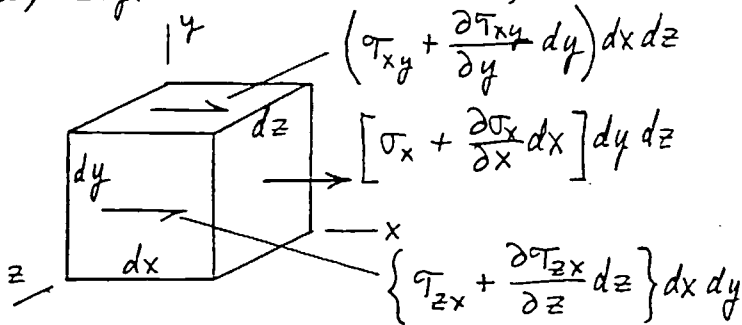


3.1-1

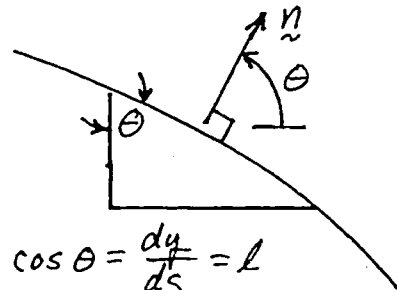
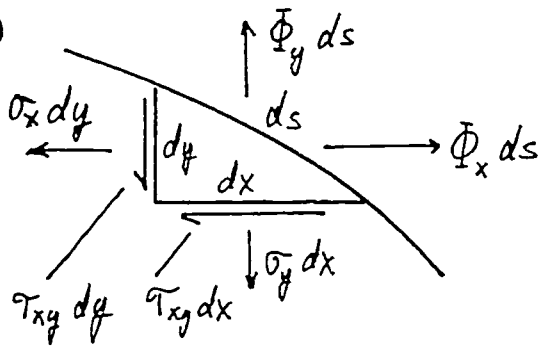
(a) E.g. in the x direction,



$$\sum F_x \text{ acts} = 0 = -\sigma_x dy dz - \tau_{xy} dx dz - \tau_{zx} dx dy + \left(\sigma_x + \frac{\partial \sigma_x}{\partial x} dx \right) dy dz + \left\{ \tau_{zx} + \frac{\partial \tau_{zx}}{\partial z} dz \right\} dx dy + \left\{ \tau_{xy} + \frac{\partial \tau_{xy}}{\partial y} dy \right\} dx dz + \left[\dots \right] dy dz + \left\{ \dots \right\} dx dy + F_x dx dy dz \quad \text{yields}$$

$$\frac{\partial \sigma_x}{\partial x} + \frac{\partial \tau_{xy}}{\partial y} + \frac{\partial \tau_{zx}}{\partial z} + F_x = 0$$

(b)



$$\cos \theta = \frac{dy}{ds} = l$$

$$\sin \theta = \frac{dx}{ds} = m$$

Thickness immaterial; can take as unity.

$$\sum F_x = 0 = -\sigma_x dy - \tau_{xy} dx + \Phi_x ds$$

$$\sum F_y = 0 = -\sigma_y dx - \tau_{xy} dy + \Phi_y ds$$

$$\Phi_x = \sigma_x \frac{dy}{ds} + \tau_{xy} \frac{dx}{ds}$$

$$\Phi_y = \tau_{xy} \frac{dy}{ds} + \sigma_y \frac{dx}{ds}$$

$$\Phi_x = l \sigma_x + m \tau_{xy}$$

$$\Phi_y = l \tau_{xy} + m \sigma_y$$

3.1-2

(a)

$$\sigma_x = -6a_1 x^2, \sigma_y = 12a_1 x^2, \tau_{xy} = 12a_1 y^2$$

1-2

On 1-2: $l=1, m=0, x=b$

3	2
4	1

$$\Phi_x = \sigma_x = -6a_1 b^2$$

$$\Phi_y = \tau_{xy} = 12a_1 y^2$$

On 2-3: $l=0, m=1, y=b$

$$\Phi_x = \tau_{xy} = 12a_1 b^2, \Phi_y = \sigma_y = 12a_1 x^2$$

On 3-4: $l=-1, m=0, x=0$

$$\Phi_x = 0, \Phi_y = -\tau_{xy} = -12a_1 y^2$$

On 4-1: $l=0, m=-1, y=0$

$$\Phi_x = -\tau_{xy} = 0, \Phi_y = -\sigma_y = -12a_1 x^2$$

(b) Check equilibrium.

x-dir. $\sigma_{x,x} + \tau_{xy,y} = -12a_1 x + 24a_1 y \neq 0$

y-dir. $\tau_{xy,x} + \sigma_{y,y} = 0 + 0 = 0 \checkmark$

Equil. not satisfied; field not possible.

3.1-3

$$\sigma_x = 3a_1 x^2 y, \sigma_y = a_1 y^3, \tau_{xy} = -3a_1 x y^2$$

Equilibrium:

$$\sigma_{x,x} + \tau_{xy,y} = 6a_1 x y - 6a_1 x y = 0 \checkmark$$

$$\tau_{xy,x} + \sigma_{y,y} = -3a_1 y^2 + 3a_1 y^2 = 0 \checkmark$$

Check compatibility:

$$\epsilon_x = \frac{1}{E} (3a_1 x^2 y - \nu a_1 y^3)$$

$$\epsilon_y = \frac{1}{E} (a_1 y^3 - 3\nu a_1 x^2 y)$$

$$\gamma_{xy} = \frac{2(1+\nu)}{E} (-3a_1 x y^2)$$

$$\epsilon_{x,yy} + \epsilon_{y,xx} \stackrel{?}{=} \gamma_{xy,xy}$$

$$\frac{1}{E} (-6\nu a_1 y - 6\nu a_1 y) \stackrel{?}{=} \frac{2(1+\nu)}{E} (-6a_1 y)$$

$\nu \neq 1+\nu$ Not possible.

3.1-5

$$\epsilon_x = u_{,x} = a_2 + 2a_4 x + a_5 y$$

$$\epsilon_y = v_{,y} = a_9 + a_{11} x + 2a_{12} y$$

$$\gamma_{xy} = u_{,y} + v_{,x} = a_3 + a_5 x + 2a_6 y + a_8 + 2a_{10} x + a_{11} y$$

Let $A = \frac{E}{1-\nu^2}$. Then

$$\sigma_x = A(\epsilon_x + \nu\epsilon_y), \quad \sigma_y = A(\epsilon_y + \nu\epsilon_x), \quad \tau_{xy} = G\gamma_{xy}$$

Equilibrium eqs. become

$$\sigma_{x,x} + \tau_{xy,y} = A(2a_4 + \nu a_{11}) + G(2a_6 + a_{11})$$

$$\tau_{xy,x} + \sigma_{y,y} = G(a_5 + 2a_{10}) + A(2a_{12} + \nu a_5)$$

When are these 2 eqs. satisfied?

With $G = (1-\nu)A/2$,

$$2a_4 + \nu a_{11} + \frac{1-\nu}{2}(2a_6 + a_{11}) = 0$$

$$\frac{1-\nu}{2}(a_5 + 2a_{10}) + 2a_{12} + \nu a_5 = 0$$

or $2a_4 + (1-\nu)a_6 + \frac{1+\nu}{2}a_{11} = 0$

$$2a_{12} + (1-\nu)a_{10} + \frac{1+\nu}{2}a_5 = 0$$

3.1-4

(a) $\Delta V = V_{\text{final}} - V_{\text{original}}$

$$\Delta V = [(1+\epsilon_x)dx(1+\epsilon_y)dy(1+\epsilon_z)dz] - [dx dy dz]$$

$$\Delta V = (\epsilon_x + \epsilon_y + \epsilon_z)dx dy dz + \text{higher order terms}$$

$$\Delta V = (\epsilon_x + \epsilon_y + \epsilon_z)dx dy dz \quad \text{for small strains}$$

$$\frac{\Delta V}{V} = \frac{\Delta V}{dx dy dz} = \epsilon_x + \epsilon_y + \epsilon_z$$

(b) Use 3D form of Eqs. 3.1-4, with $\sigma_x = \sigma_y = \sigma_z = -p$:

$$\epsilon_x = \frac{-p}{E}(1-2\nu) \quad \epsilon_y = \epsilon_x \quad \epsilon_z = \epsilon_x$$

$$\text{Hence } \frac{\Delta V}{V} = -\frac{3p}{E}(1-2\nu) \quad \text{and} \quad \frac{\Delta V/V}{p} = -\frac{3(1-2\nu)}{E}$$

(c) As $\nu \rightarrow 0.5$, $\frac{\Delta V}{V} \rightarrow 0$

3.1-4

(a) $\Delta V = V_{final} - V_{original}$

$$\Delta V = [(1 + \epsilon_x)dx (1 + \epsilon_y)dy (1 + \epsilon_z)dz] - [dx dy dz]$$

$$\Delta V = (\epsilon_x + \epsilon_y + \epsilon_z) dx dy dz + \text{higher order terms}$$

$$\Delta V = (\epsilon_x + \epsilon_y + \epsilon_z) dx dy dz \quad \text{for small strains}$$

$$\frac{\Delta V}{V} = \frac{\Delta V}{dx dy dz} = \epsilon_x + \epsilon_y + \epsilon_z$$

(b) Use 3D form of Eqs. 3.1-4, with $\sigma_x = \sigma_y = \sigma_z = -p$:

$$\epsilon_x = \frac{-p}{E} (1 - 2\nu) \quad \epsilon_y = \epsilon_x \quad \epsilon_z = \epsilon_x$$

$$\text{Hence } \frac{\Delta V}{V} = -\frac{3p}{E} (1 - 2\nu) \quad \text{and} \quad \frac{\Delta V/V}{p} = -\frac{3(1 - 2\nu)}{E}$$

(c) As $\nu \rightarrow 0.5$, $\frac{\Delta V}{V} \rightarrow 0$

3.2-1

$$N_1 = \frac{(2-x)(3-x)}{(2)(3)} = \frac{1}{6} (6 - 5x + x^2)$$

$$N_2 = \frac{-x(3-x)}{(-2)(1)} = \frac{1}{2} (3x - x^2)$$

$$N_3 = \frac{-x(2-x)}{(-3)(-1)} = \frac{1}{3} (-2x + x^2)$$

(a) $\sum_1^3 N_i = \frac{1}{6} (6 - 5x + x^2 + 9x - 3x^2 - 4x + 2x^2) = 1$

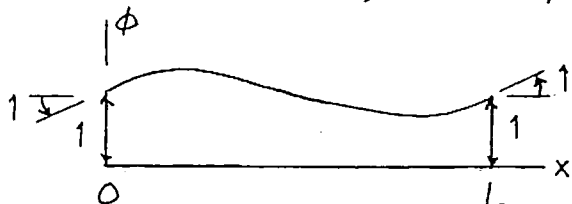
(b) $N_{1,x} = \frac{1}{6} (-5 + 2x) \quad \sum_1^3 N_{i,x} = \frac{1}{6} (-5 + 2x$

$$N_{2,x} = \frac{1}{2} (3 - 2x) \quad + 9 - 6x - 4 + 4x)$$

$$N_{3,x} = \frac{1}{3} (-2 + 2x) \quad = 0$$

3.2-2

If all 4 $\phi_i = 1$, the shape is



Clearly $\phi \neq 1$ throughout ($\sum N_i = 1$ if

$\sum N_i \phi_i = \phi$ gives $1 = 1$ for $\phi = \phi_i = 1$).

