 0 & 0 & -1 \end{bmatrix} \begin{Bmatrix} w_A \\ u_A \\ v_A \\ w_B \\ u_B \\ v_B \\ w_c \\ u_c \\ v_c \\ \beta_c \end{Bmatrix} = \{0\}$$

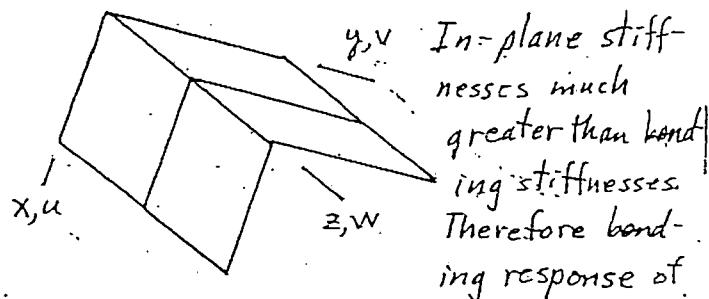
To account for other d.o.f. in
the structure above & below section
line, expand the rectangular matrix
left & right by adding zeros, &
expand the column vector up &
down by adding d.o.f.

If there is axial symmetry, we
can omit the information content
associated with v_A , v_B , and v_c .

16.4-1

Let s be a coordinate parallel to an element edge. Element-normal displacement is cubic or quadratic in s (depending on element type), while element-tangent (but edge-normal) displacement is linear in s . Thus there is incompatibility, most strongly if adjacent elements meet at a right angle. There is no incompatibility if elements are coplanar, and such a condition is approached as the mesh is indefinitely refined.

16.4-2



each side is modeled well if nodal d.o.f. Ψ_y and Ψ_x set to zero (retain Ψ_z). Indeed, each side could be analyzed separately, except that Ψ_z provides elastic support to element edges that lie on the z axis.

16.5-1

Let j and k refer to upper and lower surface nodes respectively. The N 's are given by Eqs. 6.4-1.

$$\begin{Bmatrix} x \\ y \\ z \end{Bmatrix} = \sum_{k=1}^8 N_k \frac{1-z}{2} \begin{Bmatrix} x_k \\ y_k \\ z_k \end{Bmatrix} + \sum_{j=1}^8 N_j \frac{1+z}{2} \begin{Bmatrix} x_j \\ y_j \\ z_j \end{Bmatrix}$$

(a) $[\underline{\underline{J}}] = \begin{bmatrix} x_{,z} & y_{,z} & z_{,z} \\ x_{,\eta} & y_{,\eta} & z_{,\eta} \\ x_{,z} & y_{,z} & z_{,z} \end{bmatrix}$ From Eq. 16.5-2,

$$\begin{Bmatrix} x \\ y \\ z \end{Bmatrix} = \begin{bmatrix} N_i & 0 & 0 & N_i \zeta/2 & 0 & 0 \\ 0 & N_i & 0 & 0 & N_i \zeta/2 & 0 \\ 0 & 0 & N_i & 0 & 0 & N_i \zeta/2 \end{bmatrix} \begin{Bmatrix} x_i \\ y_i \\ z_i \\ t_i l_{3i} \\ t_i m_{3i} \\ t_i n_{3i} \end{Bmatrix}$$

Terms in (e.g.) col. 1 of $[\underline{\underline{J}}]$ are

6×1

$$\frac{\partial x}{\partial z} = \sum [N_{i,z} \ 0 \ 0 \ \frac{\zeta}{2} N_{i,z} \ 0 \ 0] \{ \}$$

$$\frac{\partial x}{\partial \eta} = \sum [N_{i,\eta} \ 0 \ 0 \ \frac{\zeta}{2} N_{i,\eta} \ 0 \ 0] \{ \}$$

$$\frac{\partial x}{\partial \zeta} = \sum [0 \ 0 \ 0 \ \frac{N_i}{2} \ 0 \ 0] \{ \}$$

(the N_i are functions of ζ and η but not of x). Sum over no. of nodes in element.

(b) Here $z_i = l_{3i} = m_{3i} = 0$, $n_{3i} = 1$, $t_i = t$.

Note that $\sum N_i = 1$ and $\sum N_{i,z} = \sum N_{i,\eta} = \sum N_{i,\zeta} = 0$.