Also, not all N_i have the same units.

3.1-5

$$\epsilon_x = u_{,x} = a_2 + 2a_4 x + a_5 y$$

$$\epsilon_y = v_{,y} = a_9 + a_{11} x + 2a_{12} y$$

$$\gamma_{xy} = u_{,y} + v_{,x} = a_3 + a_5 x + 2a_6 y + a_8 + 2a_{10} x + a_{11} y$$

Let $A = \frac{E}{1-\nu^2}$. Then

$$\sigma_x = A(\epsilon_x + \nu\epsilon_y), \quad \sigma_y = A(\epsilon_y + \nu\epsilon_x), \quad \tau_{xy} = G\gamma_{xy}$$

Equilibrium eqs. become

$$\sigma_{x,x} + \tau_{xy,y} = A(2a_4 + \nu a_{11}) + G(2a_6 + a_{11})$$

$$\tau_{xy,x} + \sigma_{y,y} = G(a_5 + 2a_{10}) + A(2a_{12} + \nu a_5)$$

When are these 2 eqs. satisfied?

With $G = (1-\nu)A/2$,

$$2a_4 + \nu a_{11} + \frac{1-\nu}{2}(2a_6 + a_{11}) = 0$$

$$\frac{1-\nu}{2}(a_5 + 2a_{10}) + 2a_{12} + \nu a_5 = 0$$

or $2a_4 + (1-\nu)a_6 + \frac{1+\nu}{2}a_{11} = 0$

$$2a_{12} + (1-\nu)a_{10} + \frac{1+\nu}{2}a_5 = 0$$

3.2-3

(a) $\phi = \sum N_i \phi_i$; use Eqs. 3.2-7

$$\begin{aligned} \phi &= \frac{(3-x)(5-x)(8-x)}{(2)(4)(7)} 2 + \frac{(1-x)(5-x)(8-x)}{(-2)(2)(5)} 2 \\ &+ \frac{(1-x)(3-x)(8-x)}{(-4)(-2)(3)} 2 + \frac{(1-x)(3-x)(5-x)}{(-7)(-5)(-3)} 5 \end{aligned}$$

$$\begin{aligned} (b) \phi_0 &= \frac{3 \cdot 5 \cdot 8}{56} 2 + \frac{1 \cdot 5 \cdot 8}{-20} 2 + \frac{1 \cdot 3 \cdot 8}{24} 2 + \frac{1 \cdot 3 \cdot 5}{-105} 5 \\ &= \frac{120}{28} - \frac{40}{10} + \frac{24}{12} - \frac{15}{21} = 1.57 \end{aligned}$$

Similarly

$$\phi_2 = \frac{1 \cdot 3 \cdot 6}{28} - \frac{(-1) \cdot 3 \cdot 6}{10} + \frac{(-1) \cdot 1 \cdot 6}{12} - \frac{(-1) \cdot 1 \cdot 3}{21} = 2.086$$

$$\phi_4 = \frac{(-1) \cdot 1 \cdot 4}{28} - \frac{(-3) \cdot 1 \cdot 4}{10} + \frac{(-3) \cdot (-1) \cdot 4}{12} - \frac{(-3) \cdot (-1) \cdot 1}{21} = 1.914$$

$$\phi_7 = \frac{(-4) \cdot (-2) \cdot 1}{28} - \frac{(-6) \cdot (-2) \cdot 1}{10} + \frac{(-6) \cdot (-4) \cdot 1}{12} - \frac{(-6) \cdot (-4) \cdot (-2)}{21} = 3.371$$

3.2-4

First two of Eqs. 3.2-8 yield $a_1 = \phi_1$, $a_2 = \phi_{,x1}$

The second two equations become

$$\begin{bmatrix} L^2 & L^3 \\ 2L & 3L^2 \end{bmatrix} \begin{Bmatrix} a_3 \\ a_4 \end{Bmatrix} = \begin{Bmatrix} \phi_2 - \phi_1 - L\phi_{,x1} \\ \phi_{,x2} - \phi_{,x1} \end{Bmatrix},$$

$$\begin{Bmatrix} a_3 \\ a_4 \end{Bmatrix} = \frac{1}{L^4} \begin{bmatrix} 3L^2 & -L^3 \\ -2L & L^2 \end{bmatrix} \begin{Bmatrix} \phi_2 - \phi_1 - L\phi_{,x1} \\ \phi_{,x2} - \phi_{,x1} \end{Bmatrix} = \begin{Bmatrix} (-3\phi_1 - 2L\phi_{,x1} + 3\phi_2 - L\phi_{,x2})/L^2 \\ (2\phi_1 + L\phi_{,x1} - 2\phi_2 + L\phi_{,x2})/L^3 \end{Bmatrix}$$

$$[\tilde{A}]^{-1} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ -3/L^2 & -2/L & 3/L^2 & -1/L \\ 2/L^3 & 1/L^2 & -2/L^3 & 1/L^2 \end{bmatrix}$$

with $[\tilde{X}] = [1 \ x \ x^2 \ x^3]$,

$$[\tilde{X}][\tilde{A}]^{-1} = \left[1 - \frac{3x^2}{L^2} + \frac{2x^3}{L^3}, \quad x - \frac{2x^2}{L} + \frac{x^3}{L^2}, \quad \frac{3x^2}{L^2} - \frac{2x^3}{L^3}, \quad -\frac{x^2}{L} + \frac{x^3}{L^2} \right]$$

N_1
 N_2
 N_3
 N_4

3.2-5

$$\phi = [X] \{a\} = [1 \ x \ x^2] \begin{Bmatrix} a_1 \\ a_2 \\ a_3 \end{Bmatrix}$$

$$\phi_{,x} = [0 \ 1 \ 2x] \{a\}$$

Evaluate given conditions (let $\theta_1 = \phi_{,x1}$)

$$\begin{Bmatrix} \phi_1 \\ \theta_1 \\ \phi_2 \end{Bmatrix} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 1 & L & L^2 \end{bmatrix} \begin{Bmatrix} a_1 \\ a_2 \\ a_3 \end{Bmatrix} \quad \begin{array}{l} a_1 = \phi_1, a_2 = \theta_1 \\ \text{3rd eq. is then} \\ \phi_2 - \phi_1 - L\theta_1 = L^2 a_3 \end{array}$$

$$\begin{Bmatrix} a_1 \\ a_2 \\ a_3 \end{Bmatrix} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ -\frac{1}{L^2} & -\frac{1}{L} & \frac{1}{L^2} \end{bmatrix} \begin{Bmatrix} \phi_1 \\ \theta_1 \\ \phi_2 \end{Bmatrix} = [A]^{-1} \{d\}$$

$\therefore a_3 = \frac{\phi_2 - \phi_1}{L^2} - \frac{\theta_1}{L}$

$$\phi = [X] [A]^{-1} \{d\} = [N] \{d\} \text{ where}$$

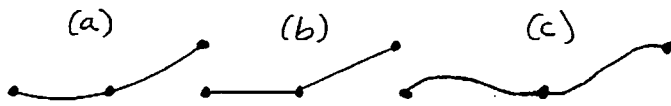
$$[N] = \begin{bmatrix} 1 - \frac{x^2}{L^2} & x(1 - \frac{x}{L}) & \frac{x^2}{L^2} \end{bmatrix} \cdot \text{Also}$$

$$[N_{,x}] = \begin{bmatrix} -\frac{2x}{L^2} & 1 - \frac{2x}{L} & \frac{2x}{L^2} \end{bmatrix}$$

i	At x=0		At x=L		Diagram
	N_i	$N_{i,x}$	N_i	$N_{i,x}$	
1	1	0	0	-2/L	
2	0	1	0	-1	
3	0	0	1	2/L	

3.2-6

Lagrange: 3 pts. define parabola.
Piecewise C^0 : two straight lines.
Piecewise C^1 : two cubics.



3.3-1

$$(a) u = [1 \ x] \begin{Bmatrix} a_1 \\ a_2 \end{Bmatrix}, [B_a] = \frac{\partial}{\partial x} [1 \ x] = [0 \ 1]$$

$$[k_a] = \int_0^L [B_a]^T [B_a] AE dx = AEL \begin{bmatrix} 0 & 0 \\ 0 & 1 \end{bmatrix}$$

Let $x_1=0, x_2=L$ in Eq. 3.2-4. Then $[A] = \begin{bmatrix} 1 & 0 \\ 1 & L \end{bmatrix}, [A]^{-1} = \begin{bmatrix} 1 & 0 \\ -1/L & 1/L \end{bmatrix}$

$$[k] = [A]^{-T} [k_a] [A]^{-1} = \begin{bmatrix} 1 & -1/L \\ 0 & 1/L \end{bmatrix} \left(AEL \begin{bmatrix} 0 & 0 \\ -1/L & 1/L \end{bmatrix} \right) = \frac{AE}{L} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \checkmark$$

$$(b) v = [1 \ x \ x^2 \ x^3] \begin{Bmatrix} a_1 \\ a_2 \\ a_3 \\ a_4 \end{Bmatrix}, \frac{d^2 v}{dx^2} = [0 \ 0 \ 2 \ 6x] \begin{Bmatrix} a_1 \\ a_2 \\ a_3 \\ a_4 \end{Bmatrix} = [B_a] \begin{Bmatrix} a_1 \\ a_2 \\ a_3 \\ a_4 \end{Bmatrix}$$

$$[k_a] = \int_0^L [B_a]^T [B_a] EI dx = EI \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 4L & 6L^2 \\ 0 & 0 & 6L^2 & 12L^3 \end{bmatrix}$$

$[A]^{-1}$ is calculated in Problem 3.2-4

$$[k] = [A]^{-1} [k_a] [A]^{-1} = \begin{bmatrix} 1 & 0 & -3/L^2 & 2/L^3 \\ 0 & 1 & -2/L & 1/L^2 \\ 0 & 0 & 3/L^2 & -2/L^3 \\ 0 & 0 & -1/L & 1/L^2 \end{bmatrix} \left(EI \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & -2 & 0 & 2 \\ 6 & 0 & -6 & 6L \end{bmatrix} \right)$$

$$[k] = EI \begin{bmatrix} 12/L^3 & 6/L^2 & -12/L^3 & 6/L^2 \\ 6/L^2 & 4/L & -6/L^2 & 2/L \\ -12/L^3 & -6/L^2 & 12/L^3 & -6/L^2 \\ 6/L^2 & 2/L & -6/L^2 & 4/L \end{bmatrix} \checkmark$$

3.3-2

(a)

Follow the method of Prob. 3.3-1. That is, use the "a-basis", $u = [1 \ x \ x^2 \ x^3] \{a\}$,

$$[k_a] = \int_0^L [B_a]^T [B_a] AE dx, \text{ where } [B_a] =$$

$$[0 \ 1 \ 2x \ 3x^2], [k] = [A]^{-T} [k_a] [A]^{-1},$$

and $[A]^{-1}$ is computed in Prob. 3.2-4.

$$AE \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & L & L^2 & L^3 \\ 0 & L^2 & \frac{4}{3}L^3 & \frac{3}{2}L^4 \\ 0 & L^3 & \frac{3}{2}L^4 & \frac{9}{5}L^5 \end{bmatrix}, AE \begin{bmatrix} 0 & 0 & 0 & 0 \\ -1 & 0 & 1 & 0 \\ -L & -\frac{L^2}{6} & L & \frac{L^2}{6} \\ -9L^2 & -2L^3 & 9L^2 & 3L^3 \end{bmatrix}$$

$[k_a] \qquad [k_a][A]^{-1}$

$$\begin{bmatrix} 1 & 0 & -3/L^2 & 2/L^3 \\ 0 & 1 & -2/L & 1/L^2 \\ 0 & 0 & 3/L^2 & -2/L^3 \\ 0 & 0 & -1/L & 1/L^2 \end{bmatrix}, \frac{AE}{30L} \begin{bmatrix} 36 & 3L & -36 & 3L \\ 3L & 4L^2 & -3L & -L^2 \\ -36 & -3L & 36 & -3L \\ 3L & -L^2 & -3L & 4L^2 \end{bmatrix}$$

$[A]^{-T} \qquad [k]$

(b) Must not enforce interelement continuity of ϵ_x at these locations. Where two elements meet, can condense the ϵ_x d.o.f. in one of them before assembly.

$$\boxed{3.3-3} \quad \text{Eqs. 3.2-7: } x_1 = 0, x_2 = \frac{L}{2}, x_3 = L$$

$$N_1 = \frac{\left(\frac{L}{2} - x\right)(L - x)}{\frac{L}{2}L} = \frac{1}{L^2}(L^2 - 3Lx + 2x^2)$$

$$N_2 = \frac{(-x)(L - x)}{-\frac{L}{2}\left(\frac{L}{2}\right)} = \frac{4}{L^2}(Lx - x^2)$$

$$N_3 = \frac{-x\left(\frac{L}{2} - x\right)}{-L\left(-\frac{L}{2}\right)} = \frac{1}{L^2}(2x^2 - Lx)$$

$$[B] = \frac{d}{dx} [N] = \frac{1}{L^2} \begin{bmatrix} -3L + 4x & 4(L - 2x) & 4x - L \end{bmatrix}$$

$$[k] = \int_0^L [B]^T [B] AE dx$$

$$[k] = \frac{AE}{L^4} \int_0^L \begin{bmatrix} 9L - 24Lx + 16x^2 & -12L^2 + 40Lx - 32x^2 & 3L^2 - 16Lx + 16x^2 \\ \text{symmetric} & 16(L^2 - 4Lx + 4x^2) & 4(-L^2 + 6Lx - 8x^2) \\ & & L^2 - 8Lx + 16x^2 \end{bmatrix} dx$$

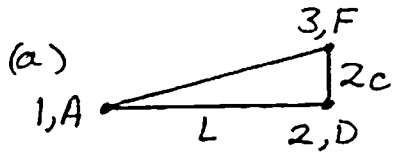
$$[k] = \frac{AE}{L^4} \int_0^L \begin{bmatrix} 9L^2x - 12Lx^2 + \frac{16}{3}x^3 & -12L^2x + 20Lx^2 - \frac{32}{3}x^3 & 3L^2x - 8Lx^2 + \frac{16}{3}x^3 \\ \text{symmetric} & 16(L^2x - 2Lx^2 + \frac{4}{3}x^3) & 4(-L^2x + 3Lx^2 - \frac{8}{3}x^3) \\ & & L^2x - 4Lx^2 + \frac{16}{3}x^3 \end{bmatrix} dx$$

$$[k] = \frac{AE}{L^4} \begin{bmatrix} 7 & -8 & 1 \\ -8 & 16 & -8 \\ 1 & -8 & 7 \end{bmatrix}$$

3.4-1 Beam theory: $I = \frac{t}{12}(2c)^3 = \frac{2tc^3}{3}$

$$V_D = V_F = -\frac{ML^2}{2EI} = -\frac{3ML^2}{4Etc^3}$$

$$u_F = -u_D = c \frac{ML}{EI} = \frac{3ML}{2Etc^2}$$



Obtain $[B]$ from Eq. 3.4-10

$$\{\underline{\epsilon}\} = [B]\{\underline{d}\} = \begin{bmatrix} -1/L & 0 & 1/L & 0 & 0 & 0 \\ 0 & 0 & 0 & -1/2c & 0 & 1/2c \\ 0 & -1/L & -1/2c & 1/L & 1/2c & 0 \end{bmatrix} \begin{Bmatrix} 0 \\ 0 \\ u_D \\ v_D \\ u_F \\ v_F \end{Bmatrix}$$

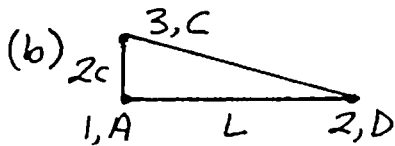
$$\epsilon_x = \frac{u_D}{L} = -\frac{3M}{2Etc^2}, \quad \epsilon_y = \frac{v_F - v_D}{2c} = 0,$$

$$\gamma_{xy} = -\frac{u_D}{2c} + \frac{v_D}{L} + \frac{u_F}{2c} = \frac{u_F}{c} + \frac{v_D}{L} = \frac{3ML}{4Etc^3}$$

With $\nu = 0$, $\sigma_x = E\epsilon_x = \frac{3M}{2tc^2}$, $\sigma_y = E\epsilon_y = 0$, $\tau_{xy} = \frac{E}{2}\gamma_{xy} = \frac{3ML}{4tc^3}$

Beam theory, top & bottom: $|\sigma_x| = \frac{Mc}{8tc^3/12} = \frac{3M}{2tc^2}$

With σ_x & τ_{xy} from FEA, $\frac{\tau_{xy}}{\sigma_x} = \frac{L}{2c} \rightarrow \infty$ as $\frac{L}{c} \rightarrow \infty$



Obtain $[B]$ from Eq. 3.4-10

$$\{\underline{\epsilon}\} = [B]\{\underline{d}\} = \begin{bmatrix} -1/L & 0 & 1/L & 0 & 0 & 0 \\ 0 & -1/2c & 0 & 0 & 0 & 1/2c \\ -1/2c & -1/L & 0 & 1/L & 1/2c & 0 \end{bmatrix} \begin{Bmatrix} 0 \\ 0 \\ u_D \\ v_D \\ 0 \\ 0 \end{Bmatrix}$$

$$\epsilon_x = \frac{v_D}{L} = -\frac{3M}{2Etc^2}, \quad \epsilon_y = 0, \quad \gamma_{xy} = \frac{v_D}{L} = -\frac{3ML}{4Etc^3}$$

With $\nu = 0$, $\sigma_x = E\epsilon_x = -\frac{3M}{2tc^2}$, $\sigma_y = E\epsilon_y = 0$

$$\tau_{xy} = \frac{E}{2}\gamma_{xy} = -\frac{3ML}{8tc^3}$$

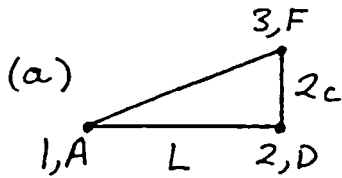
Beam theory: as in part (a)

With σ_x & τ_{xy} from FEA, $\frac{\tau_{xy}}{\sigma_x} = \frac{L}{4c} \rightarrow \infty$ as $\frac{L}{c} \rightarrow \infty$

3.4-2 Beam theory: $I = \frac{t}{12} (2c)^3 = \frac{2tc^3}{3}$

$$v_D = v_F = \frac{PL^3}{3EI} = \frac{PL^3}{2Etc^3}$$

$$u_F = -u_D = -c \frac{PL^2}{2EI} = -\frac{3PL^2}{4Etc^2}$$



Obtain $[B]$ from Eq. 3.4-10

$$\{\underline{\epsilon}\} = [\underline{B}]\{\underline{d}\} = \begin{bmatrix} -1/L & 0 & 1/L & 0 & 0 & 0 \\ 0 & 0 & 0 & -1/2c & 0 & 1/2c \\ 0 & -1/L & -1/2c & 1/L & 1/2c & 0 \end{bmatrix} \begin{Bmatrix} 0 \\ 0 \\ u_D \\ v_D \\ u_F \\ v_F \end{Bmatrix}$$

$$\epsilon_x = \frac{u_D}{L} = \frac{3PL}{4Etc^2}, \quad \epsilon_y = \frac{v_F - v_D}{2c} = 0,$$

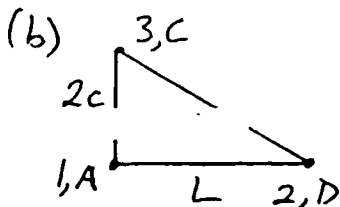
$$\gamma_{xy} = -\frac{u_D}{2c} + \frac{v_D}{L} + \frac{u_F}{2c} = \frac{u_F}{c} + \frac{v_D}{L} = -\frac{PL^2}{4Etc^3}$$

$$\text{With } \nu = 0, \quad \sigma_x = E\epsilon_x = \frac{3PL}{4tc^2}, \quad \sigma_y = E\epsilon_y = 0, \quad \tau_{xy} = \frac{E}{2}\gamma_{xy} = -\frac{PL^2}{8tc^3}$$

$$\text{Beam theory, top \& bottom: } |\sigma_x| = \frac{P(L-x)c}{I} = \frac{3P(L-x)}{2tc^2}$$

$$\text{Beam theory, neutral axis: } \tau_{xy} = \frac{3}{2} \frac{P}{2tc} = \frac{3P}{4tc}$$

$$\text{With } \sigma_x \text{ and } \tau_{xy} \text{ from FEA, } \frac{\tau_{xy}}{\sigma_x} = -\frac{L}{6c} \rightarrow -\infty \text{ as } \frac{L}{c} \rightarrow \infty$$



Obtain $[B]$ from Eq. 3.4-10

$$\{\underline{\epsilon}\} = [\underline{B}]\{\underline{d}\} = \begin{bmatrix} -1/L & 0 & 1/L & 0 & 0 & 0 \\ 0 & -1/2c & 0 & 0 & 0 & 1/2c \\ -1/2c & -1/L & 0 & 1/L & 1/2c & 0 \end{bmatrix} \begin{Bmatrix} 0 \\ 0 \\ u_D \\ v_D \\ 0 \\ 0 \end{Bmatrix}$$

$$\epsilon_x = \frac{u_D}{L} = \frac{3PL}{4Etc^2}, \quad \epsilon_y = 0, \quad \gamma_{xy} = \frac{v_D}{L} = \frac{PL^2}{2Etc^3}$$

$$\text{With } \nu = 0, \quad \sigma_x = E\epsilon_x = \frac{3PL}{4tc^2}, \quad \sigma_y = E\epsilon_y = 0, \quad \tau_{xy} = \frac{E}{2}\gamma_{xy} = \frac{PL^2}{4tc^3}$$

Beam theory: as in part (a)

$$\text{With } \sigma_x \text{ and } \tau_{xy} \text{ from FEA, } \frac{\tau_{xy}}{\sigma_x} = \frac{L}{3c} \rightarrow \infty \text{ as } \frac{L}{c} \rightarrow \infty$$

3.4-3

(a) The only nonzero d.o.f. are u_3 and v_3 . From Eq. 3.4-10,

$$\begin{Bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \end{Bmatrix} = [B] \{d\} = \begin{bmatrix} 0 & 0 \\ 0 & 1/a \\ 1/a & 0 \end{bmatrix} \begin{Bmatrix} u_3 \\ v_3 \end{Bmatrix} \quad \text{For } v=0,$$

$$[k] = Et \frac{a^2}{2} \begin{bmatrix} 0 & 0 & 1/a \\ 0 & 1/a & 0 \end{bmatrix} \begin{bmatrix} 1 & & \\ & 1 & \\ & & 1/2 \end{bmatrix} \begin{bmatrix} 0 & 0 \\ 0 & 1 \\ 1 & 0 \end{bmatrix}$$

$$[k] = \frac{Et}{2} \begin{bmatrix} 0 & 0 & 1 \\ 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} 0 & 0 \\ 0 & 1 \\ 1/2 & 0 \end{bmatrix} = \frac{Et}{2} \begin{bmatrix} 1/2 & 0 \\ 0 & 1 \end{bmatrix}$$

(b) 4 triangles each resist P with ϵ_y stiffness $Et/4$
4 triangles each resist P with γ_{xy} stiffness $Et/2$

Net stiffness = $4\left(\frac{Et}{4} + \frac{Et}{2}\right) = 3Et$, so $v = \frac{P}{3Et}$
at center node (independent of a).