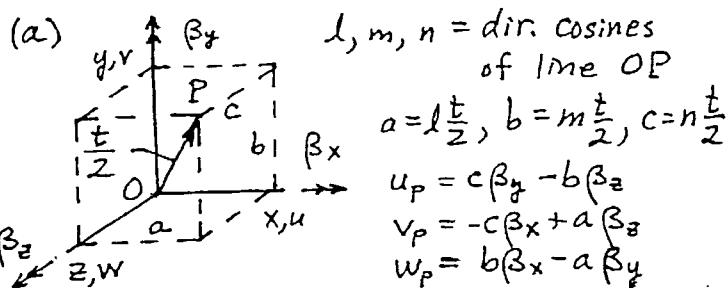
$$\frac{\partial x}{\partial z} = \sum N_{i,z} x_i \quad \frac{\partial y}{\partial z} = \sum N_{i,z} y_i \quad \frac{\partial z}{\partial z} = 0$$

$$\frac{\partial x}{\partial \eta} = \sum N_{i,\eta} x_i \quad \frac{\partial y}{\partial \eta} = \sum N_{i,\eta} y_i \quad \frac{\partial z}{\partial \eta} = 0$$

$$\frac{\partial x}{\partial \zeta} = 0 \quad \frac{\partial y}{\partial \zeta} = 0 \quad \frac{\partial z}{\partial \zeta} = \frac{t}{2}$$

Thus $[\underline{\underline{J}}]$ becomes as for a flat plate.

16.5-3



$$\begin{Bmatrix} u \\ v \\ w \end{Bmatrix} = \sum N_i \begin{Bmatrix} u_i \\ v_i \\ w_i \end{Bmatrix} + \sum N_i \frac{t_i}{2} \begin{Bmatrix} 0 & n_i & -m_i \\ -n_i & 0 & l_i \\ m_i & -l_i & 0 \end{Bmatrix} \begin{Bmatrix} \beta_x \\ \beta_y \\ \beta_z \end{Bmatrix}$$

(b) E.g., if V_3 is x-parallel, then $l_i = 1$, $m_i = n_i = 0$, β_y becomes α and β_z becomes β (see Fig. 16.5-2). And, in Fig. 16.5-2b, V_1 becomes y-parallel & V_2 becomes z-parallel. Thus, $\{\underline{M}\}\{\underline{\beta}\}$ products in above eq. & in Eq. 16.5-6 become respectively.

$$\begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & 1 \\ 0 & -1 & 0 \end{bmatrix} \begin{Bmatrix} \beta_x \\ \alpha \\ \beta \end{Bmatrix} \text{ and } \begin{bmatrix} 0 & 0 \\ 0 & 1 \\ -1 & 0 \end{bmatrix} \begin{Bmatrix} \alpha \\ \beta \end{Bmatrix} \text{ These agree.}$$

16.5-4

$$\begin{Bmatrix} \epsilon_x \\ \vdots \\ \delta_{zx} \end{Bmatrix}_{6 \times 3} = [\tilde{H}] \begin{Bmatrix} u_x \\ \vdots \\ w_z \end{Bmatrix}_{3 \times 3} = [\tilde{H}] \begin{Bmatrix} \Gamma & 0 & 0 \\ 0 & \Gamma & 0 \\ 0 & 0 & \Gamma \end{Bmatrix} \begin{Bmatrix} u_x \\ \vdots \\ w_z \end{Bmatrix}$$

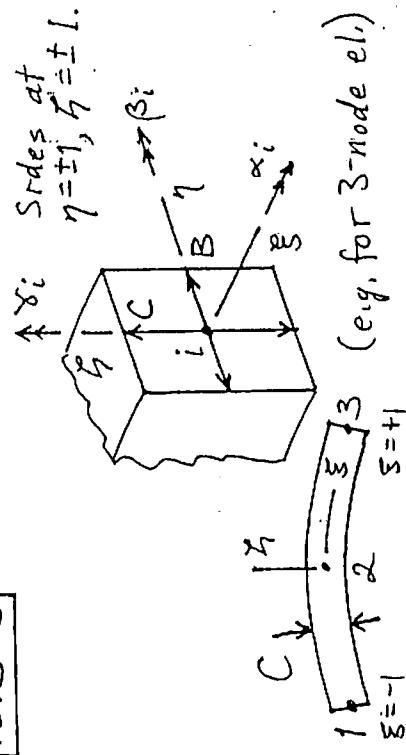
where $[\tilde{J}] = [\tilde{\Gamma}]^{-1}$ is given by Eq. 6.5-2,