3.4-4 With u_1 and v_1 the only nonzero d.o.f., $\epsilon_x = 0$, $\epsilon_y = \frac{v_1}{b}$, $\gamma_{xy} = \frac{u_1}{b}$

$$\{\underline{\epsilon}\} = [\underline{B}]\{\underline{d}\} = \frac{1}{b} \begin{bmatrix} 0 & 0 \\ 0 & 1 \\ 1 & 0 \end{bmatrix} \begin{Bmatrix} u_1 \\ v_1 \end{Bmatrix} \quad \text{For } \nu = 0,$$

$$[\underline{k}] = E(abt) \frac{1}{b^2} \begin{bmatrix} 0 & 0 & 1 \\ 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} 1 & & \\ & 1 & \\ & & 1/2 \end{bmatrix} \begin{bmatrix} 0 & 0 \\ 0 & 1 \\ 1 & 0 \end{bmatrix}$$

$$[\underline{k}] = \frac{Eat}{b} \begin{bmatrix} 0 & 0 & 1 \\ 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} 0 & 0 \\ 0 & 1 \\ 1/2 & 0 \end{bmatrix} = \frac{Eat}{b} \begin{bmatrix} 1/2 & 0 \\ 0 & 1 \end{bmatrix}$$

3.5-1

(a) Let $\phi = u$ or v . ϕ depends only on y . Assume

$$\phi = c_1 + c_2 y + c_3 y^2 \quad (a)$$

Substitute:

$$\phi = 0 \text{ at } y = 0, \therefore c_1 = 0$$

$$\phi = 0 \text{ at } y = b, \quad 0 = c_2 b + c_3 b^2$$

$$\phi = 1 \text{ at } y = 2b, \quad 1 = c_2(2b) + c_3(4b^2)$$

$$\text{Solve: } c_2 = -\frac{1}{2b}, \quad c_3 = \frac{1}{2b^2}$$

$$\text{Eq. (a) becomes } \phi = \underbrace{-\frac{y}{2b} + \frac{y^2}{2b^2}}_{N_3}$$

$$(b) \text{ On } y = 0, N_4 = 1 - \left(\frac{x}{a}\right)^2$$

Zero at $x = \pm a$, unity at $x = 0$

$$\text{On } y = b, N_4 = -\left(\frac{x}{a}\right)^2 + \frac{1}{4}$$

Zero at $x = \pm(a/2)$

$$\text{At } y = 2b, N_4 = 1 - 2 + 1 = 0$$

(c) Here $u = N_3 u_3 + N_4 u_4$, $v = N_3 v_3 + N_4 v_4$

$$\epsilon_x = \frac{\partial N_3}{\partial x} u_3 + \frac{\partial N_4}{\partial x} u_4 = -\frac{2x}{a^2} u_4$$

$$\epsilon_y = \frac{\partial N_3}{\partial y} v_3 + \frac{\partial N_4}{\partial y} v_4$$

$$= \left(-\frac{1}{2b} + \frac{y}{b^2}\right) v_3 + \left[-\frac{1}{b} + \frac{y}{2b^2}\right] v_4$$

$$\gamma_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} = (\dots) u_3 + [\dots] u_4 - \frac{2x}{a^2} v_4$$

3.6-1

$$\epsilon_x = a_2 + a_4 y, \quad \epsilon_y = a_7 + a_8 x$$

$$\gamma_{xy} = a_3 + a_4 x + a_6 + a_8 y$$

$$\sigma_{x,x} = \frac{E}{1-\nu^2} (\epsilon_x + \nu \epsilon_y)_{,x} = \frac{E}{1-\nu^2} (\nu a_8)$$

$$\sigma_{y,y} = \frac{E}{1-\nu^2} (\epsilon_y + \nu \epsilon_x)_{,y} = \frac{E}{1-\nu^2} (\nu a_4)$$

$$\tau_{xy} = \frac{E}{2(1+\nu)} \gamma_{xy} \quad \tau_{xy,x} = \frac{E}{2(1+\nu)} a_4$$

Subs. into $\tau_{xy,y} = \frac{E}{2(1+\nu)} a_8$

$$\left. \begin{aligned} \sigma_{x,x} + \tau_{xy,y} + F_x &= 0 \\ \sigma_{y,y} + \tau_{xy,x} + F_y &= 0 \end{aligned} \right\} \text{yields } \begin{cases} C E a_8 + F_x = 0 \\ C E a_4 + F_y = 0 \end{cases}$$

where $C = \frac{1}{1-\nu^2} + \frac{1}{2(1+\nu)} = \frac{1}{2(1-\nu)}$

Eqs. satisfied if $a_4 = a_8 = 0$ when $F_x = F_y = 0$
or if particular values for F_x & F_y const.

3.6-2

$$(a) \quad \begin{Bmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \\ u_4 \\ v_4 \end{Bmatrix} = \begin{Bmatrix} a_1 - a_2 a - a_3 b \\ a_4 - a_5 a - a_6 b \\ a_1 + a_2 a - a_3 b \\ a_4 + a_5 a - a_6 b \\ a_1 + a_2 a + a_3 b \\ a_4 + a_5 a + a_6 b \\ a_1 - a_2 a + a_3 b \\ a_4 - a_5 a + a_6 b \end{Bmatrix}$$

Eqs. 3.6-4 and 3.6-5 show that

$$[N] \{d\} = \begin{Bmatrix} a_1 + a_2 x + a_3 y \\ a_4 + a_5 x + a_6 y \end{Bmatrix}$$

$$(b) \quad \epsilon_x = \frac{1}{4ab} \begin{bmatrix} -(b-y) & 0 & b-y & 0 & b+y & 0 & -(b+y) & 0 \end{bmatrix} \{d\}$$

$$= a_2$$

$$\epsilon_y = \frac{1}{4ab} \begin{bmatrix} 0 & -(a-x) & 0 & -(a+x) & 0 & a+x & 0 & a-x \end{bmatrix} \{d\}$$

$$= a_6$$

$$\gamma_{xy} = \frac{1}{4ab} \begin{bmatrix} -(a-x) & -(b-y) & -(a+x) & b-y \\ a+x & b+y & a-x & -(b+y) \end{bmatrix} \{d\}$$

$$= a_3 + a_5$$

3.6-3 Beam theory will give $u_F = -u_D$, $v_F = v_D$

(a) From Eq. 3.6-6,

$$\begin{Bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \end{Bmatrix} = \frac{1}{4ab} \begin{bmatrix} \cdots & b-y & 0 & b+y & 0 & \cdots \\ \cdots & 0 & -(a+x) & 0 & a+x & \cdots \\ \cdots & -(a+x) & b-y & a+x & b+y & \cdots \end{bmatrix} \begin{Bmatrix} \vdots \\ u_D \\ v_D \\ -u_D \\ v_D \\ \vdots \end{Bmatrix}$$

From beam theory, Prob. 3.4-1, $u_D = -\frac{3ML}{2Etc^2}$, $v_D = -\frac{3ML^2}{4Etc^3}$

Hence

$$\epsilon_x = \frac{3MLy}{4Etabc^2}, \quad \epsilon_y = 0, \quad \gamma_{xy} = \frac{3ML(a+x)}{4abEtc^2} - \frac{3ML^2}{8aEtc^3}$$

For $v=0$, $\sigma_x = E\epsilon_x = \frac{3MLy}{4tabc^2}$, $\sigma_y = E\epsilon_y = 0$,

$$\tau_{xy} = \frac{E}{2} \gamma_{xy} = \frac{3ML(a+x)}{8abtc^2} - \frac{3ML^2}{16atc^3}$$

But $L=2a$ and $c=b$, so $\sigma_x = \frac{3My}{2tb^3}$, $\sigma_y = 0$, $\tau_{xy} = \frac{3Mx}{4b^3t}$

(in the coordinate system of Fig. 3.6-1). In this system, $\sigma_x = \frac{My}{I} = \frac{3My}{2tb^3}$, $\sigma_y = \tau_{xy} = 0$ according to beam theory.

(b) From beam theory, Prob. 3.4-2, $u_D = \frac{3PL^2}{4Etc^2}$, $v_D = \frac{PL^3}{2Etc^3}$

$$\text{Hence } \epsilon_x = -\frac{3PL^2y}{8abEtc^2}, \quad \epsilon_y = 0, \quad \gamma_{xy} = \frac{PL^3}{4aEtc^3} - \frac{3PL^2(a+x)}{8abEtc^2}$$

For $v=0$, $\sigma_x = E\epsilon_x = -\frac{3PL^2y}{8abtc^2}$, $\sigma_y = E\epsilon_y = 0$,

$$\tau_{xy} = \frac{E}{2} \gamma_{xy} = \frac{PL^3}{8atc^3} - \frac{3PL^2(a+x)}{16abtc^2}$$

But $L=2a$ and $c=b$, so $\sigma_x = \frac{3Pay}{2tb^3}$, $\sigma_y = 0$, $\tau_{xy} = \frac{Pa}{4tb^3}(a-3x)$

(in the coordinate system of Fig. 3.6-1). In this system,

$$\sigma_x = \frac{My}{I} = \frac{P(a-x)y}{t(2b)^3/12} = \frac{3P(a-x)y}{2tb^3}, \quad \sigma_y = 0, \quad \text{and}$$

$$\tau_{xy} = \frac{3}{2} \frac{P}{2tb} = \frac{3P}{4tb} \quad \text{according to beam theory.}$$

3.6-4

(a) $K_{48,39}$ is associated with v_{24} and u_{20} . Since node 24 is not in element j , el. j does not contribute to $K_{48,39}$.

(b) $K_{37,37} \leftarrow k_{1,1}$ of el. j . $[E]$ is diagonal (since $\nu = 0$), $[E] = 10^7 \begin{bmatrix} 1 & & \\ & 1 & \\ & & \frac{1}{2} \end{bmatrix}$. We need only col. 1 of $[B]$, Eq. 3.6-6

$$k_{1,1} = \frac{1}{16} \int_{-1}^1 \int_{-1}^1 \begin{Bmatrix} -(1-y) \\ 0 \\ -(1-x) \end{Bmatrix}^T \begin{Bmatrix} -(1-y) \\ 0 \\ -\frac{1}{2}(1-x) \end{Bmatrix} 10^7 (1) dx dy$$

After integration, $k_{1,1} = 10^7/2$.

(c) $K_{59,61} \leftarrow k_{3,5}$ of el. j . Need cols. 3 & 5 of $[B]$, Eq. 3.6-6. As in (b),

$$k_{3,5} = \frac{1}{16} \int_{-1}^1 \int_{-1}^1 \begin{Bmatrix} 1-y \\ 0 \\ -(1+x) \end{Bmatrix}^T \begin{Bmatrix} 1+y \\ 0 \\ \frac{1}{2}(1+x) \end{Bmatrix} 10^7 (1) dx dy$$

After integration, $k_{3,5} = 0$

3.6-5

Method 1: Evaluate $u = a_1 + a_2x + a_3y + a_4xy$ at nodes, solve for a 's, gather coefficients of u_1, u_2, u_3, u_4 .

$$u_1 = a_1 \quad \text{Node 1}$$

$$u_2 = u_1 + a_2(2a); \quad a_2 = \frac{u_2 - u_1}{2a} \quad \text{Node 2}$$

$$u_4 = u_1 + a_3(2b); \quad a_3 = \frac{u_4 - u_1}{2b} \quad \text{Node 4}$$

$$u_3 = u_1 + \frac{u_2 - u_1}{2a} 2a + \frac{u_4 - u_1}{2b} 2b + a_4(2a)(2b)$$

$$\text{from which } a_4 = \frac{u_1 - u_2 + u_3 - u_4}{4ab}$$

$$u = u_1 + \frac{u_2 - u_1}{2a} x + \frac{u_4 - u_1}{2b} y + \frac{u_1 - u_2 + u_3 - u_4}{4ab} xy$$

$$u = \left(1 - \frac{x}{2a} - \frac{y}{2b} + \frac{xy}{4ab}\right) u_1 + \left(\frac{x}{2a} - \frac{xy}{4ab}\right) u_2 + \left(\frac{xy}{4ab}\right) u_3 + \left(\frac{y}{2b} - \frac{xy}{4ab}\right) u_4$$

Coefficients of the u_i are the N_i .

Method 2: Take the product of one-dimensional N_i (as for a bar element).

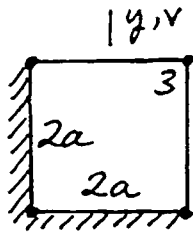
$$N_1 = \frac{2a-x}{2a} \frac{2b-y}{2b} \quad N_2 = \frac{x}{2a} \frac{2b-y}{2b}$$

$$N_3 = \frac{x}{2a} \frac{y}{2b} \quad N_4 = \frac{2a-x}{2a} \frac{y}{2b}$$

Method 3: In Eqs. 3.6-4, replace x by $x-a$ and y by $y-b$.

3.6-6

Consider the lower-left element of the four-element structure.



In the assembled structure, v_3 is the only nonzero d.o.f., so we need only this one stiffness coefficient.

$$\begin{Bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \end{Bmatrix} = [B] \{d\} = \frac{1}{4a^2} \begin{Bmatrix} 0 \\ a+x \\ a+y \end{Bmatrix} v_3 \quad (\text{from Eq. 3.6-6})$$

$$k = \frac{1}{16a^4} \int_{-a}^a \int_{-a}^a \begin{bmatrix} 0 & a+x & a+y \end{bmatrix} \begin{bmatrix} E \\ E \\ E/2 \end{bmatrix} \begin{Bmatrix} 0 \\ a+x \\ a+y \end{Bmatrix} t \, dx \, dy$$

$$k = \frac{Et}{16a^4} \int_{-a}^a \int_{-a}^a \left[(a+x)^2 + \frac{(a+y)^2}{2} \right] dx \, dy = \frac{Et}{16a^4} \left[\frac{16a^4}{3} + \frac{8a^4}{3} \right] = \frac{Et}{2}$$

For the four-element structure, $v_{\text{center}} = \frac{P/4}{Et/2} = \frac{P}{2Et}$

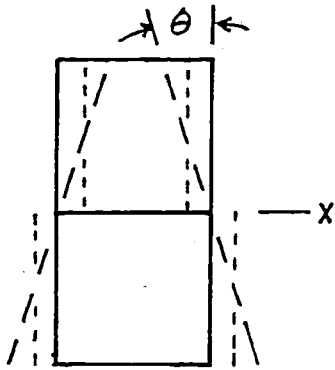
3.6-7

Assume that beam theory would provide almost the exact tip deflection, and recall Eq. 3.6-11:

$$\frac{\theta_{el}}{\theta_b} = \frac{1-\nu^2}{1 + \frac{1-\nu}{2} \left(\frac{a}{b}\right)^2}$$

(a) Expect the 2nd mesh to be better, since a/b is half as large in the 2nd mesh

(b) Two elements of the first mesh, with bending deformation:



Smaller dashed lines represent deformation due to ϵ_x in elements. Assume this is exact. Remainder comes from rotation of element faces and suffers from parasitic shear: with $\nu=0$,

$$\frac{\theta_{el}}{\theta_b} = \frac{1}{1 + \frac{1}{2}(1)} = \frac{2}{3}$$

Assume that γ_{xy} and ϵ_x contribute equally, therefore estimate error as the average:

$$e \approx \frac{1}{2} \left[0 + \left(1 - \frac{2}{3}\right) \right] \approx 17\%$$

For the second mesh, $\frac{a}{b} = \frac{1}{2}$, so with $\nu=0$

$$\frac{\theta_{el}}{\theta_b} = \frac{1}{1 + \frac{1}{2} \left(\frac{1}{2}\right)^2} = 0.89 \quad \text{i.e. } e \approx 11\%$$

3.7-1

$$\phi_{152} = \frac{-x(a-x)}{2a^2} \phi_1 + \frac{(a-x)(a+x)}{a^2} \phi_5 + \frac{x(a+x)}{2a^2} \phi_2$$

$$\phi_{43} = \frac{a-x}{2a} \phi_4 + \frac{a+x}{2a} \phi_3$$

$$\phi = \frac{b-y}{2b} \phi_{152} + \frac{b+y}{2b} \phi_{43} = \sum_{i=1}^5 N_i \phi_i$$

$$N_1 = \frac{-x(a-x)(b-y)}{4a^2b}, \quad N_2 = \frac{x(a+x)(b-y)}{4a^2b}$$

$$N_3 = \frac{(a+x)(b+y)}{4ab}, \quad N_4 = \frac{(a-x)(b+y)}{4ab}$$

$$N_5 = \frac{(a^2-x^2)(b-y)}{2a^2b}$$

3.7-2

$$(a) \quad \phi_{184} = \sum_{i=1}^3 \bar{N}_i * (\phi_i \text{ on } x=-a)$$

$$\phi_{597} = \sum_{i=1}^3 \bar{N}_i * (\phi_i \text{ on } x=0)$$

$$\phi_{263} = \sum_{i=1}^3 \bar{N}_i * (\phi_i \text{ on } x=+a)$$

for which $\bar{N}_1, \bar{N}_2, \bar{N}_3$ come from Eq.

3.2-7 with $x \rightarrow y, x_1 \rightarrow -b, x_2 = 0, x_3 \rightarrow b$

$$\bar{N}_1 = \frac{-y(b-y)}{2b^2}, \quad \bar{N}_2 = \frac{(-b-y)(b-y)}{-b(b)} = \frac{b^2-y^2}{b^2}$$

$$\bar{N}_3 = \frac{(-b-y)(-y)}{-2b(-b)} = \frac{y(b+y)}{2b^2}. \quad \text{In 9-node el.,}$$

$$\phi = \frac{-x(a-x)}{2a^2} \phi_{184} + \frac{a^2-x^2}{a^2} \phi_{597} + \frac{x(a+x)}{2a^2} \phi_{263}$$

$$\phi = N_a \phi_{184} + N_b \phi_{597} + N_c \phi_{263}$$

$$\phi = N_a (\bar{N}_1 \phi_1 + \bar{N}_2 \phi_8 + \bar{N}_3 \phi_4) + N_b (\bar{N}_1 \phi_5 + \bar{N}_2 \phi_9 + \bar{N}_3 \phi_7) + N_c (\bar{N}_1 \phi_2 + \bar{N}_2 \phi_6 + \bar{N}_3 \phi_3)$$

$$\phi = \sum_{i=1}^9 N_i \phi_i \quad \text{where}$$

$$N_1 = N_a \bar{N}_1 = xy(a-x)(b-y)/4a^2b^2$$

$$N_2 = N_c \bar{N}_1 = -xy(a+x)(b-y)/4a^2b^2$$

$$N_3 = N_c \bar{N}_3 = xy(a+x)(b+y)/4a^2b^2$$

$$N_4 = N_a \bar{N}_3 = -xy(a-x)(b+y)/4a^2b^2$$

$$N_5 = N_b \bar{N}_1 = -(a^2-x^2)y(b-y)/2a^2b^2$$

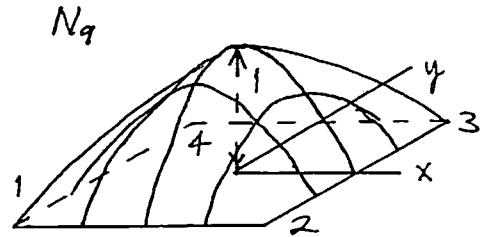
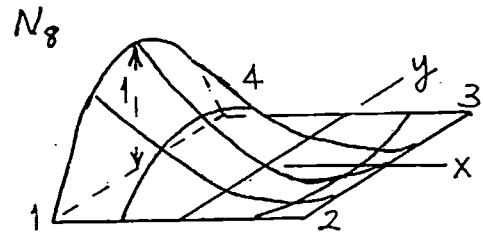
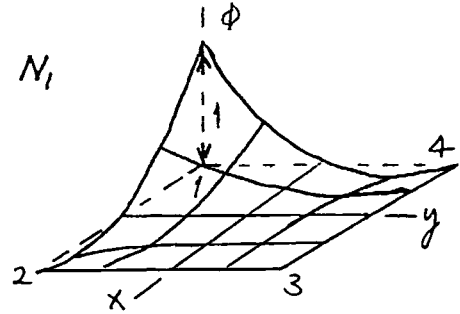
$$N_6 = N_c \bar{N}_2 = x(a+x)(b^2-y^2)/2a^2b^2$$

$$N_7 = N_b \bar{N}_3 = (a^2-x^2)y(b+y)/2a^2b^2$$

$$N_8 = N_a \bar{N}_2 = -x(a-x)(b^2-y^2)/2a^2b^2$$

$$N_9 = N_b \bar{N}_3 = (a^2-x^2)(b^2-y^2)/a^2b^2$$

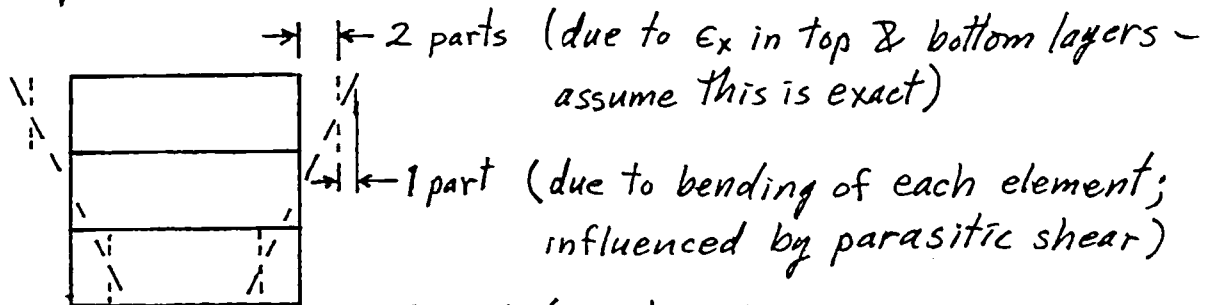
(b)



3.8-1 Similar to Problem 3.6-7. Again use Eq. 3.6-11 with $\nu = 0$ for simplicity. Best mesh to worst, with estimated error e :

(c) Aspect ratio $\frac{1}{3}$, $\frac{1}{1 + \frac{1}{2}(\frac{1}{3})^2} = 0.947$, $e \approx 1 - 0.947 = 5.3\%$.