$[\tilde{H}]$ is given by Eq. 6.5-3, and the column vector $\begin{Bmatrix} u_x \\ \vdots \\ w_z \end{Bmatrix}^T$ by Eq. 16.5-9. We want

$$[\tilde{B}_i] = \begin{Bmatrix} \Gamma & 0 & 0 \\ 0 & \Gamma & 0 \\ 0 & 0 & \Gamma \end{Bmatrix} \text{ [square array in Eq. 16.5-9]}$$

Let $A_i = \Gamma_{11}N_{i,5} + \Gamma_{12}N_{i,7}$, $B_i = \Gamma_{21}N_{i,5} + \Gamma_{22}N_{i,7}$, $C_i = \Gamma_{31}N_{i,5} + \Gamma_{32}N_{i,7}$. Then $[\tilde{B}_i] =$

$$\begin{bmatrix} A_i & 0 & 0 & -(A_i\gamma + \Gamma_{13}N_i)l_2t_i/2 & (A_i\gamma + \Gamma_{13}N_i)l_1t_i/2 \\ B_i & 0 & 0 & -(B_i\gamma + \Gamma_{23}N_i)l_2t_i/2 & (B_i\gamma + \Gamma_{23}N_i)l_1t_i/2 \\ C_i & 0 & 0 & -(C_i\gamma + \Gamma_{33}N_i)l_2t_i/2 & (C_i\gamma + \Gamma_{33}N_i)l_1t_i/2 \\ 0 & A_i & 0 & -(A_i\gamma + \Gamma_{13}N_i)m_2t_i/2 & (A_i\gamma + \Gamma_{13}N_i)m_1t_i/2 \\ 0 & B_i & 0 & -(B_i\gamma + \Gamma_{23}N_i)m_2t_i/2 & (B_i\gamma + \Gamma_{23}N_i)m_1t_i/2 \\ 0 & C_i & 0 & -(C_i\gamma + \Gamma_{33}N_i)m_2t_i/2 & (C_i\gamma + \Gamma_{33}N_i)m_1t_i/2 \\ 0 & 0 & A_i & -(A_i\gamma + \Gamma_{13}N_i)n_2t_i/2 & (A_i\gamma + \Gamma_{13}N_i)n_1t_i/2 \\ 0 & 0 & B_i & -(B_i\gamma + \Gamma_{23}N_i)n_2t_i/2 & (B_i\gamma + \Gamma_{23}N_i)n_1t_i/2 \\ 0 & 0 & C_i & -(C_i\gamma + \Gamma_{33}N_i)n_2t_i/2 & (C_i\gamma + \Gamma_{33}N_i)n_1t_i/2 \end{bmatrix}$$



(a)

$$\begin{Bmatrix} x \\ y \\ z \end{Bmatrix} = \sum N_i \begin{Bmatrix} x_i \\ y_i \\ z_i \end{Bmatrix} + \sum \frac{B}{2} \eta N_i \begin{Bmatrix} x_{2i} \\ y_{2i} \\ z_{2i} \end{Bmatrix}$$

$$(V_{2i}) = B; \quad + \sum \frac{C}{2} \xi N_i \begin{Bmatrix} x_{3i} \\ y_{3i} \\ z_{3i} \end{Bmatrix}$$

$$\begin{cases} V_{2i} \\ V_{3i} \end{cases} = C$$

(b)

Consider e.g. motion of corner $\eta = \xi = +1$,
due to rotations, it is:

$$\beta: C/2 - \gamma: B/2 \quad \text{in } \xi \text{ direction}$$

$$-\alpha: C/2 \quad \text{in } \eta \text{ direction}$$

$$\alpha: B/2 \quad \text{in } \eta \text{ direction}$$

In this way, and resolving into xyz components, for an arbitrary point,

$$\begin{Bmatrix} u \\ v \\ w \end{Bmatrix} = \sum N_i \begin{Bmatrix} u_i \\ v_i \\ w_i \end{Bmatrix}$$

$$\begin{aligned} & \left[\begin{array}{ccc} x_{3i} & 0 & -x_{1i} \\ m_{3i} & 0 & -m_{1i} \\ n_{3i} & 0 & -n_{1i} \end{array} \right] \begin{Bmatrix} \beta_i \\ \gamma_i \\ \alpha_i \end{Bmatrix} \\ & + \sum N_i \xi \frac{C}{2} \begin{Bmatrix} -x_{2i} & l_{ii} & 0 \\ -m_{2i} & m_{ii} & 0 \\ -n_{2i} & n_{ii} & 0 \end{Bmatrix} \begin{Bmatrix} \beta_i \\ \gamma_i \\ \alpha_i \end{Bmatrix} \end{aligned}$$