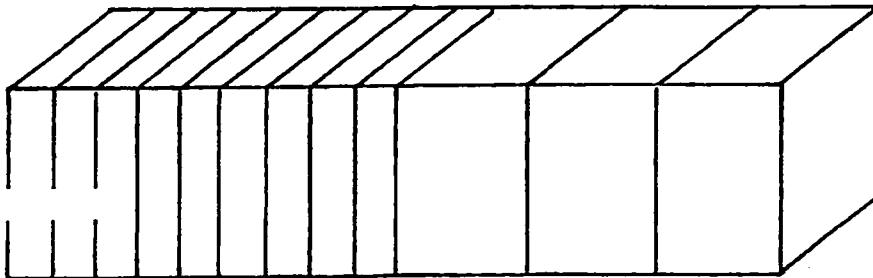
(b) Aspect ratio 3



$$\frac{2}{3} + \frac{1}{3} \left(\frac{1}{1 + \frac{1}{2}(3)^2} \right) = 0.727, e \approx 1 - 0.727 = 27\%$$

(c) Aspect ratio 1, $\frac{1}{1 + \frac{1}{2}(1)^2} = 0.667$, $e \approx 1 - 0.667 = 33\%$

A better 12-element mesh (for the given loading):



3.9-1

Rigid-body lateral translation \bar{u} : $\{\underline{d}\} = [\bar{u} \ 0 \ \bar{u} \ 0]^T$

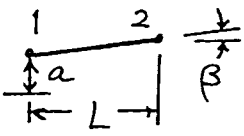
Rigid-body rotation through small angle θ about node 1:

$$\{\underline{d}\} = [0 \ \theta \ L\theta \ \theta]^T$$

In both cases, straightforward multiplication shows that

$[\underline{B}]\{\underline{d}\}$ and $[\underline{k}]\{\underline{d}\}$ are both zero.

3.9-2

(a)  $w = a + \beta x$ gives
 $w_1 = a, \theta_1 = \beta$
 $w_2 = a + \beta L, \theta_2 = \beta$

Subs. into given field; get $w = a + \beta x$
 OK

(b) From given field, $w_{,xx} = (\theta_2 - \theta_1) / L$
 By beam theory: let $M_0 = \text{const. moment}$

$EI w_{,xx} = M_0$ (a) Eliminate M_0
 $\int_{\theta_1}^{\theta_2} EI dw_{,x} = \int_0^L M_0 dx$ between (a) &
 (b); get
 $EI(\theta_2 - \theta_1) = M_0 L$ (b) $w_{,xx} = (\theta_2 - \theta_1) / L$

(c) $w_{,xx} = [\underline{B}] \{ \underline{d} \} = [0, -\frac{1}{L}, 0, \frac{1}{L}] [w_1, \theta_1, w_2, \theta_2]^T$

$[\underline{k}] = \int_0^L [\underline{B}]^T [\underline{B}] EI dx = \frac{EI}{L} \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & -1 \\ 0 & 0 & 0 & 0 \\ 0 & -1 & 0 & 1 \end{bmatrix}$

This $[\underline{k}]$ offers no resistance to w_1 & w_2 and develops no nodal forces in response to θ_1 & θ_2 . $[\underline{k}]$ has rank 1.

3.9-3

(a) Evaluate $u = [1, x^2] \{a\}$ at nodes.

$$\begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} = \begin{bmatrix} 1 & 0 \\ 1 & L^2 \end{bmatrix} \begin{Bmatrix} a_1 \\ a_2 \end{Bmatrix}, \quad \begin{Bmatrix} a_1 \\ a_2 \end{Bmatrix} = \frac{1}{L^2} \begin{bmatrix} L^2 & 0 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} = [\underline{A}]^{-1} \{u\}$$

$$u = [1, x^2] [\underline{A}]^{-1} \{u\} = \begin{bmatrix} \frac{L^2 - x^2}{L^2} & \frac{x^2}{L^2} \end{bmatrix} \{u\}$$

$$\epsilon_x = \left[-\frac{2x}{L^2} \quad \frac{2x}{L^2} \right] \{u\} = [\underline{B}] \{u\}$$

$$[\underline{k}] = \int_0^L [\underline{B}]^T [\underline{B}] AE dx = \frac{4AE}{3L} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

Too large by factor $\frac{4}{3}$. Source is defective u : no x^1 term present, so state of constant ϵ_x is not possible.

(b) Evaluate $u = [x, x^2] \{a\}$ at nodes.

$$\begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} = \begin{bmatrix} -\frac{L}{2} & \frac{L^2}{4} \\ \frac{L}{2} & \frac{L^2}{4} \end{bmatrix} \begin{Bmatrix} a_1 \\ a_2 \end{Bmatrix}, \quad \begin{Bmatrix} a_1 \\ a_2 \end{Bmatrix} = \frac{1}{L^2} \begin{bmatrix} -L & L \\ 2 & 2 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} = [\underline{A}]^{-1} \{u\}$$

$$u = [x, x^2] [\underline{A}]^{-1} \{u\} = \begin{bmatrix} \frac{2x^2 - Lx}{L^2} & \frac{2x^2 + Lx}{L^2} \end{bmatrix} \{u\}$$

$$\epsilon_x = \left[\frac{4x - L}{L^2} \quad \frac{4x + L}{L^2} \right] \{u\} = [\underline{B}] \{u\}$$

$$[\underline{k}] = \int_{-L/2}^{L/2} [\underline{B}]^T [\underline{B}] AE dx = \frac{AE}{3L} \begin{bmatrix} 7 & 1 \\ 1 & 7 \end{bmatrix}$$

\dots when $\{d\} = \{1, 1\}$ (rigid body translation). Source of trouble is defective u : no constant term present.

3.9-4

$$\text{Exact ans.: } u_2 = \frac{PL}{AE}, \quad u_3 = 2 \frac{PL}{AE}$$

$$(a) \frac{4AE}{3L} \begin{bmatrix} 2 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} 0 \\ P \end{Bmatrix}, \quad \begin{Bmatrix} u_2 \\ u_3 \end{Bmatrix} = \frac{PL}{AE} \begin{Bmatrix} 3/4 \\ 3/2 \end{Bmatrix}$$

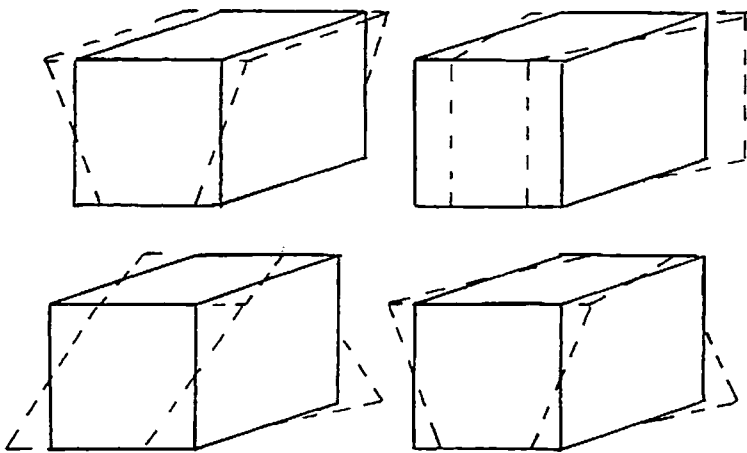
$$\sigma_{x=0} = E [0, 0] \{d\} = 0 \quad (\text{Exact } \sigma_x \text{ is } P/A \text{ for all } x)$$

$$(b) \frac{AE}{3L} \begin{bmatrix} 14 & 1 \\ 1 & 7 \end{bmatrix} \begin{Bmatrix} u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} 0 \\ P \end{Bmatrix}, \quad \begin{Bmatrix} u_2 \\ u_3 \end{Bmatrix} = \frac{PL}{AE} \begin{Bmatrix} -3/97 \\ 42/97 \end{Bmatrix}$$

$$\sigma_{x=0} = E \left[\frac{-3L}{L^2}, \frac{-L}{L^2} \right] \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} = -\frac{1-3PL}{L \cdot 97AE} = \frac{3P}{97A}$$

at $x=0$ in structure; at $x=-\frac{L}{2}$ in element.

3.9-5



With the node numbering of Fig. 3.8 1a, x-direction nodal d.o.f. of the foregoing states are, with c a constant,

$$\begin{Bmatrix} c \\ c \\ -c \\ -c \\ -c \\ -c \\ c \\ c \end{Bmatrix}$$

$$\begin{Bmatrix} c \\ -c \\ -c \\ c \\ -c \\ c \\ c \\ -c \end{Bmatrix}$$

$$\begin{Bmatrix} -c \\ -c \\ c \\ -c \\ c \\ c \\ c \end{Bmatrix}$$

$$\begin{Bmatrix} c \\ -c \\ c \\ -c \\ -c \\ c \\ -c \\ c \end{Bmatrix}$$

3.10-1

$$\text{Let } a_1 = \frac{1}{8}(x_2 - x_3)$$

$$a_2 = \frac{1}{8}(x_3 - x_1)$$

$$a_3 = \frac{1}{8}(x_1 - x_2)$$

$$b_1 = \frac{1}{8}(y_3 - y_2)$$

$$b_2 = \frac{1}{8}(y_1 - y_3)$$

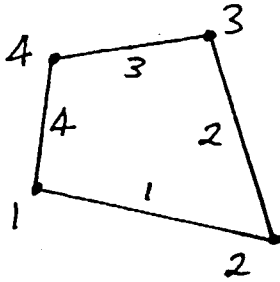
$$b_3 = \frac{1}{8}(y_2 - y_1)$$

$$\begin{pmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \\ u_4 \\ v_4 \\ u_5 \\ v_5 \\ u_6 \\ v_6 \end{pmatrix} = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 1 & 0 \\ \frac{1}{2} & 0 & -b_3 & \frac{1}{2} & 0 & b_3 & 0 & 0 & 0 \\ 0 & \frac{1}{2} & -a_3 & 0 & \frac{1}{2} & a_3 & 0 & 0 & 0 \\ 0 & 0 & 0 & \frac{1}{2} & 0 & -b_1 & \frac{1}{2} & 0 & b_1 \\ 0 & 0 & 0 & 0 & \frac{1}{2} & -a_1 & 0 & \frac{1}{2} & a_1 \\ \frac{1}{2} & 0 & b_2 & 0 & 0 & 0 & \frac{1}{2} & 0 & -b_2 \\ 0 & \frac{1}{2} & a_2 & 0 & 0 & 0 & 0 & \frac{1}{2} & -a_2 \end{bmatrix} \begin{pmatrix} u_1 \\ v_1 \\ w_1 \\ u_2 \\ v_2 \\ w_2 \\ u_3 \\ v_3 \\ w_3 \end{pmatrix}$$

$[T]$

3.10-2

Consider e.g. a quadrilateral element, and apply Eq. 3.10-1 to each side.



$$\frac{\delta_{m1}}{L_1} = \frac{1}{8}(\omega_2 - \omega_1)$$

$$\frac{\delta_{m2}}{L_2} = \frac{1}{8}(\omega_3 - \omega_2)$$

$$\frac{\delta_{m3}}{L_3} = \frac{1}{8}(\omega_4 - \omega_3)$$

$$\frac{\delta_{m4}}{L_4} = \frac{1}{8}(\omega_1 - \omega_4)$$

Add; thus

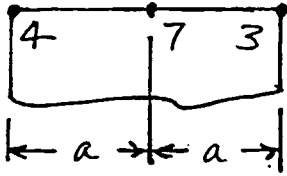
$$\sum_{i=1}^4 \frac{\delta_{mi}}{L_i} = 0$$

3.11-1

Eq. 3.11-6:

$$\begin{Bmatrix} F_4 \\ F_7 \\ F_3 \end{Bmatrix} = \frac{a}{15} \begin{bmatrix} 4 & 2 & -1 \\ 2 & 16 & 2 \\ -1 & 2 & 4 \end{bmatrix} \begin{Bmatrix} q_4 \\ q_7 \\ q_3 \end{Bmatrix}$$

[H]



Unit thickness

$$(a) \begin{Bmatrix} F_4 \\ F_7 \\ F_3 \end{Bmatrix} = [H] \begin{Bmatrix} \sigma \\ 0 \\ -\sigma \end{Bmatrix} = \frac{\sigma a}{3} \begin{Bmatrix} 1 \\ 0 \\ -1 \end{Bmatrix}$$

Couple-moment: $M = F_4(2a) = \frac{2\sigma a^2}{3}$

Flexure formula:

$$M = \frac{\sigma I}{c} = \frac{\sigma (2a)^3 / 12}{a} = \frac{2\sigma a^2}{3}$$

$$(b) \begin{Bmatrix} F_4 \\ F_7 \\ F_3 \end{Bmatrix} = [H] \begin{Bmatrix} 0 \\ \sigma/2 \\ \sigma \end{Bmatrix} = \frac{\sigma a}{3} \begin{Bmatrix} 0 \\ 2 \\ 1 \end{Bmatrix}$$

Moment: $M = 2aF_3 + aF_7 = \frac{4\sigma a^2}{3}$

Flexure formula, for section

4a units deep:

$$m = \frac{\sigma I}{c} = \frac{\sigma (4a)^3 / 12}{2a} = \frac{8\sigma a^2}{3}$$

Contribution of half the section is

$$\frac{M}{2} = \frac{4\sigma a^2}{3}$$

$$(c) \begin{Bmatrix} F_4 \\ F_7 \\ F_3 \end{Bmatrix} = [H] \begin{Bmatrix} 0 \\ \tau \\ 0 \end{Bmatrix} = \frac{\tau a}{15} \begin{Bmatrix} 2 \\ 16 \\ 2 \end{Bmatrix}$$

Shear force: $F_4 + F_7 + F_3 = \frac{4\tau a}{3}$

Beam theory (parabolic distribution):

$$\tau = \frac{3}{2} \frac{V}{2a}, \quad V = \frac{4\tau a}{3}$$

3.11-2

With $[N]$ from Eq. 3.11-5, and constant q ,

$$\begin{aligned} \begin{Bmatrix} F_1 \\ F_2 \\ F_3 \end{Bmatrix} &= \int_{-a}^a [N] q dx = \int_{-a}^a \frac{1}{2a^2} \begin{Bmatrix} x(x-a) \\ 2(a^2-x^2) \\ x(x+a) \end{Bmatrix} q dx \\ &= \frac{q}{2a^2} \begin{Bmatrix} \frac{x^3}{3} - \frac{ax^2}{2} \\ 2a^2x - \frac{2x^3}{3} \\ \frac{x^3}{3} + \frac{ax^2}{2} \end{Bmatrix} \Big|_{-a}^a = \frac{q}{2a^2} \begin{Bmatrix} 2a^3/3 \\ 8a^3/3 \\ 2a^3/3 \end{Bmatrix} = q \begin{Bmatrix} a/3 \\ 4a/3 \\ a/3 \end{Bmatrix} = F \begin{Bmatrix} 1/6 \\ 2/3 \\ 1/6 \end{Bmatrix} \end{aligned}$$

where $F = 2qa$

3.11-3

$$M_1 \delta \theta_{z_1} = \int_0^L v q dx \quad \text{where } v = N_2 \delta \theta_{z_1} \quad \text{and } q \text{ is constant}$$

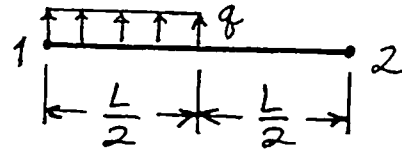
Hence

$$M_1 = q \int_0^L \left(x - \frac{2x^2}{L} + \frac{x^3}{L^2} \right) dx = qL^2 \left(\frac{1}{2} - \frac{2}{3} + \frac{1}{4} \right) = \frac{qL^2}{12}$$

3.11-4

(a)

$$\{r_c\} = \int_0^{L/2} [N]^T dx = q \begin{Bmatrix} x - \frac{x^3}{L^2} + \frac{x^4}{2L^3} \\ \frac{x^2}{2} - \frac{2x^3}{3L} + \frac{x^4}{4L^2} \\ \frac{x^3}{L^2} - \frac{x^4}{2L^3} \\ -\frac{x^3}{3L} + \frac{x^4}{4L^2} \end{Bmatrix}_{L/2}$$



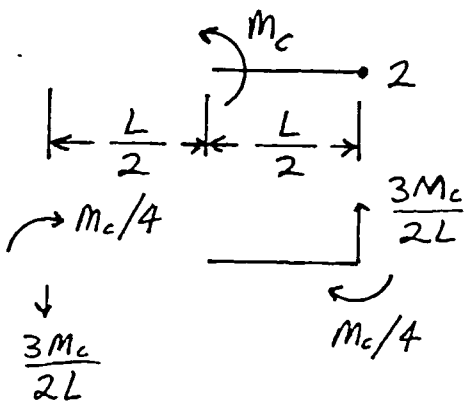
$$\{r_c\} = q \begin{Bmatrix} 0.40625L \\ 0.05729L^2 \\ 0.09375L \\ -0.02604L^2 \end{Bmatrix} \begin{cases} \sum F_z = \frac{qL}{2} \checkmark \\ \sum M_o = r_{e3}L + r_{e2} + r_{e4} \\ = \frac{qL^2}{8} \checkmark \end{cases}$$



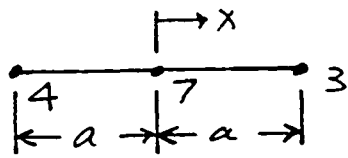
(b) $\{r_e\} = \left[\frac{dN}{dx} \right]_*^T M_c$ where $\left[\frac{dN}{dx} \right]_*^T$ is evaluated at $x = \frac{L}{2}$

From N_i in Fig. 3.2-4 we obtain

$$\begin{Bmatrix} -\frac{6x}{L^2} + \frac{6x^2}{L^3} \\ 1 - \frac{4x}{L} + \frac{3x^2}{L^2} \\ \frac{6x}{L^2} - \frac{6x^2}{L^3} \\ -\frac{2x}{L} + \frac{3x^2}{L^2} \end{Bmatrix}_{x=L/2} = \begin{Bmatrix} -\frac{6}{2L} + \frac{6}{4L} \\ 1 - 2 + \frac{3}{4} \\ \frac{6}{2L} - \frac{6}{4L} \\ -1 + \frac{3}{4} \end{Bmatrix} = \begin{Bmatrix} -\frac{3}{2L} \\ -\frac{1}{4} \\ \frac{3}{2L} \\ -\frac{1}{4} \end{Bmatrix}$$



3.11-5



$$v = [N_4 \quad N_7 \quad N_3]^T \begin{Bmatrix} v_4 \\ v_7 \\ v_3 \end{Bmatrix}$$

Apply Eq. 3.2-7: $N_4 = \frac{-x(a-x)}{a(2a)} = -\frac{x(a-x)}{2a^2}$

$$N_7 = \frac{(-a-x)(a-x)}{-a(a)} = \frac{a^2 - x^2}{a^2}$$

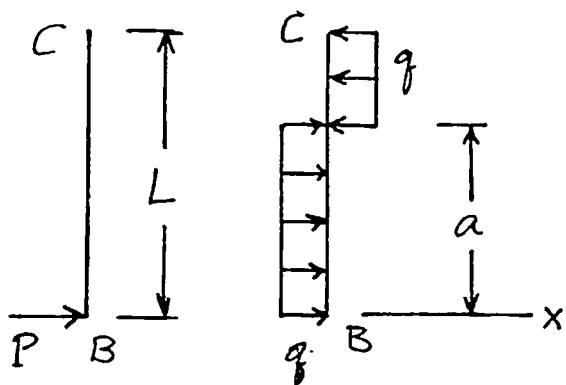
$$N_3 = \frac{(-a-x)(-x)}{-2a(-a)} = \frac{x(a+x)}{2a^2}$$

$$\begin{Bmatrix} F_4 \\ F_7 \\ F_3 \end{Bmatrix} = \begin{Bmatrix} N_4 \\ N_7 \\ N_3 \end{Bmatrix} F = \begin{Bmatrix} -\frac{(a/2)^2}{2a^2} \\ \frac{a^2 - (a/2)^2}{a^2} \\ \frac{a/2(3a/2)}{2a^2} \end{Bmatrix} F = \begin{Bmatrix} -1/8 \\ 3/4 \\ 3/8 \end{Bmatrix} F$$

$$F_4 + F_7 + F_3 = F \quad \checkmark$$

$$\sum M_4 = \frac{3F}{4}a + \frac{3F}{8}(2a) = \frac{3Fa}{4} \quad \checkmark$$

3.11-6



$$\text{where } q = \frac{(1+\sqrt{2})P}{L}$$

$$a = \frac{L}{\sqrt{2}}$$

$$\text{x-direction force: } qa - q(L-a) = q(2a-L)$$

$$= \frac{(1+\sqrt{2})P}{L} \left(\frac{2L}{\sqrt{2}} - L \right)$$

$$= \frac{(1+\sqrt{2})(2-\sqrt{2})}{\sqrt{2}} P$$

$$= \frac{2+\sqrt{2}-2}{\sqrt{2}} P = P \checkmark$$

$$\text{Moment about B: } qa \frac{a}{2} - q(L-a) \frac{L+a}{2} = \frac{q}{2} (2a^2 - L^2)$$

$$= \frac{q}{2} (L^2 - L^2) = 0 \checkmark$$

3.11-7

Evaluate N_i of Eqs. 3.6-4 at Q ($x = \frac{a}{2}$, $y = 0$):

$$\{\underline{r}_e\} = [N]_a^T \begin{Bmatrix} P_x \\ P_y \end{Bmatrix} = \frac{1}{4ab} \begin{bmatrix} ab/2 & 0 & 3ab/2 & 0 & 3ab/2 & 0 & ab/2 & 0 \\ 0 & ab/2 & 0 & 3ab/2 & 0 & 3ab/2 & 0 & ab/2 \end{bmatrix} \begin{Bmatrix} P_x \\ P_y \end{Bmatrix}$$

$$\{\underline{r}_e\} = \frac{1}{8} \begin{bmatrix} 1 & 0 & 3 & 0 & 3 & 0 & 1 & 0 \\ 0 & 1 & 0 & 3 & 0 & 3 & 0 & 1 \end{bmatrix}^T \begin{Bmatrix} 8 \\ 6 \end{Bmatrix}$$

$$\{\underline{r}_e\} = \begin{bmatrix} 1 & \frac{3}{4} & 3 & \frac{9}{4} & 3 & \frac{9}{4} & 1 & \frac{3}{4} \end{bmatrix}^T$$

3.11-8

Get N_i from Eq. 3.2-7. Let $a = \frac{L}{2}$. $B_i = \frac{d}{dx} N_i$

$$N_1 = -\frac{x(a-x)}{2a^2} \quad N_2 = \frac{a^2-x^2}{a^2} \quad N_3 = \frac{x(a+x)}{2a^2}$$

$$B_1 = -\frac{a-2x}{2a^2} \quad B_2 = -\frac{2x}{a^2} \quad B_3 = \frac{a+2x}{2a^2}$$

$$\{\underline{r}_e\} = \int_{-a}^a [B]^T [E] \{\underline{\epsilon}_0\} dV = \int_{-a}^a \begin{Bmatrix} B_1 \\ B_2 \\ B_3 \end{Bmatrix} E \left(\alpha T_3 \frac{a+x}{2a} \right) A dx$$

$$\{\underline{r}_e\} = \frac{EA\alpha T_3}{4a^3} \int_{-a}^a \begin{Bmatrix} -(a-2x) \\ -4x \\ a+2x \end{Bmatrix} (a+x) dx = \frac{EA\alpha T_3}{4a^3} \int_{-a}^a \begin{Bmatrix} -a^2+x+2x^2 \\ -4ax-4x^2 \\ a^2+3x+2x^2 \end{Bmatrix} dx$$

$$\{\underline{r}_e\} = \frac{EA\alpha T_3}{4a^3} \begin{Bmatrix} -2a^3+4a^3/3 \\ -8a^3/3 \\ 2a^3+4a^3/3 \end{Bmatrix} = \frac{EA\alpha T_3}{4a^3} \begin{Bmatrix} -2a^3/3 \\ -8a^3/3 \\ 10a^3/3 \end{Bmatrix} = \frac{EA\alpha T_3}{6} \begin{Bmatrix} -1 \\ -4 \\ 5 \end{Bmatrix}$$

3.11-9

$$\{r_e\} = - \int [B]^T \{\sigma_0\} dV$$

To show $\sum r_{ei} = 0$, we ask if $[I] \{r_e\} = 0$, where $[I]$ is a row vector filled with 1's. We get $\sum r_{ei} = 0$ if $[I] \{B_j\} = 0$, where $\{B_j\}$ is a column of $[B]^T$, for all j in $[B]^T$. Now $[B] = [\partial][N]$, so each row of $[B]$ (and each column of $[B]^T$) is a derivative of $[N]$. But $\sum N_i = 1$ for C^0 elements, so $\sum N_{i,x} = 0$.

3.11-10

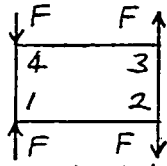
$$[\underline{E}]\{\underline{\epsilon}_0\} = \frac{E}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix} \begin{Bmatrix} \alpha T_0 x \\ \alpha T_0 x \\ 0 \end{Bmatrix} = \frac{E\alpha T_0 x}{1+\nu} \begin{Bmatrix} 1 \\ 1 \\ 0 \end{Bmatrix}$$

$$\{r_e\} = \iint [B]^T [\underline{E}]\{\underline{\epsilon}_0\} t dx dy, N_i = \frac{(a \pm x)(b \pm y)}{4ab}$$

$N_{i,x}$ indep. of x but $\{\underline{\epsilon}_0\}$ linear in x , so integration w.r.t. x with limits $-a$ to a yields zero x -direction nodal loads.

$$\{r_e\}_{y\text{-dir.}} = \frac{E\alpha T_0}{(1+\nu)4ab} \int_{-b}^b \int_{-a}^a \begin{Bmatrix} -(a-x) \\ -(a+x) \\ (a+x) \\ (a-x) \end{Bmatrix} x dx dy$$

$$\{r_e\}_{y\text{-dir.}} = \frac{Ea^2\alpha T_0}{3(1+\nu)} \begin{Bmatrix} 1 \\ -1 \\ 1 \\ -1 \end{Bmatrix} = \begin{Bmatrix} F \\ -F \\ F \\ -F \end{Bmatrix}$$



(as applied to structure nodes)

3.11-11

(a) First the nodal loads, all 3 cases.

$$[N] = \begin{bmatrix} \frac{L-s}{L} & \frac{s}{L} \end{bmatrix}, F = \frac{c}{A}(L_T - x)$$

$$\{r_e\} = \int_0^L [N]^T F dV \text{ where } dV = Adx$$

$$\text{One element: } s=x, L_T=L, \{r_e\} = \frac{cL^2}{6} \begin{Bmatrix} 2 \\ 1 \end{Bmatrix}$$

Two elements: in the first, $s=x, L_T=2L$
in the second, $x=L+s, L_T=2L$

$$\{r_e\}_1 = \frac{cL^2}{6} \begin{Bmatrix} 5 \\ 4 \end{Bmatrix}, \{r_e\}_2 = \frac{cL^2}{6} \begin{Bmatrix} 2 \\ 1 \end{Bmatrix}$$

Three elements: $L_T=3L$; in the respective elements, $x=s, x=L+s, x=2L+s$.

$$\{r_e\}_1 = \frac{cL^2}{6} \begin{Bmatrix} 8 \\ 7 \end{Bmatrix}, \{r_e\}_2 = \frac{cL^2}{6} \begin{Bmatrix} 5 \\ 4 \end{Bmatrix}, \{r_e\}_3 = \frac{cL^2}{6} \begin{Bmatrix} 2 \\ 1 \end{Bmatrix}$$

(b) • One el. $\frac{AE}{L} u_2 = \frac{cL^2}{6}, u_2 = \frac{cL^3}{6AE}$
($L=L_T$)
 $\sigma_x = E \frac{u_2}{L} = \frac{cL^2}{6A}$

• Two els. $\frac{AE}{L} \begin{bmatrix} 2 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} u_2 \\ u_3 \end{Bmatrix} = \frac{cL^2}{6} \begin{Bmatrix} 4+2 \\ 1 \end{Bmatrix}$
($L_T=2L$)

$$\begin{Bmatrix} u_2 \\ u_3 \end{Bmatrix} = \frac{cL^3}{6AE} \begin{Bmatrix} 7 \\ 8 \end{Bmatrix} = \frac{cL_T^3}{6EA} \begin{Bmatrix} 7/8 \\ 1 \end{Bmatrix} \text{ (exact)}$$

$$\sigma_{x1} = E \frac{u_2}{L} = \frac{7cL^2}{6A} = \frac{7}{24} \frac{cL_T^2}{A} = 0.2917 \frac{cL_T^2}{A}$$

$$\sigma_{x2} = E \frac{u_3 - u_2}{L} = \frac{1}{7} \sigma_{x1} = 0.0417 \frac{cL_T^2}{A}$$

• $\frac{AE}{L} \begin{bmatrix} & & 0 \\ -1 & 2 & -1 \\ 0 & -1 & 1 \end{bmatrix} \begin{Bmatrix} u_2 \\ u_3 \\ u_4 \end{Bmatrix} = \frac{cL^2}{6} \begin{Bmatrix} 7+5 \\ 4+2 \\ 1 \end{Bmatrix}$ 3 els.
($L_T=3L$)

$$\begin{Bmatrix} u_2 \\ u_3 \\ u_4 \end{Bmatrix} = \frac{cL^3}{6AE} \begin{Bmatrix} 19 \\ 26 \\ 27 \end{Bmatrix} \sigma_{x1} = E \frac{u_2}{L} = 0.3519 \frac{cL_T^2}{A}$$

$$\sigma_{x2} = E \frac{u_3 - u_2}{L} = 0.1296 \frac{cL_T^2}{A}$$

$$\sigma_{x3} = E \frac{u_4 - u_3}{L} = 0.0185 \frac{cL_T^2}{A}$$

Exact stresses: multipliers of cL_T^2/A are:

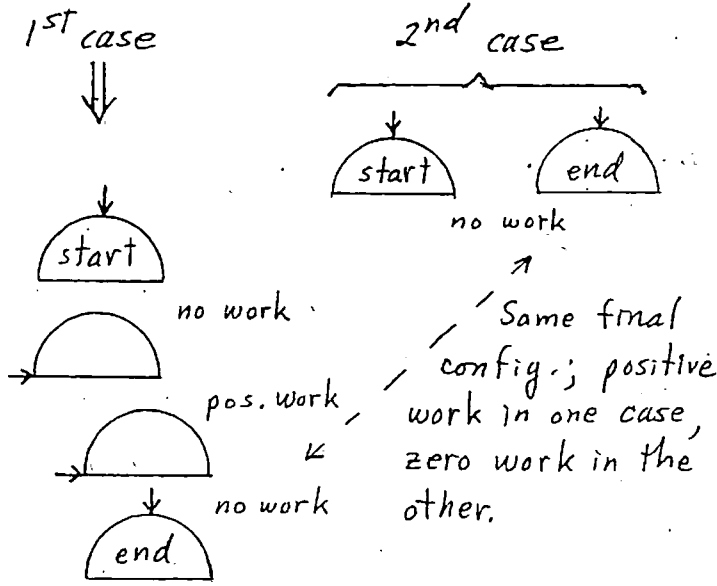
$\frac{x}{L_T}$	0	$\frac{1}{6}$	$\frac{1}{4}$	$\frac{1}{2}$	$\frac{3}{4}$	$\frac{5}{6}$	1
factor	.5	.3472	.2813	.1250	.0313	.0139	0

left
end

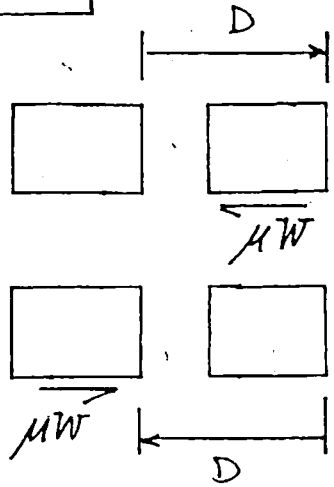
middle

right
end

4.2-1



4.2-2



Work of friction force:

$$(-\mu W) D$$

$$\mu W (-D)$$

Net work of friction force is $-2\mu W D$
 Not zero, so not conservative.